



U.S. Department of Transportation
Federal Aviation Administration
Washington, D.C.

Master Minimum Equipment List

Revision: 54a
Date: 05/12/2010

BOEING
B-737
100/200/300/400/500/600/700/800/900

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MASTER MINIMUM EQUIPMENT LIST
B-737

Contents

SYSTEM	PAGE(S)	REVISION	DATE
Title Page		54a	05/12/2010
Contents	i	54a	05/12/2010
Highlights of Change	ii, iii, iv	54a	05/12/2010
Definitions (Operator Responsibility)	v	54a	05/12/2010
Preamble (Operator Responsibility)	vi	54a	05/12/2010
21 - Air Conditioning	21-1 Thru 21-37	54a	05/12/2010
22 - Auto Flight	22-1 Thru 22-8	53	08/01/2009
23 - Communications	23-1 Thru 23-16	54a	05/12/2010
24 - Electrical Power	24-1 Thru 24-4	53	08/01/2009
25 - Equipment/Furnishings	25-1 Thru 25-14	54a	05/12/2010
26 - Fire Protection	26-1 Thru 26-21	54a	05/12/2010
27 - Flight Controls	27-1 Thru 27-12	53	08/01/2009
28 - Fuel	28-1 Thru 28-18	54a	05/12/2010
29 - Hydraulic Power	29-1 Thru 29-3	53	08/01/2009
30 - Ice and Rain Protection	30-1 Thru 30-11	53	08/01/2009
31 - Indicating/Recording Systems	31-1 Thru 31-6	53	08/01/2009
32 - Landing Gear	32-1 Thru 32-6	53	08/01/2009
33 - Lights	33-1 Thru 33-11	54	10/12/2009
34 - Navigation	34-1 Thru 34-20	54a	05/12/2010
35 - Oxygen	35-1, 35-2	52	04/29/2008
36 - Pneumatic	36-1 Thru 36-3	52	04/29/2008
38 - Water/Waste	38-1	52	04/29/2008
46 - Information Systems	46-1, 46-2	53	08/01/2009
47 - Inert Gas System	47-1	54	10/12/2009
49 - Airborne Auxiliary Power	49-1 Thru 49-3	54a	05/12/2010
52 - Doors	52-1 Thru 52-13	54	10/12/2009
73 - Engine Fuel & Control	73-1 Thru 73-2	54a	05/12/2010
74 - Ignition	74-1	52	04/29/2008
75 - Bleed Air	75-1	52	04/29/2008
77 - Engine Indicating	77-1 Thru 77-3	52	04/29/2008
78 - Engine Exhaust	78-1, 78-2	52	04/29/2008
79 - Engine Oil	79-1	52	04/29/2008
80 - Engine Starting	80-1	52	04/29/2008

MASTER MINIMUM EQUIPMENT LIST B-737

Highlights of Change

EFFECTIVE ABOVE DATE, the Boeing 737 Master Minimum Equipment List has been revised. Please replace Affected sections for a complete up-to-date MMEL. SEE NOTE BELOW FOR IMPORTANT INFORMATION ABOUT THIS REVISION. Retain this sheet with your MMEL until the next revision is issued.

NOTE 1: Pages ARE NOT itemized in a CONTROL PAGE for individual revisions. Rather, the ATA section will contain the applicable number of pages and revision number. Any changes to any item within an ATA section will result in the entire section receiving a revision number. To find any revised item within the section see the list below.

	System/Sequence Number	Remarks
21	AIR CONDITIONING	
	22. Forward Outflow Valve (-100/-200/-300/-400/-500, including STC's SA2969SO, ST01566LA, and ST01961SE)	Revised MMEL item title to clarify that the item is applicable to all -100 through -500, including those with the STC's installed. The previous title implied that it was applicable only to those models when the STC's are installed.
23	COMMUNICATIONS	
	10. Cockpit Voice Recorder System (CVR)	
***	2) Models with Recorder Independent Power Supply (RIPS)	Added new sub-item and reformatted relief for aircraft "with" and "without" RIPS. 14 CFR section 25.1457(d)(5), Amendment 124, requires an independent power source (RIPS) be installed on new production aircraft. Item is optional for aircraft manufactured before Apr 7, 2010. The RIPS contains an independent backup power source (battery) and interface circuitry to maintain operation of the cockpit voice recorder and area microphone for 10 minutes after loss of normal power to the CVR.
	a) Recorder Independent Power Supply (RIPS)	Added new sub-item for RIPS specific faults. Requires the CVR operate normally and the battery removed.
***	25. Automated Flight Information Reporting System (AFIRS) (STC's ST10345SC and ST02361NY)	Added STC ST02361NY, AeroMechanical Services Ltd. AFIRS 220 with optional Global Voice and Global Text Satellite Communications.
25	EQUIPMENT/ FURNISHINGS	
	5. Cargo Compartment Restraint Components	Revised in accordance with MMEL Policy Letter 100, Revision 2, dated January 20, 2009.
	17. Emergency Medical	Edited provisos to reflect wording in MMEL Policy Letter 73, R4, dated

MASTER MINIMUM EQUIPMENT LIST B-737

System/Sequence Number	Remarks
Equipment (Includes STC ST10238SC)	4/18/2006. Changed "Automated" to "Automatic."
26 FIRE PROTECTION	
15. Lavatory Fire Extinguisher Systems	Updated MMEL coverage to be consistent with MMEL Policy Letter 24, Revision 4. Changes to this item clarify the relief available to cargo configurations.
16. Lavatory Smoke Detection System	Updated MMEL coverage to be consistent with MMEL Policy Letter 24, Revision 4. Changes to this item clarify the relief available to cargo configurations.
*** 19. Lower Cargo Compartment Fire Detection/Suppression Systems (All models and STC's)	
*** 13) DETECTOR FAULT Light (Boeing Installed System Only)	<p>Added new sub-item for DETECTOR FAULT light (Boeing installed units only). The amber DETECTOR FAULT light is located on the cargo fire panel on the aft electronic panel. The light illuminates when one or more of the detector loops in either cargo compartment has failed. The QRH contains an entry for when the DETECTOR FAULT light illuminates, but there are no flight crew procedures for this condition.</p> <p>The DETECTOR FAULT light will also illuminate when the cargo fire TEST switch is pushed and one or more detectors in the loop(s) have failed. The flight crew will normally do a cargo fire test once per flight day or following a flight crew change.</p> <p>When the DETECTOR FAULT light is inoperative, it cannot be used to determine whether a cargo fire detector or detector loop is inoperative. However, when the cargo fire TEST switch is pushed, the FWD and AFT cargo fire warning lights (red) illuminate when ALL detectors in the selected loops respond to the fire test. If any loop or detector is inoperative, the associated cargo fire warning light will NOT illuminate during the test. This test can be used in lieu of the DETECTOR FAULT light to determine if a detector or detector loop is faulty. If a detector or detector loop is determined to be faulty, dispatch would also be available for that condition under a separate sub-item in this MMEL item.</p>
28 FUEL	
25. Center Tank Fuel Boost Pump Automatic Shut Off System (Service Bulletin 737-28A1228, 737-28A1216, 737-28A1206, or Equivalent Installed)	New item added.
34 NAVIGATION	
25. Altitude Alerting System	Revised relief in accordance with Policy Letter 39, Revision 5, dated January 29, 2010.
48. EFIS Control Panel	Revised sub-item titles by adding "Map Switches" for -300/-400/-500 and -

MASTER MINIMUM EQUIPMENT LIST B-737

System/Sequence Number	Remarks
	<p>600/-700/-800/-900. Added additional sub-item for the -300/-400/-500 as sub-item 2 under EFIS Control Panel for Decision Height Reference (DH REF) Indication and renumbered subsequent sub-item.</p> <p>This new sub-item for the Decision Height Reference Indication applies to -300/-400/-500 airplanes which have a decision height selector and indication on the EFIS control panel. The primary display for the decision height that is selected on the EFIS control panel is on the EADI. The EFIS control panel also has an indication of the selected decision height. With the decision height indication inoperative on either or both of the EFIS control panels, the EADI indications of decision height must be available to the flight crew.</p>
49 AIRBORNE AUXILIARY POWER	
7. APU Bleed Air System	Revised proviso for "APU Not Operated" option by deleting the (M), Maintenance procedure and removed the word "open." Since the APU is not operated under this option, it does not matter whether the bleed air valve is open or closed.
73 ENGINE FUEL & CONTROL	
12. Electronic Engine Control (EEC) Alternate Power Supply System (-600/-700/-800/-900)	<p>Added new item for EEC's. The Electronic Engine Controls (EEC) installed on the 737NG are a dual-channel design with full redundancy in each channel. Each EEC channel has its own dedicated power supply, processing capability, input sensors and output drivers and is, therefore, a candidate for the MMEL due to redundancy. A single inoperative EEC channel is permitted for up to a 150 hour inspection/repair interval under the EEC time-limited dispatch limits.</p> <p>Relief is provided in accordance with MMEL Policy Letter 45, Revision 2, dated 3/04/2004.</p>

MASTER MINIMUM EQUIPMENT LIST
B-737

Definitions

Insert current Policy Letter 25 DEFINITIONS here.

MASTER MINIMUM EQUIPMENT LIST
B-737

Preamble

Insert current Policy Letter 34 or 36, as applicable, PREAMBLE here.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-1

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**21 - AIR CONDITIONING

1.

Air Conditioning
Packs1) All Passenger
Configuration (All
Models)a) (-100/-200/
-300/-400/
-500/-600 and
-700/-800
Without PATS
Auxiliary Fuel
Tanks)

C

2

1

(O) Except for ER operations, one may be inoperative provided flight altitude remains at or below FL 250.

b) (-700IGW/-
800 with PATS
Auxiliary Fuel
Tanks)

C

2

1

(M)(O) Except for ER operations, one may be inoperative provided:
a) Flight altitude remains at or below FL 250,
and
b) Auxiliary fuel bleed air pressurization system (if installed) is verified to be operational before each departure.

c) (-900)

C

2

1

(M)(O) Except for ER operations, one may be inoperative provided:
c) Flight altitude remains at or below FL 250,
d) Forward cargo heat duct is secured closed,
and
e) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).

d) (-100/-200)

C

2

0

(M)(O) Except for ER operations, both may be inoperative provided flight is conducted in an unpressurized configuration.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-2

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED	
			3.	NUMBER REQUIRED FOR DISPATCH	
				4.	REMARKS OR EXCEPTIONS
<u>21 - AIR CONDITIONING</u>					
1.	Air Conditioning Packs (Cont'd)				
	1) All Passenger Configuration (All Models) (Cont'd)				
	e) (-300/-400/-500)	C	2	0	(M)(O) Except for ER operations, both may be inoperative provided: a) Flight is conducted in an unpressurized configuration, and b) Procedures are established and used to ensure lower cargo compartments remain empty or are verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits. NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.
	f) (-600/-700/-800)	C	2	0	(M)(O) Except for ER operations, both may be inoperative provided: a) Flight is conducted in an unpressurized configuration, b) Recirculation fan(s) operates normally, c) Both E / E equipment cooling exhaust fans operate normally, d) Procedures are established and used to ensure lower cargo compartments remain empty or are verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits, and e) Auxiliary tanks, if installed, remain empty or auxiliary fuel is included as part of zero fuel weight. NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-3

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

21 - AIR CONDITIONING

1.

Air Conditioning
Packs (Cont'd)1) All Passenger
Configuration (All
Models) (Cont'd)

g) (-900)

C

2

0

(M)(O) Except for ER operations, both may be
inoperative provided:

- a) Flight is conducted in an unpressurized configuration,
- b) Recirculation fans operate normally,
- c) Both E / E equipment cooling exhaust fans operate normally,
- d) Procedures are established and used to ensure lower cargo compartments remain empty or are verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits, and
- e) Forward cargo heat duct is secured closed, and
- f) Airport ambient temperature does not exceed 103° F (39° C).

NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.

2) Combi and All
Cargo
Configurations
(737C, QC, and
STC's
ST01566LA,
ST01827LA,
ST00283AT, and
ST01961SE)

C

2

0

(M)(O) Except for ER operations, both may be
inoperative provided:

- a) Flight is conducted in an unpressurized configuration, and
- b) Procedures are established and used to ensure main deck cargo compartment remains empty or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits.

NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.

a) Right Pack

C

1

0

(O) Except for ER operations, may be inoperative
provided flight altitude remains at or below FL 250.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-4

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED		
			3.	NUMBER REQUIRED FOR DISPATCH	
				4. REMARKS OR EXCEPTIONS	
<u>21 - AIR CONDITIONING</u>					
1.	Air Conditioning Packs (Cont'd)				
	2) Combi and All Cargo Configurations (737C, QC, and STC's ST01566LA, ST01827LA, ST00283AT, and ST01961SE) (Cont'd)				
	a) Left Pack	C	1	0	(O) Except for ER operations, may be inoperative provided:
					a) Flight Altitude remains at or below FL250, and
					b) Procedures are established and used to ensure main deck cargo compartment remains empty or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits.
					NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.
	3) All Cargo Configuration (-700C)	C	2	1	(O) Except for ER operations may be inoperative provided flight altitude remains at or below FL 250.
	4) Pemco COMBI (STC ST03387AT), and All Cargo Configurations	C	2	1	Except for ER operations, one may be inoperative provided only flight deck is occupied.
		C	2	1	(O) Except for ER operations one may be inoperative provided flight altitude remains at or below FL 250
					(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-5

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**21 - AIR CONDITIONING

1. Air Conditioning Packs (Cont'd)
- 4) Pemco COMBI (STC ST03387AT), and All Cargo Configurations (Cont'd)

C

2

0

(M)(O) Except for ER operations, may be inoperative provided:

- a) Flight is conducted in an unpressurized configuration, and
- b) Procedures are established and used to ensure main deck cargo compartments remain empty or are verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits.

NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.

2. Pack Air Flow/Shut-off Valves (includes STC SA2969SO)

C

2

0

(M)(O) May be inoperative deactivated Closed.

- 1) High Flow Mode (-300/-400/-500/-600/-700/-800/-900)

C

2

0

- 2) APU High Flow Mode

C

2

0

3. Pack Trip Warning Systems

C

2

0

(M)(O) May be inoperative provided associated pack is not used.

4. Pack Turbofan (-100/-200/-300/-400/-500)

- 1) All Passenger Configuration (All Models)

C

2

0

(O) May be inoperative provided associated pack(s) is operated only in flight with flaps retracted.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-6

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED		
			3. NUMBER REQUIRED FOR DISPATCH		
			4. REMARKS OR EXCEPTIONS		
<u>21 - AIR CONDITIONING</u>					
4.	Pack Turbofan (-100/-200/-300/-400/ -500) (Cont'd)				
	2) Combi and All Cargo Configurations (737C, QC, STC's SA2969SO, ST01566LA, ST01827LA, ST00283AT, and ST01961SE)				
	a) Right Pack Turbofan	C	1	0	(O) May be inoperative provided right pack is operated only in flight with flaps retracted.
	b) Left Pack Turbofan	C	1	0	(O) May be inoperative provided: a) Left pack is operated only in flight with flaps retracted, and b) Procedures are established and used to ensure main deck cargo compartment remains empty or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits.
					NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.
5.	Pack Ram Air Systems	C	2	0	(M)(O) May be inoperative in FLIGHT OPEN position provided: a) Operations are not conducted on runways covered with slush, or on gravel runways, and b) Associated pack is not operated during takeoff or landing on wet runways or runways with standing water.
***	1) Exhaust Louver Assemblies (-100/-200/-300/ -400/-500)	C	2	0	(M)(O) May be inoperative provided: a) Actuator(s) is disconnected, and b) Louver(s) is secured in full open position.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT: B-737			REVISION NO : DATE:		54a 05/12/2010		PAGE: 21-7
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED		
					3.	NUMBER REQUIRED FOR DISPATCH	
						4.	REMARKS OR EXCEPTIONS
21 - AIR CONDITIONING							
6.	Pack Turbofan Valves (-100/-200/-300/-400/-500)						
	1) All Passenger Configuration (All Models)	C	2	0			(M)(O) May be inoperative Closed provided associated pack(s) is operated only in flight with flaps retracted.
	2) Combi and All Cargo Configurations (737C, QC, STCs SA2969SO, ST01566LA, ST01827LA, ST00283AT, and ST01961SE)						
	a) Right Pack Turbofan Valve	C	1	0			(M)(O) May be inoperative Closed provided right pack is operated only in flight with flaps retracted.
	b) Left Pack Turbofan Valve	C	1	0			(M)(O) May be inoperative Closed provided: a) Left pack is operated only in flight with flaps retracted, and b) Procedures are established and used to ensure main deck cargo compartment remains empty or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits.
	NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.						
7.	RAM DOOR FULL OPEN Indicating Lights	C	2	0			
8.	Air Mix Valves (-100/-200/-300/-500/-600/-700)	C	2	0			(M)(O) May be inoperative provided associated pack is not used.
9.	Air Mix Valve Position Indicators (-100/-200/-300/-500/-600/-700)	C	2	0			

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-8

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
<u>21 - AIR CONDITIONING</u>			3.	NUMBER REQUIRED FOR DISPATCH
				4. REMARKS OR EXCEPTIONS
10. Cabin Rate of Climb Indicator				
1) Analog Control System (-100/-200/-300/-400/-500)	C	1	0	May be inoperative provided AUTO and STBY control modes operate normally.
	C	1	0	(M)(O) May be inoperative provided flight is conducted in unpressurized configuration.
2) Digital Control System (-300/-400/-500/-600/-700/-800/-900)	C	1	0	May be inoperative provided AUTO and ALTN control modes operate normally.
a) (-300/-400/-500)	C	1	0	(M)(O) May be inoperative provided: a) Flight is conducted in unpressurized configuration, and b) Outflow valve is positioned to 25% open position.
b)(-600/-700/-800 prior to incorporation of Boeing Service Bulletins 737-21-1135, 737-26-1121 and 737-26-1122, or production equivalent)	C	1	0	(M)(O) May be inoperative provided: a) Flight is conducted in unpressurized configuration, b) Outflow valve is positioned to 25% open position, and c) Recirculation fan(s) operates normally.
(Continued)				

MASTER MINIMUM EQUIPMENT LIST

PAGE:

21-9

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	3.	NUMBER REQUIRED FOR DISPATCH	4.	REMARKS OR EXCEPTIONS
21 - AIR CONDITIONING									
10.	Cabin Rate of Climb Indicator (Cont'd)								
	2) Digital Control System (-300/-400/-500/-600/-700/-800/-900) (Cont'd)								
	c) (-600/-700/-800 upon incorporation of Boeing Service Bulletins 737-21-1135, 737-26-1121 and 737-26-1122, or production equivalent)	C	1	0	(M) (O) May be inoperative provided:				
					a) Flight is conducted in an unpressurized configuration,				
					b) Procedures are established and used to ensure lower forward cargo compartment remains empty, or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits.				
					c) Outflow valve is positioned to 25% open position, and				
					d) Recirculation fan(s) operate normally.				
					NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.				
	d) (-900)	C	1	0	(M)(O) May be inoperative provided:				
					a) Flight is conducted in unpressurized configuration,				
					b) Outflow valve is positioned to 25% open position,				
					c) Recirculation fans operate normally,				
					d) Forward cargo heat duct is secured closed, and				
					e) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).				
11.	Cabin Altitude Warning System	C	1	0	May be inoperative provided flight altitude remains at or below 10,000 feet MSL.				
***	1) High Altitude Warning System	C	1	0	May be inoperative provided procedures do not require its use.				
(Continued)									

MASTER MINIMUM EQUIPMENT LIST

PAGE:

21-10

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED				3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
21 - AIR CONDITIONING									
11.	Cabin Altitude Warning System (Cont'd)								
***	2) CABIN ALTITUDE Light								
	a) -100/-200/-300/-400/-500 (upon incorporation of Boeing Service Bulletin 737-31A1325)	C	1	0				May be inoperative provided TAKEOFF CONFIG warning light operates normally.	
		C	1	0				(O) May be inoperative provided flight crew performs a briefing on cabin altitude warning indications and procedures before engine start for the first flight of the day or following any change of either flight crew member.	
	b) -600/-700/-800/-900 (upon incorporation of Boeing Service Bulletin 737-31A1332, or production equivalent)	C	2	0				May be inoperative provided associated TAKEOFF CONFIG warning light operates normally.	
		C	2	0				(O) May be inoperative provided flight crew performs a briefing on cabin altitude warning indications and procedures before engine start for the first flight of the day or following any change of either flight crew member.	
12.	Cabin Altitude Indicator								
	1) Analog Control system (-100/-200/-300/-400/-500)	C	1	0				May be inoperative provided: a) Cabin differential pressure indicator operates normally, and b) A chart is provided to crew to convert differential pressure to cabin altitude.	
		C	1	0				(M)(O) May be inoperative provided flight is conducted in an unpressurized configuration.	
(Continued)									

MASTER MINIMUM EQUIPMENT LIST

PAGE:

21-11

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-12

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED		
<u>21 - AIR CONDITIONING</u>			3. NUMBER REQUIRED FOR DISPATCH		
			4. REMARKS OR EXCEPTIONS		
12.	Cabin Altitude Indicator (Cont'd)				
	2) Digital Control System (-300/-400/-500/-600/-700/-800/-900) (Cont'd)				
	d) (-900)	C	1	0	(M)(O) May be inoperative provided: a) Flight is conducted in unpressurized configuration, b) Outflow valve is positioned to 25% open position, c) Recirculation fans operate normally, d) Forward cargo heat duct is secured closed, and e) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).
13.	Cabin Differential Pressure Indicator				
	1) Analog Control System (-100/-200/-300/-400/-500)	C	1	0	May be inoperative provided: a) Cabin altitude indicator operates normally, and b) A chart is provided to crew to convert cabin altitude to differential pressure.
		C	1	0	(M)(O) May be inoperative provided flight is conducted in an unpressurized configuration. (Continued)

MASTER MINIMUM EQUIPMENT LIST

PAGE:

21-13

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-14

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED		
			3.	NUMBER REQUIRED FOR DISPATCH	
				4. REMARKS OR EXCEPTIONS	
<u>21 - AIR CONDITIONING</u>					
13.	Cabin Differential Pressure Indicator (Cont'd)				
	2) Digital Control System (-300/-400/-500/-600/-700/-800/-900) (Cont'd)				
	d) (-900)	C	1	0	(M)(O) May be inoperative provided:
					a) Flight is conducted in unpressurized configuration,
					b) Outflow valve is positioned to 25% open position,
					c) Recirculation fans operate normally,
					d) Forward cargo heat duct is secured closed, and
					e) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).
14.	Cabin Pressure Control System				
	1) Analog Control System	C	2	1	(O) One may be inoperative provided manual mode (AC and DC actuators) operates normally.
	Automatic/ Standby Modes (-100/ -200/-300/ -400/-500)				
	2) Analog Control System	C	3	0	(M)(O) May be inoperative for unpressurized flight provided:
	Automatic/ Standby/Manual Modes (-100/ -200/-300/ -400/-500)				a) Outflow valve is deactivated open or removed, and
					b) Extended overwater flight is prohibited.
					(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-15

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**21 - AIR CONDITIONING

14.

Cabin Pressure
Control System
(Cont'd)

C

2

1

3) Digital Control
System
Automatic
Modes (-300/
-400/-500/-600/
-700/-800/-900)(M)(O) One may be inoperative provided:
a) Manual mode operates normally,
b) Inoperative controller is deactivated, and
c) Auxiliary fuel bleed air pressurization system (if
installed) is verified to be operational before
each departure.a) (-300/-400/
-500)

C

2

0

(M)(O) May be inoperative for unpressurized flight
provided:

- a) Outflow valve is deactivated in 25% open
position or removed, and
- b) Extended overwater flight is prohibited.

b) (-600/-700/
-800 prior to
incorporation
of Boeing
Service
Bulletins 737-
21-1135,
737-26-1121
and 737-26-
1122, or
production
equivalent)

C

2

0

(M)(O) May be inoperative for unpressurized flight
provided:
a) Outflow valve is deactivated in 25% open
position or removed,
b) Recirculation fan(s) operates normally,
c) Extended overwater flight is prohibited, and
d) Auxiliary tanks (if installed) remain empty or
auxiliary fuel is included as part of zero fuel
weight.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-16

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3.

NUMBER REQUIRED FOR DISPATCH

4.

REMARKS OR EXCEPTIONS

21 - AIR CONDITIONING

14.

Cabin Pressure
Control System
(Cont'd)3) Digital Control
System
Automatic
Modes (-300/
-400/-500/-600/
-700/-800/-900)
(Cont'd)c) (-600/-700/ -
800 upon
incorporation
of Boeing
Service
Bulletins 737-
21-1135,
737-26-1121
and 737-26-
1122, or
production
equivalent)

C

2

0

(M) (O) May be inoperative for unpressurized flight
provided:

- a) Procedures are established and used to ensure lower forward cargo compartment remains empty, or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits.
- b) Outflow valve is deactivated in 25% open position or removed,
- c) Recirculation fan(s) operate normally,
- d) Extended overwater flight is prohibited, and
- e) Auxiliary tanks (if installed) remain empty or auxiliary fuel is included as part of zero fuel weight.

NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.

d) (-900)

C

2

0

(M)(O) May be inoperative for unpressurized flight
provided:

- a) Outflow valve is deactivated in 25% open position or removed,
- b) Recirculation fans operate normally,
- c) Extended overwater flight is prohibited,
- d) Forward cargo heat duct is secured closed, and
- e) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-17

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**21 - AIR CONDITIONING

14.

Cabin Pressure
Control System
(Cont'd)4) Digital Control
System Manual
Mode (-300/
-400/-500/-600/
-700/-800/-900)a) (-300/-400/
-500)

C

1

0

(M)(O) May be inoperative for unpressurized flight
provided:a) Outflow valve is deactivated in 25% open
position or removed, and

b) Extended overwater flight is prohibited.

b) (-600/-700/
-800 all
passenger
configuration
prior to
incorporation
of Boeing
Service
Bulletins 737-
21-1135, 737-
26-1121 and
737-26-1122,
or production
equivalent)

C

1

0

(M)(O) May be inoperative for unpressurized flight
provided:a) Outflow valve deactivated to 25% open
position or removed,

b) Recirculation fan(s) operates normally, and

c) Extended overwater flight is prohibited.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-18

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
21 - AIR CONDITIONING			3. NUMBER REQUIRED FOR DISPATCH	
			4. REMARKS OR EXCEPTIONS	
14. Cabin Pressure Control System (Cont'd)				
4) Digital Control System Manual Mode (-300/-400/-500/-600/-700/-800/-900) (Cont'd)				
c) (-600/-700/-800 all passenger configuration upon incorporation of Boeing Service Bulletins 737-21-1135, 737-26-1121 and 737-26-1122, or production equivalent)	C	1	0	(M) (O) May be inoperative for unpressurized flight provided: a) Procedures are established and used to ensure lower forward cargo compartment remains empty, or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits. b) Outflow valve is deactivated in 25% open position or removed, c) Recirculation fan(s) operate normally, and d) Extended overwater flight is prohibited. NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.
d) (-900)	C	1	0	(M)(O) May be inoperative for unpressurized flight provided: a) Outflow valve is deactivated in 25% open position or removed, b) Recirculation fans operate normally, c) Extended overwater flight is prohibited, d) Forward cargo heat duct is secured closed, and e) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737	REVISION NO : DATE:	54a 05/12/2010	PAGE: 21-19
---------------------------	-------------------------------	-------------------	-----------------------

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	
					3.	NUMBER REQUIRED FOR DISPATCH
						4. REMARKS OR EXCEPTIONS
21 - AIR CONDITIONING						
15.	Main Outflow Valve					
	1)	Analog Control System Outflow Valve Actuators (AC and/or DC) (-100/ -200/-300/ -400/-500)	C	2	1	One actuator may be inoperative for pressurized cargo-only flight, provided airplane is depressurized before landing.
			C	2	0	(M)(O) May be inoperative for unpressurized flight provided: a) Outflow valve is deactivated open or removed, and b) Extended overwater flight is prohibited.
	2)	Digital Control System Outflow Valve Automatic Mode Actuators				
	a)	(-300/-400/ -500)	C	2	1	One may be inoperative provided manual mode actuator operates normally.
			C	2	0	(M)(O) May be inoperative for unpressurized flight provided: a) Outflow valve is deactivated in 25% open position or removed, and b) Extended overwater flight is prohibited.
	b)	(-600/-700/ -800/-900)	C	2	1	One may be inoperative provided manual mode actuator operates normally.
	c)	(-600/-700/ -800 prior to incorporation of Boeing Service Bulletins 737-21-1135, 737-26-1121 and 737-26-1122, or production equivalent)	C	2	0	(M)(O) May be inoperative for unpressurized flight provided: a) Outflow valve is deactivated in 25% open position or removed, b) Recirculation fan(s) operate normally, and c) Extended overwater flight is prohibited.
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-20

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
			3.	NUMBER REQUIRED FOR DISPATCH
				4. REMARKS OR EXCEPTIONS
<u>21 - AIR CONDITIONING</u>				
15.	Main Outflow Valve (Cont'd)			
	2) Digital Control System Outflow Valve Automatic Mode Actuators (Cont'd)			
	d) (-600/-700/ -800 upon incorporation of Boeing Service Bulletins 737- 21-1135, 737- 26-1121 and 737-26-1122, or production equivalent)	C	2	0 (M) (O) May be inoperative for unpressurized flight provided: a) Procedures are established and used to ensure lower forward cargo compartment remains empty, or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits. b) Outflow valve is deactivated in 25% open position or removed, c) Recirculation fan(s) operate normally, and d) Extended overwater flight is prohibited. NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.
	e) (-900)	C	2	0 (M)(O) May be inoperative for unpressurized flight provided: a) Outflow valve is deactivated in 25% open position or removed, b) Recirculation fans operate normally, c) Extended overwater flight is prohibited, d) Forward cargo heat duct is secured closed, and e) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).
				(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-21

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**21 - AIR CONDITIONING15. Main Outflow Valve
(Cont'd)3) Digital Control
System Outflow
Valve Manual
Mode Actuatora) (-300/-400/
-500) C

1

0

(M)(O) May be inoperative for unpressurized flight
provided:

- a) Outflow valve is deactivated in 25% open position or removed, and
- b) Extended overwater flight is prohibited.

b) (-600/-700/
-800 prior to
incorporation
of Boeing
Service
Bulletins 737-
21-1135, 737-
26-1121 and
737-26-1122,
or production
equivalent) C

1

0

(M)(O) May be inoperative for unpressurized flight
provided:

- a) Outflow valve is deactivated in 25% open position or removed,
- b) Recirculation fan(s) operate normally, and
- c) Extended overwater flight is prohibited.

c) (-600/-700/
-800 upon
incorporation
of Boeing
Service
Bulletins 737-
21-1135, 737-
26-1121 and
737-26-1122,
or production
equivalent) C

1

0

(M) (O) May be inoperative for unpressurized
flight provided:

- a) Procedures are established and used to ensure lower forward cargo compartment remains empty, or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits.
- b) Outflow valve is deactivated in 25% open position or removed,
- c) Recirculation fan(s) operate normally, and
- d) Extended overwater flight is prohibited.

NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-22

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED	
				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
21 - AIR CONDITIONING					
15.	Main Outflow Valve (Cont'd)				
	3) Digital Control System Outflow Valve Manual Mode Actuator (Cont'd)				
	d) (-900)	C	1	0	(M)(O) May be inoperative for unpressurized flight provided: a) Outflow valve is deactivated in 25% open position or removed, b) Recirculation fans operate normally, c) Extended overwater flight is prohibited, d) Forward cargo heat duct is secured closed, and e) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).
16.	Pressure Relief Valves				
	1) Analog Control System (-100/-200/-300/-400/-500)	C	2	1	(M) One may be inoperative closed for pressurized flight.
		C	2	0	(M)(O) May be inoperative provided flight is conducted in an unpressurized configuration.
	2) Digital Control System (-300/-400/-500/-600/-700/-800/-900)	C	2	1	(M) One may be inoperative closed for pressurized flight.
	a) (-300/-400/-500)	C	2	0	(M)(O) May be inoperative provided: a) Flight is conducted in an unpressurized configuration, and b) Outflow valve is positioned to 25% open position.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737	REVISION NO : DATE:	54a 05/12/2010	PAGE: 21-23
---------------------------	-------------------------------	-------------------	-----------------------

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
21 - AIR CONDITIONING						
16.	Pressure Relief Valves (Cont'd)					
	2) Digital Control System (-300/-400/-500/-600/-700/-800/-900) (Cont'd)					
	b) (-600/-700/-800 prior to incorporation of Boeing Service Bulletins 737-21-1135, 737-26-1121 and 737-26-1122, or production equivalent)	C	2	0	(M)(O) May be inoperative provided:	a) Flight is conducted in an unpressurized configuration, b) Outflow valve is positioned to 25% open position, and c) Recirculation fan(s) operate normally,
	c) (-600/-700/-800 upon incorporation of Boeing Service Bulletins 737-21-1135, 737-26-1121 and 737-26-1122, or production equivalent)	C	2	0	(M) (O) May be inoperative provided:	a) Flight is conducted in an unpressurized configuration, b) Procedures are established and used to ensure lower forward cargo compartment remains empty, or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits. c) Outflow valve is positioned to 25% open position, and d) Recirculation fan(s) operate normally.
					NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.	
	d) (-900)	C	2	0	(M)(O) May be inoperative provided:	a) Flight is conducted in an unpressurized configuration, b) Outflow valve is positioned to 25% open position, c) Recirculation fans operate normally, d) Forward cargo heat duct is secured closed, and e) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-24

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED	
			3.	NUMBER REQUIRED FOR DISPATCH	
				4. REMARKS OR EXCEPTIONS	
<u>21 - AIR CONDITIONING</u>					
17.	Temperature Indicators				
	1) Supply Duct (-100/-200/-300/-500/-600/-700)	C	1	0	May be inoperative provided both duct overheat warning systems operate normally.
	2) Supply Duct (-400/-800/-900)	C	3	0	May be inoperative provided associated ZONE TEMP light operates normally.
	3) Pass Cabin	C	-	0	
	4) Pack (-400/-800/-900)	C	2	0	
18.	Duct Overheat Warning Lights				
	1) DUCT OVERHEAT (-100/-200/-300/-500/-600/-700)	C	2	0	May be inoperative provided supply duct temperature indicators operate normally.
	2) ZONE TEMP (-400/-800/-900)	C	3	0	May be inoperative provided associated supply duct temperature indicator operates normally.
19.	Passenger Cabin Temperature Control Systems				
	1) Automatic/Manual Controls (-100/-200/-300/-500/-600/-700)	C	2	1	
		C	2	0	(O) May be inoperative provided right pack is not used.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-25

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

21 - AIR CONDITIONING

19.

Passenger Cabin
Temperature Control
Systems (Cont'd)

2) FWD/AFT

a) (-400/-800/
-900)

C

2

0

(O) May be dispatched with faults indicated by ZONE
TEMP Light(s) during Master Caution recall provided
associated temperature control system is checked to
operate normally before each takeoff.

b) (-400/-800)

C

2

0

(M)(O) May be inoperative provided Trim Air Pressure
Regulating and Shutoff Valve remains CLOSED.

C

2

0

(M)(O) May be inoperative provided associated Trim
Air Modulating Valve is deactivated CLOSED.

c) (-900)

C

2

0

(M)(O) May be inoperative provided:
a) Trim Air Pressure Regulating and Shutoff
Valve remains Closed,
b) Forward cargo heat duct is secured closed,
and
c) Airport ambient temperature does not exceed
103 degrees F (39 degrees C).

C

2

0

(M)(O) May be inoperative provided:
a) Associated Trim Air Modulating Valve is
deactivated CLOSED,
b) Forward cargo heat duct is secured closed,
and
c) Airport ambient temperature does not exceed
103 degrees F (39 degrees C).

20.

Cabin Temperature
Indicator

Incorporated into item 21-17 Revision 34a.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-26

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
			3. NUMBER REQUIRED FOR DISPATCH	
21 - AIR CONDITIONING				4. REMARKS OR EXCEPTIONS
21.	Flight Deck Temperature Control Systems			
	1) Automatic/Manual Controls (-100/- 200/ -300/-500/- 600/-700)	C 2	1	
	C	2	0	(O) May be inoperative provided left pack is not used.
	2) Primary/Back-up Modes			
	a) (-400/-800/ -900)	C 2	1	(O) One may be inoperative provided remaining temperature control is verified to operate normally.
	b) (-400/-800)	C 2	0	(M)(O) May be inoperative provided Trim Air Pressure Regulating and Shutoff Valve remains CLOSED.
	C	2	0	(M)(O) May be inoperative provided associated Trim Air Modulating Valve is deactivated CLOSED.
	c) (-900)	C 2	0	(M)(O) May be inoperative provided: a) Trim Air Pressure Regulating and Shutoff Valve remains CLOSED, b) Forward Cargo heat duct is secured closed, and c) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).
	C	2	0	(M)(O) May be inoperative provided: a) Associated Trim Air Modulating Valve is deactivated CLOSED, b) Forward cargo heat duct is secured closed, and c) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010		PAGE: 21-27
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
<u>21 - AIR CONDITIONING</u>					
22.	Forward Outflow Valve (-100/-200/-300/-400/-500, including STC's SA2969SO, ST01566LA, and ST01961SE)	C	1	0	Except for 737C and STC ST01566LA and ST01961SE cargo or cargo/passenger operations, may be inoperative closed.
		C	1	0	May be inoperative open provided both packs operate normally.
		C	1	0	(O) May be inoperative open with one pack operating normally provided flight altitude remains at or below FL200.
23.	FORWARD OUTFLOW CLOSED Indicating Light (-100/-200)	C	1	0	
24. ***	Gasper Fan (-100/-200/-300/-500/-600/-700)	D	1	0	
25.	Water Separator Anti-Icing Systems (-100/ -200/-300/-500/-600/-700)	C	2	0	(M)(O) May be inoperative provided associated pack is not used.
26.	Ground Preconditioned Air Connection Check Valve	C	1	0	May be inoperative closed.
	1) Analog Control System (-100/-200/-300/-400/-500)	C	1	0	(M)(O) May be inoperative open provided: a) Flight is conducted in an unpressurized configuration, and b) Procedures are established and used to ensure main deck cargo compartment (as applicable) remains empty, or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits.
NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.					
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-28

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED		
			3. NUMBER REQUIRED FOR DISPATCH		
21 - AIR CONDITIONING				4. REMARKS OR EXCEPTIONS	
26.	Ground Preconditioned Air Connection Check Valve (Cont'd)				
	2) Digital Control System				
	a) (-300/-400/ -500)	C	1	0	(M)(O) May be inoperative provided: a) Flight is conducted in an unpressurized configuration, and b) Outflow valve is positioned to 25% open position.
	b)(-600/-700/ -800 prior to incorporation of Boeing Service Bulletins 737- 21-1135, 737- 26-1121 and 737-26-1122, or production equivalent)	C	1	0	(M)(O) May be inoperative provided: a) Flight is conducted in an unpressurized configuration, b) Outflow valve is positioned to 25% open position, and c) Recirculation fan(s) operates normally.
	c) (-600/-700/ -800 upon incorporation of Boeing Service Bulletins 737- 21-1135, 737- 26-1121 and 737-26-1122, or production equivalent)	C	1	0	(M) (O) May be inoperative provided: a) Flight is conducted in an unpressurized configuration, b) Procedures are established and used to ensure lower forward cargo compartment remains empty, or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits. c) Outflow valve is positioned to 25% open position, and d) Recirculation fan(s) operate normally.
					NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.
					(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-29

SYSTEM & SEQUENCE NUMBER		1.	2. NUMBER INSTALLED		
			3. NUMBER REQUIRED FOR DISPATCH		
			4. REMARKS OR EXCEPTIONS		
<u>21 - AIR CONDITIONING</u>					
26.	Ground Preconditioned Air Connection Check Valve (Cont'd)				
	2) Digital Control System (Cont'd)				
	d) (-900)	C	1	0	(M)(O) May be inoperative provided: a) Flight is conducted in an unpressurized configuration, b) Outflow valve is positioned to 25% open position, c) Recirculation fans operate normally, d) Forward cargo heat duct is secured closed, and e) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).
27.	Electrical/Electronic Equipment Cooling Blowers				
	1) Non-EFIS (-100/ -200/-300/-400/ -500)	C	2	1	Except for ER operations, one may be inoperative.
	2) EFIS (-300/-400/ -500)				
	a) Supply Fans	C	2	1	Except for ER operations, one may be inoperative.
	b) Exhaust Fans	C	2	1	Except for ER operations, one may be inoperative.
	3) CDS (-600/-700/ -800/-900)	B	4	3	(M) One fan may be inoperative provided: a) All remaining fans are verified to operate normally, and b) Both low flow detectors are verified to operate normally.
28. ***	Equipment Cooling Check Valve (-100/-200)	D	1	0	May be inoperative open.
29. ***	Air Cleaner Purge Valves (-100/-200/-300)	C	2	0	

MASTER MINIMUM EQUIPMENT LIST

PAGE:

21-30

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	
					3.	NUMBER REQUIRED FOR DISPATCH
21 - AIR CONDITIONING						4. REMARKS OR EXCEPTIONS
30.	Control Cabin	C	1	0		(M)(O) May be inoperative with fan wind-milling provided OAT remains at or below 120 degrees F (49 degrees C).
***	Augmentation Fan (-200)	C	1	0		(M)(O) May be inoperative with fan wind-milling provided OAT remains at or below 115 degrees F (46 degrees C) if PDCS is installed and operates normally.
		C	1	0		(M)(O) May be inoperative with fan seized provided: a) One air condition pack operates normally, b) OAT remains at or below 100 degrees F (38 degrees C), and c) Window heat operates normally.
31.	Recirculation Fan(s)	C	1	0		May be inoperative provided left pack is operating when OAT is above 100 degrees F (38 degrees C).
	1) (-300/-500)	C	1	0		
	2) (-400 and Pemco -400 COMBI)	C	2	1		One fan may be inoperative provided left pack is operating when OAT is above 100 degrees F (38 degrees C).
		C	2	0		May be inoperative provided OAT remains below 100 degrees F (38 degrees C).
	3) (-600/-700)	C	1	0		May be inoperative provided: a) Left pack is operating when OAT is above 100 degrees F (38 degrees C), b) Flight is conducted pressurized, and c) Both packs operate normally.
	4) (-800/-900)	C	2	1		Left fan may be inoperative provided left pack is operating when OAT is above 100 degrees F (38 degrees C).
		C	2	1		Right fan may be inoperative provided: a) Left pack is operating when OAT is above 100 degrees F (38 degrees C), and b) Flight is conducted pressurized.
		C	2	0		May be inoperative provided: a) OAT remains below 100 degrees F (38 degrees C), and b) Flight is conducted pressurized.
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737		REVISION NO : 54a DATE: 05/12/2010		PAGE: 21-31	
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
<u>21 - AIR CONDITIONING</u>					
31.	Recirculation Fan(s) (Cont'd)				
	4) (-800/-900)				
	a) (-800EF STC ST02000NY)	C	1	0	May be inoperative provided: a) Left pack is operating when OAT is above 100 degrees F (38 degrees C), b) Flight is conducted pressurized, and c) Both packs operate normally.
	5) (-300QC/F, - 400F) (STC's ST01566LA, SA2969SO, and SA2970SO Only)	C	1	0	May be inoperative in cargo configuration.
	a) (STC SA2970SO)	C	1	0	May be inoperative in PAX mode provided OAT remains below 100 degrees F (38 degrees C).
32.	Pack Temperature Control System(s) (Electronic Pack/ Zone Controller) (-400/-800/-900)	C	4	2	(O) One system (primary or standby) on each pack may be inoperative provided remaining system on associated pack is checked to operate normally.
33.	Pack Temperature Control Valves (-400/-800/-900)	C	2	0	(O) May be inoperative provided associated Standby Pack Temperature Control Valve(s) is checked to operate normally.
		C	2	0	(M)(O) May be inoperative provided associated pack is not used.
34.	Standby Pack Temperature Control Valves (-400/-800/-900)	C	2	0	(O) May be inoperative provided associated Pack Temperature Control Valve(s) is checked to operate normally.
		C	2	0	(M)(O) May be inoperative provided associated pack is not used.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-32

SYSTEM & SEQUENCE NUMBER		1.	2.		NUMBER INSTALLED	
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
21 - AIR CONDITIONING						
35.	Trim Air Pressure Regulating and Shutoff Valve					
	1) (-400/-800)	C	1	0	(M) May be inoperative secured closed.	
	2) (-900)	C	1	0	(M)(O) May be inoperative secured closed provided: a) Forward cargo heat duct is secured closed, and b) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).	
36.	Trim Air Modulating Valves					
	1) (-400/-800)	C	3	0	(M) May be inoperative closed.	
		C	3	0	(O) May be inoperative in any position provided Trim Air Pressure Regulating and Shutoff Valve remains closed.	
	2) (-900)	C	3	0	(M)(O) May be inoperative closed provided: a) Forward cargo heat duct is secured closed, and b) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).	
		C	3	0	(M)(O) May be inoperative in any position provided: a) Trim Air Pressure Regulating and Shutoff Valve remains CLOSED, b) Forward cargo heat duct is secured closed, and c) Airport ambient temperature does not exceed 103 degrees F (39 degrees C).	
37. ***	Outflow Valve heater Gasket (-100/-200/-300/-400/-500)	C	1	0		
38.	Outflow Valve Position Indicator	C	1	0	(M)(O) May be inoperative provided valve is verified to be operating normally.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-33

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**21 - AIR CONDITIONING

39.

Trim Air Check
Valves

1) (-400/-800/-900)

C

2

1

(M) One may be inoperative provided associated valve
is deactivated closed.

40.

Equipment Cooling
Automatic Flow
Control Valve/
Overboard Exhaust
Valve1) Analog Control
System (-100/
-200/-300/-400/
-500)

C

1

0

(M)(O) May be inoperative in open position provided
flight is conducted in an unpressurized configuration.

C

1

0

May be inoperative in closed position provided both
packs and recirculation fan(s) (if installed) are
operated during ground taxi operations.2) Digital Control
Systema) (-300/-400/
-500)

C

1

0

(M)(O) May be inoperative in open position provided:
a) Flight is conducted in an unpressurized
configuration, and
b) Outflow valve is positioned to 25% open
position.

C

1

0

May be inoperative in closed position provided both
packs and recirculation fan(s) (if installed) are
operated during ground taxi operations.b) (-600/-700/
-800 prior to
incorporation
of Boeing
Service
Bulletins 737-
21-1135, 737-
26-1121 and
737-26-1122,
or production
equivalent)

C

1

0

(M)(O) May be inoperative in open position provided:
a) Flight is conducted in an unpressurized
configuration,
b) Outflow valve is positioned to 25% open
position, and
c) Recirculation fan(s) operate normally.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-34

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
21 - AIR CONDITIONING						
40.	Equipment Cooling Automatic Flow Control Valve/ Overboard Exhaust Valve (Cont'd)					
	2) Digital Control System (Cont'd)					
	c) (-600/-700/-800 upon incorporation of Boeing Service Bulletins 737-21-1135, 737-26-1121 and 737-26-1122, or production equivalent)	C	1	0	(M) (O) May be inoperative in open position provided: a) Flight is conducted in an unpressurized configuration, b) Procedures are established and used to ensure lower forward cargo compartment remains empty, or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits, c) Outflow valve is positioned to 25% open position, and d) Recirculation fan(s) operate normally.	
					NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.	
	d) (-900)	C	1	0	(M)(O) May be inoperative in open position provided: a) Flight is conducted in an unpressurized configuration, b) Outflow valve is positioned to 25% open position, c) Recirculation fan(s) operate normally, d) Forward cargo heat duct is secured closed, and e) Airport ambient air temperature does not exceed 103 degrees F (30 degrees C).	
	e) (-600/-700/-800/-900)	C	1	0	(M)(O) Except for ER operations, may be inoperative provided: a) Actuator is verified to be in smoke position, and b) Both packs operate normally.	

U.S. DEPARTMENT OF TRANSPORTATION					MASTER MINIMUM EQUIPMENT LIST				
FEDERAL AVIATION ADMINISTRATION									
AIRCRAFT: B-737				REVISION NO : DATE:			54a 05/12/2010	PAGE: 21-35	
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED				
					3.	NUMBER REQUIRED FOR DISPATCH			
						4.	REMARKS OR EXCEPTIONS		
<u>21 - AIR CONDITIONING</u>									
41.	Door Area Heater Systems								
***	1) Main Deck Cargo Door Heating Blankets/ Systems (737C and -700C)	D	-	0					
	2) Entry Door Area and Overwing Emergency Exit Hatch Area Heater Systems (-600/-700/-800/-900)	D	-	0			(M) May be inoperative deactivated.		
	3) Main Cargo Door Heater System (STC ST01566LA)	D	1	0			(M) May be inoperative in Quick Change cargo configuration.		
***	4) Mid-Exit Door Area Heater System (-900ER)	D	1	0			(M) May be inoperative deactivated.		
42.	Equipment Cooling Low Flow Detector Systems (-600/-700/-800/-900)	B	2	1			(M)(O) One may be inoperative provided associated fans (supply or exhaust) are verified to operate normally.		
43.	Equipment Cooling Air Filter (-600/-700/-800/-900)	C	1	0			(M) Equipment Cooling System may be operated with filter removed.		
44.	Fan Bypass Check Valves (-600/-700/-800/-900)	C	2	0			May be inoperative open/missing provided airport ambient temperature remains below 80 degrees F (27 degrees C).		
		C	2	0			May be inoperative open/missing for an associated inoperative pack.		
		D	2	1			One may be inoperative open/missing provided pack associated with remaining fan bypass check valve operates normally.		

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-36

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3.	NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
21 - AIR CONDITIONING								
45.	Air Distribution Riser Shutoff Valves (-700C)							
	1) Passenger Configuration	C	2	0		(M)	May be inoperative provided valves are deactivated open.	
	2) Passenger and Cargo Configurations	C	2	0		(M)(O)	May be inoperative in closed position provided: a) Flight is conducted in an unpressurized configuration, b) Recirculation fan operates normally, c) Both E/E equipment cooling exhaust fans operate normally, and d) Procedures are established and used to ensure main deck (as applicable) and lower cargo compartments remain empty, or are verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULD's), and/or Fly Away Kits.	
							NOTE: Operator MELs must define which items are approved for inclusion in Fly Away Kits, and which materials can be used as ballast.	
	a) Right Riser SOV	C	1	0		(M)(O)	Except for ER operations, may be inoperative closed provided operation is limited to left pack only.	
	b) Left Riser SOV	C	1	0		(M)(O)	Except for ER operations, may be inoperative closed provided operation is limited to one pack.	
46.	Air Heater Supernumerary Compartment STC ST01566LA (-300RB) and ST01961SE	D	1	0			May be inoperative provided compartment is not occupied.	
47. ***	Humidification System (-800EF STC ST02000NY)	C	3	0			May be inoperative provided: a) Manual shutoff valve is closed, and b) All Humidifier Switches are in OFF.	
48. ***	Zonal Drying System (-800EF STC ST02000NY)	C	1	0		(M)	May be inoperative provided: a) Manual shutoff valve is closed, and b) Dryer/Humidifier power is removed.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

21-37

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**21 - AIR CONDITIONING49. Return Air Grille
(-600/-700/-800/
-900)

C

-

-

(M) One may be broken or missing provided:
a) Broken or missing grille is located within a
designated area as defined by Boeing, and
b) Grille is removed and replaced with a blanking
plate.50. Flight Deck Foot and
Shoulder Heater
Systems

C

4

0

May be inoperative provided flight deck temperature is
acceptable to flight crew.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

22-1

SYSTEM & SEQUENCE NUMBER		1.	2. NUMBER INSTALLED			4. REMARKS OR EXCEPTIONS
			3. NUMBER REQUIRED FOR DISPATCH			
22 - AUTO FLIGHT						
1.	Autopilot Systems	C	-	1	Except for ER operations, may be inoperative provided: a) Approach minimums do not require their use, b) Enroute operations do not require autopilot use, and c) Number of flight segments and segment duration is acceptable to flight crew. NOTE1: Operators should make every effort to repair autopilot early in repair interval, as provided by this relief statement, in consideration of such factors as weather, traffic density, and effect of other inoperative systems. NOTE2: Any mode which functions normally may be used. If CWS is inoperative, do not use other modes (pitch or roll).	
		B	-	0		
	1) Control Wheel Autopilot Disconnect Switches	C	2	1	One may be inoperative provided: a) autopilot is not used below 1500 feet AGL, and, b) Approach minimums do not require use of autopilot.	
		B	2	0	May be inoperative provided autopilot is not used.	
***	2) Autopilot Disengage Bar	C	1	0		
2.	Autopilot Disengaged Warning System					
	1) Lights	C	2	1	One may be inoperative when autopilot is used in any axis.	
		B	2	0	(O) Except for ER operations, may be inoperative provided autopilots are not used.	
***	2) Aural Warning	C	1	0	May be inoperative provided approach minimums do not require its use.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

22-2

SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED		
			3.	NUMBER REQUIRED FOR DISPATCH		
				4.	REMARKS OR EXCEPTIONS	
22 - AUTO FLIGHT						
3.	Yaw Damper					
	1) (-100/-200/-300/ -400/-500)					
	a) Without Rudder Pressure Reducer System installed	C	1	0	(O) May be inoperative provided yaw damper switch remains OFF.	
					NOTE: Refer to AFM Limitations for SP-77 autopilot.	
	b) With Rudder Pressure Reducer System installed	C	1	0	(M)(O) May be inoperative provided: a) Yaw damper switch remains OFF, and b) Rudder Pressure Reducer System is verified to operate normally.	
					NOTE: Refer to AFM Limitations for SP-77 autopilot.	
		C	1	0	(M)(O) May be inoperative provided yaw damper is deactivated.	
					NOTE: Refer to AFM Limitations for SP-77 autopilot.	
	2) (-600/-700/-800/ -900)	C	1	0	(O) May be inoperative provided yaw damper switch remains off.	
***	3) Yaw Damper Indicator	C	1	0		
4. ***	Autothrottle System	C	1	0	May be inoperative provided approach minimums do not require its use.	
5.	Mach Trim Systems	C	-	0	(M)(O) May be inoperative provided: a) AFM limitations are observed, and b) Mach trim actuator is verified to be in null/uncommanded elevator position.	
	1) (-300/-400/-500/- 600/-700/-800/ -900)	C	2	1	(M) One may be inoperative deactivated provided: a) Remaining Mach trim system is verified to operate normally, and b) Mach trim fail light operates normally.	

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST		
AIRCRAFT: B-737			REVISION NO : DATE:		53 08/01/2009	PAGE: 22-3
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	
					3.	NUMBER REQUIRED FOR DISPATCH
						4. REMARKS OR EXCEPTIONS
22 - AUTO FLIGHT						
6.	SP-77/SP-177/SP-300/Collins Flight and Approach Mode Annunciations	C	-	0		Individual mode annunciations may be inoperative provided associated system modes are not used.
***	1) SP-177/SP-300 Annunciator Panels (-200/-300/-400/-500)	C	2	1		One may be inoperative provided: a) Engaged system (AP, FD, AT, PDCS, or FMCS) is at pilot position with operative mode annunciator, and b) Approach minimums do not require their use.
		C	2	0		May be inoperative provided associated systems are not used. NOTE: PDCS or FMCS data on CDU may be valid when PDC or FMC annunciator is inoperative.
	2) SP-77 Approach Progress Displays (-100/-200)	C	2	1		One may be inoperative provided approach minimums do not require their use.
		C	2	0		May be inoperative provided associated system modes are not used.
7. ***	Dual Angle of Attack Sensors/Stall Warning system Sensors/Alpha Vanes (-100/-200/-300/-400/-500)					
	1) SP-177	C	2	1		(M) Right sensor/vane may be inoperative provided: a) Autopilot B is restricted to CWS, and b) Systems affected by inoperative sensor/vane are deactivated or turned off, and their MEL provisions observed.
	2) SP-300	C	2	1		(M) Left or right sensor/vane may be inoperative provided: a) Associated autopilot channel is restricted to CWS, and b) Systems affected by inoperative sensor/vane are deactivated or turned off, and their MEL provisions observed.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

22-4

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
22 - AUTO FLIGHT						
8.	Autothrottle	C	2	1		One may be inoperative when autothrottle is used provided approach minimums do not require their use.
***	Disengage Lights	C	2	0		May be inoperative provided autothrottle is not used.
9.	Speed Trim Fail Light System (-300/-400/-500/-600/-700/-800/-900)	C	1	0		(M) May be inoperative provided speed trim system is verified to operate normally.
10.	Speed Trim System (-300/-400/-500/-600/-700/-800/-900)	C	2	1		(M) One may be inoperative deactivated provided: a) Remaining speed trim system is verified to operate normally, and b) Speed trim fail light operates normally.
11.	STAB OUT OF TRIM Light	B	1	0		(O) Except for ER operations, may be inoperative provided autopilots are not used.
12.	Autopilot Trim Circuit Breaker Monitor (-100/-200)	C	1	0		(M) Trim circuit to monitor stabilizer trim CB may be inoperative provided remaining functions of STAB OUT OF TRIM light operate normally.
13.	Automatic Thrust Restoration (ATR) System (-300)	C	1	0		May be inoperative unless procedures require its use.

14.	Mode Control Panel Selectors (-200/-300/-400/-500/-600/-700/-800/-900)					
***	1) V/S Selector (DOWN, UP)	C	1	0		May be inoperative provided procedures do not require its use.
***	2) Bank Angle Selector (AUTO, 10, 15, 20, 25, 30)	C	1	0		

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST		
FEDERAL AVIATION ADMINISTRATION						
AIRCRAFT:		REVISION NO :		53		PAGE:
B-737		DATE:		08/01/2009		22-5
SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		
				3.	NUMBER REQUIRED FOR DISPATCH	
					4. REMARKS OR EXCEPTIONS	
22 - AUTO FLIGHT						
15.	Mode Control Panel Switches/Paddles (-200/-300/-400/-500/-600/-700/-800/-900)					
	1) A/P CWS Engage Switches	C	2	0		
	2) A/P CMD Engage Switches	C	2	1		
		B	2	0	(O) Except for ER operations may be inoperative provided autopilots are not use.	
***	3) Autothrottle Arm Switch	C	1	0	May be inoperative provided approach minimums do not require autothrottle use.	
***	4) A/T SPEED Switch	C	1	0	May be inoperative provided approach minimums do not require autothrottle use.	
***	5) F/D Switches	C	2	0	May be inoperative provided approach minimums do not require flight director use.	
***	6) IAS/MACH Change Over Switch	C	1	0		
***	7) APP Switch	C	1	0	May be inoperative provided approach minimums do not require autopilot or flight director use.	
***	8) EPR/N1, LNAV, VNAV, LVL CHG, V/S, HDG SEL, ALT HOLD, and VOR/LOC Switches	C	-	0	May be inoperative provided Enroute operations do not require their use.	
***	9) SPD INTV, PDC and ALT INTV Switches	C	-	0		

MASTER MINIMUM EQUIPMENT LIST

PAGE:

22-6

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	
					3.	NUMBER REQUIRED FOR DISPATCH
						4. REMARKS OR EXCEPTIONS
22 - AUTO FLIGHT						
16.	Mode Control Panel Windows					
***	1)	Vertical Speed (VERT SPEED) (-200/-300/-400/ -500/-600/-700/ -800/-900)	C	1	0	May be inoperative provided procedures do not require its use.
	2)	(EFIS/PFD/ND) (-300/-400/ -500/ -600/-700/-800/ -900) (Includes STC ST03355AT)				
	a)	Airspeed (IAS/MACH)	C	1	0	May be inoperative and associated selector used provided selected airspeed indications operate normally.
	b)	Heading (HEADING)	C	1	0	May be inoperative and associated selector used provided selected heading indications operate normally.
	c)	Vertical Speed (VERT SPEED)	C	1	0	May be inoperative provided procedures do not require its use.
	d)	Vertical Speed (VERT SPEED) (-600/-700/ -800/-900)	C	1	0	May be inoperative and associated selector used provided selected vertical speed indications operate normally.
	e)	Altitude (ALTITUDE) (-600/-700/ -800/-900)	C	1	0	May be inoperative and associated selector used provided selected altitude indications operate normally.
	f)	Course (COURSE)	C	2	0	May be inoperative and associated selector used provided selected course indications operate normally.
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION			
FEDERAL AVIATION ADMINISTRATION		MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737		REVISION NO : 53 DATE: 08/01/2009	PAGE: 22-7
SYSTEM & SEQUENCE NUMBER	1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH
4. REMARKS OR EXCEPTIONS			
22 - AUTO FLIGHT			
16.	Mode Control Panel Windows (Cont'd) 2) (EFIS/PFD/ND) (-300/-400/ -500/ -600/-700/-800/ -900) (Includes STC ST03355AT) (Cont'd) g) Window Lighting	B 1 0	May be inoperative provided: a) Selected airspeed indications operate normally, b) Selected heading indications operate normally, c) Selected vertical speed indications operate normally, d) Selected altitude indications operate normally, and e) Selected course indications operate normally.
17.	Takeoff/Go-Around (TO/GA) Switches	C 2 1 C 2 0	One may be inoperative provided approach minimums do not require its use. May be inoperative provided: a) Both thrust levers are operated manually for takeoff, and b) Autopilot and Flight Director are not used below Minimum Descent Altitude or 500 feet, whichever is higher. NOTE: Flight director go-around and windshear guidance are not available with both TO/GA switches inoperative.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

22-8

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
22 - AUTO FLIGHT							
18.	Mode Control Panel Switch Lights						

	1) Autopilot Engage Switch Lights						
	a) CWS	C	2	0			
	b) CMD	C	2	1			
		B	2	0			(O) Except for ER operations may be inoperative provided autopilots are not used.
	2) Mode Selector Switch Lights	C	-	0			
	3) A/T ARM Switch Light	C	1	0			
19.	Thrust Mode Annunciator/Thrust Mode Display (-300/-400/-500/-600/-700/-800/-900)	C	1	0			May be inoperative provided thrust mode limits are observed.
20.	Automatic Landing System						
***	1) Fail Passive	C	1	0			May be inoperative provided approach minimums do not require its use.
***	2) Fail Operational (LAND 3) (-600/-700/-800/-900)	C	1	0			May be inoperative provided approach minimums do not require its use.
***	3) AUTOLAND Light	C	2	0			(O) May be inoperative provided alternate procedures are established and used.
		D	2	0			May be inoperative provided procedures do not require its use.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-1

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

23 - COMMUNICATIONS

1.
***Flight Deck Speaker
System

C

1

0

May be inoperative provided:

- a) Procedures do not require its use, and
- b) Headset earphones or headphones associated with inoperative speaker(s) are installed and operate normally.

2.

Passenger Address
System (Includes
STC ST10238SC)1) Passenger
Configuration

B

1

0

(O) May be inoperative provided:

- a) Alternate, normal and emergency procedures and/or operating restrictions are established and used, and
- b) Flight attendant alerting system (audio and visual) operates normally.

NOTE: Any station function(s) that operate normally may be used.

C

1

0

(O) May be inoperative provided:

- a) PA not required by 14 CFR, and
- b) Alternate, normal and emergency procedures, and/or operating restrictions are established and used.

NOTE: Any station function(s) that operate normally may be used.

a) Lavatory
Speakers

C

-

0

(O) May be inoperative provided Alternate procedures are established and used.

b) Cabin
Speakers

C

-

-

May be inoperative provided inoperative speakers are not adjacent to each other.

(Continued)

MASTER MINIMUM EQUIPMENT LIST

PAGE:

23-2

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		
				3. NUMBER REQUIRED FOR DISPATCH		
				4. REMARKS OR EXCEPTIONS		
23 - COMMUNICATIONS						
2.	Passenger Address System (Includes STC ST10238SC) (Cont'd)					
	2) Cargo Configuration (Courier/ Supernumerary Address System)	C	1	0		(O) May be inoperative provided alternate, normal and emergency procedures and/or operating restrictions are established and used,
		D	1	0		May be inoperative provided procedures do not require its use.
	a) Lavatory Speakers	C	1	0		(O) May be inoperative provided alternate procedures are established and used.
		D	1	0		May be inoperative provided procedures do not require its use.
3.	Communication Systems (VHF and UHF)	D	-	-		Any in excess of those required by 14 CFR may be inoperative provided it is not powered by Standby Bus and is not required for emergency procedures.
	1) VHF Comm Control Panels	C	-	-		One side of VHF Comm Control panel tuning function may be inoperative provided: a) Associated transceiver can be tuned from opposite side of control panel, and b) Associated transceiver operates normally.
***	a) Active Frequency Light	C	-	0		
	b) Frequency Transfer Switch	C	-	0		May be inoperative provided associated VHF active frequency can be selected.
		D	-	-		May be inoperative provided associated VHF radio is considered inoperative.
	c) Frequency Selector Knob	C	-	2		
	d) Frequency Indication	C	-	2		
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-3

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**

23 - COMMUNICATIONS

3. Communication
Systems (VHF and
UHF) (Cont'd)*** 2) Radio Tuning
Panels C 3 2a) Off-Side
Tuning Light C - 04. Crewmember
Interphone System1) Passenger
Configuration
a) Flight Deck
to Cabin,
Cabin to
Flight Deck
Functions B - -

(O) May be inoperative provided:

- a) Flight deck to cabin and cabin to flight deck interphone functions operate normally on at least fifty percent of cabin handsets, and
- b) Alternate communications procedures between affected flight attendant station(s) are established and used.

NOTE: Any station function(s) that operate normally may be used.

b) Cabin to
Cabin
Function B 2 0

(O) May be inoperative provided alternate communications procedures between affected flight attendant station(s) are established and used.

NOTE: Any station function(s) that operate normally may be used.

B - -

(O) May be inoperative provided:

- a) Cabin to cabin interphone functions operate normally on at least fifty percent of cabin handsets, and
- b) Alternate communications procedures between affected flight attendant station(s) are established and used.

NOTE: Any station function(s) that operate normally may be used.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-4

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
23 - COMMUNICATIONS			3. NUMBER REQUIRED FOR DISPATCH	
			4. REMARKS OR EXCEPTIONS	
4.	Crewmember Interphone System			
	1) Passenger Configuration (Cont'd)			
	c) Flight Deck to Ground Function (Includes CALL functions)			
	(1) Large	C	1	0
	Turbojet			(O) Flight interphone flight deck to ground/ground to flight deck function may be inoperative provided:
	Airplanes			a) Alternate procedures are established and used, and
	Operating			b) Nose gear/forward fuselage service interphone jack operates normally.
	Under 14			
	CFR 121			
		C	1	0
				(O) Service interphone flight deck to ground/ground to flight deck function may be inoperative provided:
				a) Alternate procedures are established and used, and
				b) Nose gear/forward fuselage flight interphone jack operates normally.
		B	-	0
				(O) May be inoperative provided alternate procedures are established and used.
	(2) All Other	C	-	0
	Aircraft/ Operations			(O) May be inoperative provided alternate procedures are established and used.
		D	-	0
				May be inoperative provided procedures do not require its use.
				(Continued)

U.S. DEPARTMENT OF TRANSPORTATION					
FEDERAL AVIATION ADMINISTRATION			MASTER MINIMUM EQUIPMENT LIST		
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010	PAGE: 23-5	
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH	
23 - COMMUNICATIONS				4. REMARKS OR EXCEPTIONS	
4.	Crewmember Interphone System (Cont'd)				
	2) Cargo Configuration				
	a) Flight Deck to Cabin, Cabin to Flight Deck Functions	C	-	0	(O) May be inoperative provided alternate, normal and emergency procedures and/or operating restrictions are established and used.
		D	-	0	May be inoperative provided procedures do not require its use.
	b) Cabin to Cabin Function	D	-	0	
	c) Flight Deck to Ground Function (Includes CALL functions)				
	(1) Large Turbojet Airplanes Operating Under 14 CFR 121	C	1	0	(O) Flight interphone flight deck to ground/ground to flight deck function may be inoperative provided: a) Alternate procedures are established and used, and b) Nose gear/forward fuselage service interphone jack operates normally.
		C	1	0	(O) Service interphone flight deck to ground/ground to flight deck function may be inoperative provided: a) Alternate procedures are established and used, and b) Nose gear/forward fuselage flight interphone jack operates normally.
		B	-	0	(O) May be inoperative provided alternate procedures are established and used.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-6

1. SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	3. NUMBER INSTALLED	
				3. NUMBER REQUIRED FOR DISPATCH	
					4. REMARKS OR EXCEPTIONS
23 - COMMUNICATIONS					
4.	Crewmember Interphone System (Cont'd)				
	2) Cargo Configuration				
	c) Flight Deck to Ground Function (Includes CALL functions) (Cont'd)				
	(2) All Other Aircraft/ Operations	C	-	0	(O) May be inoperative provided alternate procedures are established and used.
		D	-	0	May be inoperative provided procedures do not require its use.
5.	Cabin Attendant(s) Inter-Cabin Phone System				Deleted prior to Revision 27, relief incorporated into Item 23-4.
6. ***	Selective Call System (SELCAL)	C	1	0	(O) May be inoperative provided alternate procedures are established and used.
		D	1	0	May be inoperative provided procedures do not require its use.
	1) Channels	C	-	0	(O) May be inoperative provided alternate procedures are established and used.
		D	-	0	May be inoperative provided procedures do not require its use.
7.	Flight Interphone System				
	1) Flight Deck Intercom				Deleted by Revision 33. Relief incorporated into Item 25-11.
	2) Flight Deck to Ground				Deleted by Revision 45, relief incorporated into Item 23-4.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-7

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
23 - COMMUNICATIONS							
8.		Forward Observer's Audio Selector Panel					Deleted revision 33, relief incorporated into Item 25-11.
9.		ACARS System	C	1	0		(O) May be inoperative provided alternate procedures are established and used.
***			D	1	0		May be inoperative provided procedures do not require its use.
		1) ACARS Printer	D	-	0		NOTE: Any portion of system that operates normally may be used.
		2) FMC Interface Function	C	-	0		(O) May be inoperative provided alternate procedures are established and used.
			D	1	0		May be inoperative provided procedures do not require its use.
							NOTE: Any portion of system that operates normally may be used.
10.		Cockpit Voice Recorder System (CVR)					
		1) Aircraft without Recorder Independent Power Supply (RIPS)	A	1	0		May be inoperative provided: a) Flight Data Recorder (FDR) operates normally, and b) Repairs are made within three flight days.
***		2) Aircraft with Recorder Independent Power Supply (RIPS)	A	1	0		(M) May be inoperative provided: a) Flight Data Recorder (FDR) operates normally, b) RIPS circuit breaker is pulled and collared, c) A 15 minute interval after the pulling of the c/b is achieved before departure, and d) Repairs are made within three flight days.
							NOTE: CVR is inoperative with the RIPS c/b pulled and collared.
		a) Recorder Independent Power Supply (RIPS)	C	1	0		(M) May be inoperative provided: a) CVR operates normally, and b) RIPS battery is removed.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-8

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
23 - COMMUNICATIONS							
11. ***	High Frequency (HF) Communication System (Includes STC's ST02959AT and ST01837LA)	D	-	-	-	-	Any in excess of those required by 14 CFR may be inoperative.
		C	-	1			(O) May be inoperative while conducting operations that require two LRCS provided: a) SATCOM Voice or Data Link operates normally, b) Alternate procedures are established and used, c) SATCOM coverage is available over intended route of flight, and d) If Inmarsat Codes are not available while using SATCOM voice, prior coordination with appropriate ATS facility is required. NOTE: SATCOM is to be used only as a backup to normal HF communications unless otherwise authorized by appropriate ATS facility.
12. ***	Emergency Locator Transmitter (ELT)						
	1) Survival Type ELTs	D	-	-	-	-	Any in excess of those required by 14 CFR may be inoperative or missing.
	2) Fixed ELTs						
	a) Required by 14 CFR	A	-	0			(M) May be inoperative provided: a) System is deactivated, and b) Repairs are made within 90 days
		A	-	0			May be missing provided repairs are made within 90 days
	b) Not Required by 14 CFR	D	-	-	-	-	(M) Any in excess of those required by 14 CFR may be inoperative provided system is deactivated.
		D	-	-	-	-	Any in excess of those required by 14 CFR may be missing.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-9

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			4. REMARKS OR EXCEPTIONS
23 - COMMUNICATIONS						
13.	Flight Crew Audio Selector/Control Panels	A	2	1		(O) Either Captain's or First Officer's audio control panel may be inoperative provided: a) Optional AUDIO transfer switch is installed and operates normally, b) Primary observer's audio control panel is located on aft electronics panel and operates normally, and c) Repairs are made within two flight days.
***	1) AUDIO Transfer Switch	C	1	0		
14.	Headsets/Headphones	D	-	-		Any in excess of those required by 14 CFR may be inoperative or missing.
	1) Headset Boom Microphones					
	a) Cockpit Voice Recorder Equipped to Record Boom Microphone	A	-	0		May be inoperative provided: a) Flight Data Recorder (FDR) operates normally, b) Associated hand microphone is installed and operates normally, and c) Repairs are made within three flight days.
***	b) Cockpit Voice Recorder Not Equipped to Record Boom Microphone	D	-	0		Any in excess of those required by 14 CFR may be inoperative.
	2) Headset Earphones/Headphones	C	-	1		Either Captain's of First Officer's headset earphones/headphones may be inoperative provided associated flight deck speaker operates normally.
15.	Pre-recorded Passenger Announcement System	C	1	0		(O) May be inoperative provided alternate procedures are established and used.
***		D	1	0		May be inoperative provided procedures do not require its use.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-10

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		
				3.	NUMBER REQUIRED FOR DISPATCH	
					4.	REMARKS OR EXCEPTIONS
23 - COMMUNICATIONS						
16.	Push-To-Talk (PTT) Switches					
	1) Control Wheel PTT Switches	C	2	1	(M) One may be inoperative provided: a) Associated audio selector panel PTT switch operates normally, and b) Affected switch is either verified failed open or is deactivated.	
	2) Flight Crew Audio Selector Panel PTT Switches	C	2	1	(M) One may be inoperative provided: a) Associated Control Wheel PTT switch operates normally, and b) Affected switch is verified failed open.	
***	3) Glareshield Panel PTT Switch(es)	C	-	0	(M) May be inoperative provided affected switch is either verified failed open or is deactivated.	
		D	-	0	(M) May be inoperative provided: a) Affected switch is either verified failed open or is deactivated, and b) Procedures do not require its use.	
***	4) Pendant Switch(es)	C	-	0	(M) May be inoperative provided affected switch is either verified failed open or is deactivated.	
		D	-	0	(M) May be inoperative provided: a) Affected switch is either verified failed open or is deactivated, and b) Procedures are not based on its use.	
17.	Flight Deck Hand Microphones	C	-	0	May be inoperative or missing provided associated boom microphone operates normally.	
		D	-	0	Any in excess of those required for each person on flight deck duty may be inoperative or missing.	
18.	Satellite Communication System (SATCOM)	C	1	0	(O) May be inoperative provided alternate procedures are established and used.	
***		D	1	0	May be inoperative provided procedures do not require its use.	

U.S. DEPARTMENT OF TRANSPORTATION				
FEDERAL AVIATION ADMINISTRATION			MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010	PAGE: 23-11
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	
23 - COMMUNICATIONS			3. NUMBER REQUIRED FOR DISPATCH	
			4. REMARKS OR EXCEPTIONS	
19.	Alerting System (Audio/Visual)			
	1) Passenger Configuration			
	a) Flight Deck Call Visual Alerting System	B	1	0
				May be inoperative provided flight deck audio alerting system operates normally. NOTE: Flight deck audio alerting system must always be operative.
	b) Flight Attendant Visual Alerting System	B	1	0
				(O) May be inoperative provided: a) PA system operates normally, b) If affected visual alerting system is used for lavatory smoke detector alerting, an alternate lavatory smoke detector alert (visual or audio) is installed and operates normally, and c) Alternate procedures for contacting flight attendants are established and used. NOTE 1: Passenger to Attendant Call System is considered Non-Essential Equipment and Furnishing (NEF) item. NOTE 2: Any visual alerting system function(s) that operates normally may be used.
	c) Flight Attendant Audio Alerting System	B	1	0
				(O) May be inoperative provided: a) PA system operates normally, b) If affected audio alerting system is used for lavatory smoke detector alerting, an alternate lavatory smoke detector alert (visual or audio) is installed and operates normally, and c) Alternate procedures for contacting flight attendants are established and used. NOTE 1: Passenger to Attendant Call System is considered Non-Essential Equipment and Furnishing (NEF) item. NOTE 2: Any audio alerting system function(s) that operates normally may be used.
(Continued)				

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-12

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
23 - COMMUNICATIONS			3. NUMBER REQUIRED FOR DISPATCH	
			4. REMARKS OR EXCEPTIONS	
19. Alerting System (Audio/Visual) (Cont'd)				
2) Cargo Configuration				
a) Flight Deck Call Visual Alerting System	B	1	0	May be inoperative provided flight deck audio alerting system operates normally.
b) Flight Deck Call System	D	1	0	May be inoperative provided Courier/Supernumerary compartment remains unoccupied.
c) Courier/ Supernumerary Visual Alerting System	B	1	0	(O) May be inoperative provided: a) Courier/Supernumerary address system operates normally, and b) Alternate procedures are established and used.
	D	1	0	May be inoperative provided Courier/Supernumerary compartment remains unoccupied. NOTE: Any visual alerting system function(s) that operates normally may be used.
d) Courier/ Supernumerary Audio Alerting System	B	1	0	((O) May be inoperative provided: a) Courier/Supernumerary address system operates normally, and b) Alternate procedures are established and used.
	D	1	0	May be inoperative provided Courier/Supernumerary compartment remains unoccupied. NOTE: Any audio alerting system function(s) that operates normally may be used.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-13

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

23 - COMMUNICATIONS

20.

Handset Systems

1) Passenger
Configuration

a) Flight Deck

C

1

0

(O) May be inoperative provided:

- a) Flight deck to cabin communication operates normally, and
- b) Alternate procedures are established and used.

D

1

0

May be inoperative provided procedures do not require its use.

b) Cabin

B

-

-

(O) May be inoperative provided:

- a) Fifty percent of cabin handsets operate normally, and
- b) Alternate communication procedures between affected flight attendant station(s) are established and used.

NOTE 1: An operative handset at an inoperative flight attendant seat shall not be counted to satisfy fifty percent requirement.

NOTE 2: Any handsets functions that operate normally may be used.

2) Cargo
Configuration

a) Flight Deck

C

1

0

(O) May be inoperative provided flight deck to courier/ supernumerary communication operates normally.

D

1

0

May be inoperative provided procedures do not require its use.

b) Courier/
Supernumerary

D

-

1

D

-

0

May be inoperative provided Courier/Supernumerary compartment remains unoccupied.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-14

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		
				3.	NUMBER REQUIRED FOR DISPATCH	
					4.	REMARKS OR EXCEPTIONS
23 - COMMUNICATIONS						
21. ***	Electronic Visual Surveillance Systems (All Installed Systems)	A	1	0	(O) May be inoperative and components may be missing provided: a. Alternate procedures are established and used, and b. Repairs are made within three flight days. NOTE: Any portion of the system which operates normally may be used.	
		C	1	0	(O) May be inoperative and components may be missing provided: a. The flight deck door viewing port is installed and operates normally, and b. Alternate procedures are established and used. NOTE: Any portion of the system which operates normally may be used.	
		D	1	0	May be inoperative and components may be missing provided procedures do not require its use.	
	1) All Cargo Configuration	C	1	0	May be inoperative provided courier/supernumerary compartment remains empty.	
		D	1	0	May be inoperative and components may be missing provided procedures do not require its use.	
22. ***	Electronic Voice Checklist	C	1	0	(O) May be inoperative provided alternate procedures are established and used.	
23. ***	Multipurpose Interactive Display Unit (MIDU)	C	1	0	(O) May be inoperative provided alternate procedures are established and used for affected subsystems.	
24. ***	Landscape Camera System (-800EF STC ST02000NY)	D	1	0		
	1) Dome Camera	D	1	0	(M) May be inoperative or missing.	

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION					MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : DATE:		54a 05/12/2010	PAGE: 23-15
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH		
				4. REMARKS OR EXCEPTIONS		
23 - COMMUNICATIONS						
25. ***	Automated Flight Information Reporting System (AFIRS) (STC's ST10345SC and ST02361NY)	C	1	0	(O) May be inoperative provided alternate procedures are established and used. NOTE: Any portion of system that operates normally may be used.	
		D	1	0	May be inoperative provided procedures do not require its use. NOTE: Any portion of system that operates normally may be used.	
***	1) Global Voice SATCOM (ST02361NY)	C	1	0	(O) May be inoperative provided alternate procedures are established and used.	
		D	1	0	May be inoperative provided procedures do not require its use.	
	a) Cockpit Dialer Pad	C	1	0	(O) May be inoperative provided alternate procedures are established and used.	
		D	1	0	May be inoperative provided procedures do not require its use.	
	b) Flt Compt. Handset	C	1	0	(O) May be inoperative provided alternate procedures are established and used.	
		D	1	0	May be inoperative provided procedures do not require its use.	
	c) Pax Compt. Handset	C	1	0	(O) May be inoperative provided alternate procedures are established and used.	
		D	1	0	May be inoperative provided procedures do not require its use.	
***	2) Global Messaging (ST02361NY)	C	1	0	(O) May be inoperative provided alternate procedures are established and used.	
		D	1	0	May be inoperative provided procedures do not require its use. NOTE: Any portion of the system that operates normally may be used.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

23-16

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

23 - COMMUNICATIONS

26.
***Avionica
secureLINK
Airborne Wireless
Router
(STC03151AT)

D

1

0

May be inoperative provided procedures do not require
its use.

NOTE: Any mode that operates normally may be used.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737	REVISION NO : 53 DATE: 08/01/2009	PAGE: 24-1
---------------------------	--	----------------------

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
24 - ELECTRICAL POWER					
1.	Engine Driven Generator Systems				
	1) (-100/-200/-300/-400/-500)	B	2	1	(M)(O) Except for ER operations, may be inoperative provided: a) APU generator operates normally and is used throughout flight, and b) An APU fuel heater is installed.
		B	2	1	(M)(O) Except for ER operations, may be inoperative provided: a) APU generator operates normally and is used throughout flight, and b) Fuel temperature is maintained at or above 32 degrees F (0 degrees C).
	2) (-600/-700/-800/-900)	B	2	1	(M)(O) Except for ER operations, may be inoperative provided APU generator operates normally and is used throughout flight.
	3) (-700/-800 with APU serial numbers P-7534 and lower, or P-7638 and Higher; or upon incorporation of Honeywell Service Bulletin 131-49-7949, or Production Equivalent)				DELETED REVISION 53
	4) (-700/-800 with APU serial numbers P-7535 through P-7637 prior to incorporation of Honeywell Service Bulletin 131-49-7949)				DELETED REVISION 53
2.	APU Generator System	C	1	0	Except for ER operations, may be inoperative.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

24-2

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3.	NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
24 - ELECTRICAL POWER								
3.	Engine Driven Generator LOW OIL PRESSURE/DRIVE Lights							
	1) (-100/-200/-300/-400/-500)	C	2	0				LOW OIL PRESSURE/DRIVE lights and associated generator low oil pressure switches may be inoperative provided associated HIGH OIL TEMP light and oil temperature indicator operate normally.
	2) (-600/-700/-800/-900)	C	2	0				DRIVE lights and associated generator low oil pressure switches may be inoperative.
4.	Engine Driven Generator Oil Temperature Indicator Systems (-100/-200/-300/-400/-500)	C	2	0				May be inoperative provided associated LOW OIL PRESSURE/DRIVE light and HIGH OIL TEMP light operate normally.
5.	Engine Driven Generator HIGH OIL TEMP lights (-100/-200/-300/-400/-500)	C	2	0				May be inoperative provided associated LOW OIL PRESSURE/DRIVE light and oil temperature indicator operate normally.
6.	Transformer Rectifiers							
	1) No. 2 TR (-100/-200)	B	1	0				Except for ER operations, may be inoperative provided: <ul style="list-style-type: none"> a) All DC busses and all generators (including APU generator) operate normally, and b) APU generator can be electrically connected to either bus.
7.	Frequency Meter	C	1	0				
8.	AC Volts Indication	B	1	0				(O) May be inoperative except in STBY PWR position provided Standby Power Test is accomplished.
	1) Residual Voltage Function (-100/-200/-300/-400/-500)	C	1	0				
9.	AC Ammeters	C	-	0				May be inoperative provided associated generator off bus lights operate normally.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT:			REVISION NO :		53		PAGE:
B-737			DATE:		08/01/2009		24-3
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED				
			3. NUMBER REQUIRED FOR DISPATCH				
			4. REMARKS OR EXCEPTIONS				
24 - ELECTRICAL POWER							
10.	Generator System Annunciator Panel (-100/ -200/-300/ -400/-500)	C	1	0			
11.	External Power System	C	1	0		NOTE: Any portion of system which operates normally may be used.	
***	1) DC Receptacle	D	1	0			
12.	GEN OFF BUS Lights	C	2	1		One may be inoperative provided associated generator AC ammeter operates normally.	
13.***	Galley Load Shed Sensor Module (-300/-400/-500)	C	1	0		May be inoperative provided GALLEY Power Switch remains OFF when APU is being used to power both generator busses on ground.	
14.***	BAT DISCHARGE Light	C	1	0			
15.***	TR UNIT Light	C	1	0			
16.***	ELEC Light						
	1) (-300/-400/-500)	C	1	0		(O) May be inoperative provided: a) Standby Power Test is accomplished, and b) Battery Charger is verified to operate normally.	
	2) (-600/-700/-800/ -900)	C	1	0		(O) May be inoperative provided: a) Standby Power Test is accomplished once each flight day, and b) Battery Charger is verified to operate normally.	
17.	DC Ammeter Indication	B	1	0		(O) May be inoperative provided: a) BAT position operates normally, b) Standby Power Test is accomplished, and c) Procedures do not require its use.	
18.	DC Volts Indication	B	1	0		(O) May be inoperative except in STBY PWR position provided Standby Power Test is accomplished.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

24-4

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
24 - ELECTRICAL POWER						
19.	APU GEN OFF BUS Light	C	1	0		May be inoperative provided: a) APU frequency meter operates normally, and b) APU ammeter operates normally.
20. ***	Cabin Power Switch (Jet Aviation Engineering Services, (JAES))	B	1	0		(M) May be inoperative provided procedures are established and used to deactivate cabin power.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

25-1

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS****25 - EQUIPMENT AND
FURNISHINGS**

1. Megaphones
(Includes STC's
SA2969SO, and
ST10238SC)

D

-

-

Any in excess of those required by 14 CFR may be inoperative or missing provided:

- a) Inoperative megaphone is removed from passenger cabin, and
- b) Required distribution is maintained.

NOTE: Not required for all-cargo operations.

2. Crewmember
Shoulder Harness
(Flight Deck)

Deleted Revision 33, relief incorporated into Item 25-11.

3. Flight Attendant
Seat Assembly
(Single or Dual
Position)

1) Required Flight
Attendant Seats

B

-

-

(M)(O) One seat position or assembly (dual position) may be inoperative provided:

- a) Affected seat or seat assembly is not occupied,
- b) Flight attendant(s) displaced by inoperative seat(s) occupies either an adjacent flight attendant seat or passenger seat which is most accessible to inoperative seat(s), so as to most effectively perform assign duties,
- c) Alternate procedures are established and used as published in crewmembers manuals,
- d) Folding type seat stows automatically or is secured in retracted position, and
- e) Passenger seat assigned to flight attendant is placarded "FOR FLIGHT ATTENDANT USE ONLY".

NOTE 1: An automatic folding seat that will not stow automatically is considered inoperative.

NOTE 2: A seat position with an inoperative or missing restraint system is considered inoperative.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION				FEDERAL AVIATION ADMINISTRATION		MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : DATE:		54a 05/12/2010		PAGE: 25-2
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED				
			3. NUMBER REQUIRED FOR DISPATCH				
			4. REMARKS OR EXCEPTIONS				
25 - EQUIPMENT AND FURNISHINGS							
3.	Flight Attendant Seat Assembly (Single or Dual Position)						
	1) Required Flight Attendant Seats (Cont'd)						NOTE 3: Individual operators when operating with inoperative seats, will consider locations and combinations of seats to ensure that proximity to exits and distribution requirements of applicable 14 CFR are met.
	2) Excess Flight Attendant Seats	C	-	-			NOTE 4: If one side of a dual seat assembly is inoperative and a flight attendant is displaced to adjacent seat, adjacent seat must operate normally. (M) May be inoperative provided: a) Affected seat position or seat assembly is not occupied, and b) Folding type seat stows automatically or is secured in retracted position.
	3) All Cargo Configuration	D	-	-			NOTE 1: An automatic folding seat that will not stow automatically is considered inoperative. NOTE 2: A seat position with an inoperative or missing restraint system is considered inoperative. May be inoperative provided affected seat or seat assembly is not occupied.
***	4) Seat Cushion Heating System	D	-	0			(M) May be inoperative provided heating system is deactivated.
4.	Cabin Window Shades	D	-	0			May be inoperative in a compartment used for cargo provided AFM Limitations are observed. NOTE: Passenger Cabin Window Shades in compartments configured for passengers only are considered a passenger convenience item.

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737	REVISION NO : DATE:	54a 05/12/2010	PAGE: 25-3
---------------------------	-------------------------------	-------------------	----------------------

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	
25 - EQUIPMENT AND FURNISHINGS			4. REMARKS OR EXCEPTIONS			
5.	Cargo Compartment Restraint Components	C	-	-	(M) May be inoperative or missing provided: a) Acceptable cargo loading limits from an approved source, i.e. an approved Cargo Loading Manual, Cargo Handling Manual, or Weight and Balance Document are observed, and b) Repairs are made prior to the completion of the next heavy maintenance visit.	
		C	-	-	May be inoperative or missing provided associated cargo compartment remains empty.	
		C	-	-	May be inoperative or missing provided pallet with inoperative lock(s) is removed.	
	1) Passenger Pallets (737C, -300 QC, and -700C)	C	-	-	(M) One lock per pallet may be inoperative provided: a) Three seats in group associated with lock are blocked by folding and securing backrests in a forward position, and b) If more than one lock is inoperative, pallet must be removed. NOTE: If a pallet lock cover is broken or missing, associated lock is considered inoperative.	
	2) Cargo Pallet Locks (Pemco 737 F/QC and COMBI)	C	-	-	(M)(O) May be inoperative or missing provided acceptable cargo loading limits from an approved source, i.e. an approved Cargo Loading Manual, Cargo Handling Manual, or Weight and Balance Document are observed.	
6.	Passenger Seat(s) (Includes STC ST10238SC)	D	-	-	May be inoperative provided: a) Seat does not block an Emergency Exit, b) Seat does not restrict any passenger from access to main aircraft aisle, and c) Affected seat(s) is blocked and placarded "DO NOT OCCUPY". NOTE 1: A seat with an inoperative seat belt is considered inoperative. NOTE 2: Inoperative seat(s) does not affect required number of Flight Attendants. NOTE 3: Affected seat(s) may include seat(s) behind and/or adjacent outboard seats.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

25-4

SYSTEM & SEQUENCE NUMBER		1.	2.		NUMBER INSTALLED	
		ITEM			3. NUMBER REQUIRED FOR DISPATCH	
25 - EQUIPMENT AND FURNISHINGS					4. REMARKS OR EXCEPTIONS	
					(Continued)	
6.	Passenger Seat(s) (Includes STC ST10238SC) (Cont'd)					
	1) Recline Mechanism	D	-	-	(M) May be inoperative and seat occupied provided seat is secured in up-right position.	
		D	-	-	May be inoperative and seat occupied provided seat back is immovable in full upright position.	
	2) Arm Rests					
	a) Armrest with Recline Mechanism	D	-	-	(M) May be inoperative or missing and seat occupied provided: a) Arm rest does not block an Emergency Exit, b) Arm rest does not restrict any passenger from access to main aircraft aisle, and c) If armrest is missing, seat is secured in full upright position.	
	b) Armrest without Recline Mechanism	D	-	-	May be inoperative or missing and seat occupied provided: a) Arm rest does not block an Emergency Exit, and b) Arm rest does not restrict any passenger from access to main aircraft aisle.	
	3) Underseat Baggage Restraining Bars	C	-	-	(O) May be inoperative provided: a) Baggage is not stowed under seat with inoperative restraining bar, b) Associated seat is placarded "DO NOT STOW BAGGAGE UNDER THIS SEAT", and c) Procedures are established to alert Cabin Crew of inoperative restraining bar.	

(M) May be inoperative and seat occupied provided seat is secured in up-right position.

May be inoperative and seat occupied provided seat back is immovable in full upright position.

(M) May be inoperative or missing and seat occupied provided:

- a) Arm rest does not block an Emergency Exit,
- b) Arm rest does not restrict any passenger from access to main aircraft aisle, and
- c) If armrest is missing, seat is secured in full upright position.

May be inoperative or missing and seat occupied provided:

- a) Arm rest does not block an Emergency Exit, and
- b) Arm rest does not restrict any passenger from access to main aircraft aisle.

(O) May be inoperative provided:

- a) Baggage is not stowed under seat with inoperative restraining bar,
- b) Associated seat is placarded "DO NOT STOW BAGGAGE UNDER THIS SEAT", and
- c) Procedures are established to alert Cabin Crew of inoperative restraining bar.

U.S. DEPARTMENT OF TRANSPORTATION			
FEDERAL AVIATION ADMINISTRATION		MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737		REVISION NO : 54a DATE: 05/12/2010	PAGE: 25-5
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2. NUMBER INSTALLED
25 - EQUIPMENT AND FURNISHINGS 4) Electrical/ Electronic Systems/ Components			3. NUMBER REQUIRED FOR DISPATCH 4. REMARKS OR EXCEPTIONS DELETED REV 49.
7. Second Observer Seat			Moved to Item 25-11 prior to Revision 30.
8. Flight Deck Door Lock Solenoid			Moved to Item 52-8 prior to Revision 30.
9. "Fasten Seat Belts While Seated" Signs or Placards	C	-	- One or more signs or placards may be illegible or missing provided a legible sign or placard is visible from each occupied passenger seat.
10. Non-Essential Equipment & Furnishings (NEF) ***		-	0 May be inoperative, damaged, or missing provided that item(s) is deferred in accordance with operator's NEF deferral program. NEF program, procedures, and processes must be outlined in operator's appropriate document. (M) and (O) procedures, if required, must be available to flight crew and included in operator's appropriate document.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

25-6

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**25 - EQUIPMENT AND
FURNISHINGSNOTE: Exterior lavatory door ash trays are not
considered NEF items.

11. Observer Seat(s)

1) Primary
Observer's Seat
(Including
Associated
Equipment)

A

1

0

May be inoperative provided:

- a) A passenger seat in passenger cabin is made available to an FAA inspector for performance of official duties, and
- b) Repairs are made within two flight days.

A

1

0

May be inoperative provided:

- a) Second observer's seat is available to an FAA inspector for performance of official duties, and
- b) Repairs are made within two flight days.

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737			REVISION NO : DATE:		54a 05/12/2010	PAGE: 25-7
SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		
25 - EQUIPMENT AND FURNISHINGS		A	1	0	3. NUMBER REQUIRED FOR DISPATCH	
					4. REMARKS OR EXCEPTIONS	
					May be inoperative provided: a) Required minimum safety equipment (safety belt and oxygen) is available, b) Seat is acceptable to an FAA inspector for performance of official duties, and c) Repairs are made within two flight days. NOTE 1: These provisos are intended to provide for occupancy of above seats by an FAA inspector when minimum safety equipment (oxygen and safety belt) is functional and inspector determines conditions to be acceptable. NOTE 2: Pilot-in-Command will determine if minimum safety equipment is functional for other persons authorized to occupy any observer seat(s).	
***	2) Second Observer's Seat (Including Associated Equipment)	D	1	0	NOTE: Pilot-in-Command will determine if minimum safety equipment is functional for other persons authorized to occupy any observer seat(s).	
***	3) Crotch Straps	C	-	0		
12. ***	Emergency Flashlight Holders/ Flashlights					
	1) Cabin	C	-	-	May be inoperative or missing provided crewmember assigned to affected position has a normally operating flashlight readily available.	
	2) Flight Deck	C	-	-	May be inoperative or missing provided crewmember assigned to affected position has a normally operating flashlight readily available.	
13 ***	Emergency Evacuation Signal System	C	1	0	(O) May be inoperative provided alternate procedures are established and used.	

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION					MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010		PAGE: 25-8	
SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		
				3.	NUMBER REQUIRED FOR DISPATCH	
					4. REMARKS OR EXCEPTIONS	
25 - EQUIPMENT AND FURNISHINGS		D	1	0	May be inoperative provided procedures do not require its use.	
14.	Main Deck Cargo 9G Barrier Net					
	1) (737F and QC) (PEMCO World Air Services, Inc.)	C	1	1	One net attachment, at any location, may be broken or missing provided maximum loading on main deck is reduced to 34,650 lbs. NOTE: Not required for all-passenger operations.	
	2) (-700C and -700 Combi)	C	1	0	In Cargo Mode, may be missing or net attachments may be broken or missing provided approved cargo loading limits in Weight and Balance Control and Loading Manual are observed. NOTE: Not required for all-passenger operations.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

25-9

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**25 - EQUIPMENT AND
FURNISHINGS

D

1

0

May be missing or net attachments may be broken or
missing provided associated cargo compartment
remains empty.

NOTE: Not required for all-passenger operations.

3) (STC
ST01566LA)

C

1

1

In Cargo Mode only one attachment may be broken or
missing provided:

- a) There are no visible defects on remaining net
fittings, and
- b) Maximum allowable load limits are observed.

D

1

0

May be missing or net attachments may be broken or
missing provided associated cargo compartment
remains empty.

NOTE: Not required for all-passenger operations.

15. Heating Blankets

Moved to Item 21-41 in Revision 33.

16. Lower Cargo
Compartment Lining
Panels and Floor
Panels

C

-

-

(M)(O) May be damaged or missing provided
procedures are established and used to ensure
associated compartment remains empty, or is verified
to contain only empty cargo handling equipment,
ballast (ballast may be loaded in ULDs), and/or Fly
Away Kits.NOTE: Operator MELs must define which items are
approved for inclusion in Fly Away Kits, and
which materials can be used as ballast.17. Emergency Medical
Equipment
(Includes STC
ST10238SC)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

25-10

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
			3.	NUMBER REQUIRED FOR DISPATCH
25 - EQUIPMENT AND FURNISHINGS				4. REMARKS OR EXCEPTIONS
1) First Aid Kit (FAK) and/or Associated Equipment	A	-	-	(O) If more than one is required by 14 CFR, only one required first aid kit may be incomplete, missing or inoperative provided: a) FAK is resealed in a manner that will identify it as a unit that can not be mistaken for a fully serviceable unit, and b) Repairs or replacements are made with-in 3 flight cycles.
	D	-	-	Any in excess of those required by 14 CFR may be incomplete, missing or inoperative.
	A	-	0	(O) May be incomplete, missing or inoperative provided: a) EMK is resealed in a manner that will identify it as a unit that can not be mistaken for a fully serviceable unit, and b) Repairs or replacements are made with-in 3 flight cycles.
	D	-	-	Any in excess of those required by 14 CFR may be incomplete, missing or inoperative.
3) Augmented Emergency Medical Kit				DELETED REVISION 46. (Continued)
17. Emergency Medical Equipment (Includes STC ST10238SC) (Cont'd)				
4) Automatic External Defibrillators (AED) and/or Associated Equipment	A	-	0	(O) May be incomplete, missing or inoperative provided: a) AED is resealed in a manner that will identify it as a unit that can not be mistaken for a fully serviceable unit, and b) Repairs or replacements are made with-in 3 flight cycles.
	D	-	-	Any in excess of those required by 14 CFR may be incomplete, missing, or inoperative.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT: B-737			REVISION NO : DATE:		54a 05/12/2010		PAGE: 25-11
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED		
					3.	NUMBER REQUIRED FOR DISPATCH	
						4.	REMARKS OR EXCEPTIONS
25 - EQUIPMENT AND FURNISHINGS							
18.	Flotation Equipment (Crew and Passengers)	D	-	-	-		Any in excess of those required by 14 CFR may be inoperative or missing provided required distribution is maintained.
19.	Underseat Baggage Restraining Bars						Moved to item 25-6 in Revision 39.
20.	Exterior Lavatory Door Ashtrays						
	1) Airplanes With More Than One Exterior Lavatory Door Ashtray Installed	A	-	-	-		One may be missing provided it is replaced within 10 calendar days.
	2) Airplanes With Only One Exterior Lavatory Door Ashtray Installed	A	1	0	0		May be missing provided it is replaced within 3 calendar days.
21.	Flight Crew Seats						
	1) Recline Mechanism	A	2	0	0		(M) May be inoperative provided: a) Seat is secured in a position acceptable to affected crewmember, and b) Repairs are made within two flight days
	2) Vertical Adjustment	A	2	0	0		(M) May be inoperative provided: a) Seat is secured in a position acceptable to affected crewmember, and b) Repairs are made within two flight days.
	3) Armrests	B	4	0	0		(M) May be inoperative in up position or removed provided seat is acceptable to affected crewmember.
	4) Lumbar/Thigh Supports	C	4	0	0		May be inoperative provided seat is acceptable to affected crewmember.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

25-12

SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3. NUMBER REQUIRED FOR DISPATCH	
25 - EQUIPMENT AND FURNISHINGS					4. REMARKS OR EXCEPTIONS
***	5) Headrests	C	2	0	May be inoperative or missing provided seat is acceptable to affected crewmember.
22.	Galley/Lavatory Waste Receptacle Access Doors/ Covers				
	1) Galley Waste Receptacle Access Doors/ Covers	C	-	-	(M)(O) May be inoperative provided: a) Associated container is empty, b) Container access is secured to prevent waste introduction into compartment, and c) Procedures are established to ensure that sufficient galley/lavatory waste receptacles are available to accommodate all waste that may be generated during flight.
22.	Galley/Lavatory Waste Receptacle Access Doors/ Covers				
	2) Lavatory Waste Receptacle Access Doors/ Covers	C	-	-	(M)(O) May be inoperative provided: a) Associated container is empty, b) Container access is secured to prevent waste introduction into compartment, and c) Lavatory is used only by crewmembers, and d) Associated lavatory entrance door is locked closed and placarded: INOPERATIVE – DO NOT ENTER.
					NOTE: These provisions are not intended to prohibit lavatory use or inspection by crewmembers.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

25-13

SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
25 - EQUIPMENT AND FURNISHINGS				3.	NUMBER REQUIRED FOR DISPATCH
				4.	REMARKS OR EXCEPTIONS
23. ***	Automatic Cargo Loading Systems	D	-	0	NOTE: Any portion of system(s) that operates normally may be used.
24. ***	Overhead Storage Bin(s)/Cabin and Galley Storage	C	-	-	(M) May be inoperative provided:
					a) Procedures are established to secure compartment CLOSED, b) Associated bin or compartment is prominently placarded DO NOT USE. c) Any emergency equipment located in affected compartment is considered inoperative, and d) Affected compartment is not used for storage of any item(s) except for those permanently affixed.
					NOTE: If no partitions are installed, entire overhead storage compartment is considered one bin or compartment.
					(Continued)
24. ***	Overhead Storage Bin(s)/Cabin and Galley Storage (Cont'd)	C	-	-	.
					(M)(O) May be inoperative provided: a) Affected door(s) is removed, b) Associated bin or compartment is not used for storage of any item(s) except for those permanently affixed, c) Associated bin or compartment is prominently placarded DO NOT USE, d) Procedures are established and used to alert crew members and passengers of inoperative bins, and e) Passengers are briefed that associated bin or compartment is not used.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

25-14

SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3. NUMBER REQUIRED FOR DISPATCH	
25 - EQUIPMENT AND FURNISHINGS					4. REMARKS OR EXCEPTIONS
				<p>NOTE 1: If no partitions are installed, entire overhead storage compartment is considered one bin or compartment.</p> <p>NOTE 2: Any emergency equipment located in the associated compartment (permanently affixed) is available for use.</p>	
1) Multi Latch/Quarter-Turn Lug Installations	C	-	-	<p>One latch/lug per compartment may be inoperative provided:</p> <p>a) Remaining latch(es)/lug(s) on affected compartments operate normally, and</p> <p>b) If affected compartment is used for a galley cart, cart remains empty.</p>	
*** 2) Storage Compartment Key Locks	D	-	0	(M) May be inoperative in the unlocked position provided doors can be secured by other means.	
25. *** Beds (Electrical Operation) (Jet Aviation Engineering Services, (JAES))	C	-	0	May be inoperative provided manual override system operates normally.	
26. *** Tables (Electrical Operation) (Jet Aviation Engineering Services, (JAES))	C	-	0	May be inoperative provided manual override system operates normally.	
	C	-	0	May be inoperative provided seats at associated inoperative table are not occupied.	
27. *** Crash Pads (Jet Aviation Engineering Services, (JAES))	C	-	0	May be inoperative or missing provided associated seat, adjacent to crash pad is not occupied.	
28. *** Emergency Vision Assurance System (EVAS) (STC SA00892LA)	C	2	0		

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

25-15

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**25 - EQUIPMENT AND
FURNISHINGS29.
***Secondary Door
Barrier (Flight Deck
Security)

C

1

0

(O) May be inoperative provided:
a) Barrier remains in retracted position, and
b) Alternate procedures are established and
used.

C

1

0

(M)(O) May be inoperative provided:
a) Barrier is removed, and
b) Alternate procedures are established and
used.

D

1

0

May be inoperative provided procedures do not require
its use.30.
***Security Kit and/or
Associated
Equipment

D

-

0

May be inoperative, missing, or have missing
equipment.

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010		PAGE: 26-1
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
26 - FIRE PROTECTION					
1.	Engine and APU Fire Extinguisher Discharge Lights	C	3	0	
2.	Engine Overheat and Fire Detection Systems				
	1) Basic Systems (-100/-200)	C	4	2	(M) One overheat detection system or one fire detection system per engine may be inoperative provided operative system is tested and operates normally before each departure
	2) Dual Loop	C	4	2	(O) Except for ER operations beyond 120 minutes, one loop (A or B) per engine may be inoperative.
3.	Portable Fire Extinguishers	D	-	-	(M) Any in excess of those required by 14 CFR may be inoperative or missing provided: a) Inoperative fire extinguisher is tagged inoperative, removed from installed location, and placed out of sight so it cannot be mistaken for a functional unit, and b) Required distribution is maintained.
4.	Wheel Well Fire Detection System	C	1	0	(M) May be inoperative provided brake temperature monitoring system (BTMS) operates normally.
		C	1	0	(M)(O) May be inoperative provided brakes are inspected and are cool to touch before engine start.
NOTE: In case of engine failure after V ₁ , performance is prime consideration, and landing gear should be retracted normally until performance penalty with gear down is not a problem. Pilots must consider possibility of ice accumulation on gear associated with delayed raising of landing gear or lowering landing gear during winter operations.					

MASTER MINIMUM EQUIPMENT LIST

PAGE:

26-2

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	
					3.	NUMBER REQUIRED FOR DISPATCH
						4. REMARKS OR EXCEPTIONS
26 - FIRE PROTECTION						
5.	APU Fire Extinguisher Discharge Discs (-100/-200/-300/-400/-500)	C	2	0		(M) Discs may be missing provided indicator reading is checked to verify proper charge.
***	1) HTL Type	C	2	0		(M) Discs may be missing provided bottle integrity is verified by checking APU fire extinguisher bottle discharge light or weighing bottle once each flight day.
6.	APU Fire Shutoff System	C	1	0		(O) Except for ER operations, may be inoperative provided APU is not used.
7.	APU Fire Extinguisher System	C	1	0		(O) Except for ER operations, may be inoperative provided APU is not used.
8.	APU Fire Detection System					
	1) Single and Dual Loop	C	-	0		(O) Except for ER operations, may be inoperative provided APU is not used.
***	2) APU DET INOP Light	C	1	0		(O) May be inoperative extinguished provided: a) APU fire detection system operates normally, and b) A fire warning test is performed before each APU start.
	3) Dual Loop	C	2	1		(O) Except for ER operations beyond 120 minutes, one loop (A or B) may be inoperative.
	4) External Warning Horn/Warning Light	C	1	0		May be inoperative for ground operation provided flight deck APU Overheat/Fire Protection Panel is continuously monitored.
9.	Engine/APU Fire Extinguisher Test System (EXT TEST) (Squib Test)	C	3	0		(M) May be inoperative provided: a. Failure is verified to be in squib test circuit. b. Squib circuit is verified to operate normally once each flight day.
	1) APU Fire Extinguisher Squib Test Circuits (EXT TEST) (-300/-400/-500/-600/-700/-800/-900)	C	2	1		(O) May be inoperative provided remaining APU Squib test circuit is verified to operate normally once each flight day.
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010		PAGE: 26-3
SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED	
				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
26 - FIRE PROTECTION					
9.	Engine/APU Fire Extinguisher Test System (EXT TEST) (Squib Test) (Cont'd)				
	2) APU Squib Light	C	1	0	(O) Except for ER operations, may be inoperative provided APU is not used.
10.	Fire Warning Bell				
	1) Bell Cutout Switch (Overheat/Fire Protection Panel)	C	1	0	May be inoperative provided: a) Bell cutout function of both Master Fire Warning lights operates normally, and b) Fire Warning Bell operates normally.
	2) Bell Cutout Function of Master Fire Warning Light	C	2	1	May be inoperative provided: a) Bell cutout function switch operates normally, and b) Fire Warning Bell operates normally.
11.	Master Fire Warning Lights				Deleted prior to Revision 27.
12.	Wing-Body Overheat Detector System (Left)	C	1	0	(O) Except for ER operations, may be inoperative provided: a) Right pack and engine bleed is used for pressurization only, b) Use of APU is prohibited except for engine start, c) Isolation valve and left engine bleed valve remain closed for all operations except engine start, and d) Airplane is not operated in known or forecast icing conditions.
13.	Wing-Body Overheat Detector System (Right)	C	1	0	(O) Except for ER operations, may be inoperative provided: a) Left pack and left engine or APU bleed air is used for pressurization only, b) Isolation valve and right engine bleed valve remain closed for all operations except engine start, and c) Airplane is not operated in known or forecast icing conditions.

MASTER MINIMUM EQUIPMENT LIST

26-4

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST	
FEDERAL AVIATION ADMINISTRATION					
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010		PAGE: 26-5
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		
			3. NUMBER REQUIRED FOR DISPATCH		
			4. REMARKS OR EXCEPTIONS		
26 - FIRE PROTECTION					
14. ***	Main Deck Cargo Compartment Fire Detection/ Suppression Systems (737C/QC/-700C/-700 Combi, STCs ST01566LA, -400C ST00235BO, -400 Combi ST00248BO, SA2970SO, ST01827LA, ST00283AT, and ST01961SE) (Cont'd)				
	2) Fire Detection (-400C ST000235BO, -400 Combi ST00248BO) (Cont'd)				
	a) Cargo Fire Flight Deck Unit (CFFU) (Cont'd)				
	(3) FIRE Legend	C	1	0	May be inoperative provided master FIRE WARN lights and master fire warning bell are checked to operate normally before each departure.
	(4) System Self Test	C	1	0	May be inoperative provided master FIRE WARN lights and master fire warning bell are checked to operate normally before each departure.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737	REVISION NO : DATE:	54a 05/12/2010	PAGE: 26-7
---------------------------	-------------------------------	-------------------	----------------------

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
26 - FIRE PROTECTION					
14. ***		Main Deck Cargo Compartment Fire Detection/ Suppression Systems (737C/QC/-700C/-700 Combi, STCs ST01566LA, -400C ST00235BO, -400 Combi ST00248BO, SA2970SO, ST01827LA, ST00283AT, and ST01961SE) (Cont'd)			
		3) Fire Suppression System (-700C/-700 Combi, STC ST01566LA) (Cont'd)			
		b) MAIN SYS Light	C 1	0	May be inoperative in Passenger mode.
			C 1	0	(M) May be inoperative in Combi or Cargo mode provided: a) Failure is verified to be in light circuit, and b) System circuit is verified to operate normally once each flight day.
		4) Smoke Detectors (737C/QC/-700C/-700 Combi, STC's ST01566LA, ST00235BO, SA2970SO, -400 Combi ST00248BO, ST01827LA, and ST01961SE)	C -	0	May be inoperative provided Main Deck Cargo Compartment Fire Detection System is considered inoperative.
		a) (STC ST01566LA Only)	C 12	6	May be inoperative provided all detectors in opposite loop operate normally.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

26-8

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
26 - FIRE PROTECTION			3. NUMBER REQUIRED FOR DISPATCH	
			4. REMARKS OR EXCEPTIONS	
14. ***	Main Deck Cargo Compartment Fire Detection/ Suppression Systems (737C/QC/ -700C/-700 Combi, STCs ST01566LA, - 400C ST00235BO, - 400 Combi ST00248BO, SA2970SO, ST01827LA, ST00283AT, and ST01961SE) (Cont'd)			
	4) Smoke Detectors (737C/QC/ -700C/-700 Combi, STC's ST01566LA, ST00235BO, SA2970SO, -400 Combi ST00248BO, ST01827LA, and ST01961SE (Cont'd)			
	b) System Test Feature (737C/QC/ -700 Combi, and STC ST01566LA)	C	1	0
	c) System Power (Blue) Light (PEMCO Aeroplex, Inc.)(-300QC, -300F, STC SA2970SO)	C	1	0
			(M) May be inoperative provided smoke detectors operate normally. (Continued)	

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST	
FEDERAL AVIATION ADMINISTRATION					
AIRCRAFT:		REVISION NO :		54a	PAGE:
B-737		DATE:		05/12/2010	26-9
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2. NUMBER INSTALLED		
			3. NUMBER REQUIRED FOR DISPATCH		
			4. REMARKS OR EXCEPTIONS		
26 - FIRE PROTECTION					
14. ***	Main Deck Cargo Compartment Fire Detection/ Suppression Systems (737C/QC/-700C/-700 Combi, STCs ST01566LA, -400C ST00235BO, -400 Combi ST00248BO, SA2970SO, ST01827LA, ST00283AT, and ST01961SE) (Cont'd)				
	4) Smoke Detectors (737C/QC/-700C/-700 Combi, STC's ST01566LA, ST00235BO, SA2970SO, -400 Combi ST00248BO, ST01827LA, and ST01961SE) (Cont'd)				
	d) (STC ST00235BO Only)				
	(1) Smoke Detector Units	C	10	-	(O) Detector(s) may be inoperative provided no cargo is carried in affected zone.
	(2) Smoke Detector Loops	C	20	10	One loop in any detector may be inoperative.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

26-10

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3.

NUMBER REQUIRED FOR DISPATCH

4.

REMARKS OR EXCEPTIONS

26 - FIRE PROTECTION

14.

Main Deck Cargo
Compartment Fire
Detection/
Suppression
Systems (737C/QC/
-700C/-700 Combi,
STCs ST01566LA, -
400C ST00235BO, -
400 Combi
ST00248BO,
SA2970SO,
ST01827LA,
ST00283AT, and
ST01961SE)
(Cont'd)

4) Smoke Detectors
(737C/QC/
-700C/-700
Combi, STC's
ST01566LA,
ST00235BO,
SA2970SO, -400
Combi
ST00248BO,
ST01827LA, and
ST01961SE)
(Cont'd)

e) (STC
ST00248BO
Only)

(1) Smoke C 4 0
Detector
Units

(2) Smoke C 8 4
Detector
Loops

May be inoperative provided Main Deck Cargo
Compartment Fire Detection System is considered
inoperative.

One loop in any detector may be inoperative.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST	
FEDERAL AVIATION ADMINISTRATION					
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010		PAGE: 26-11
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		
26 - FIRE PROTECTION			3. NUMBER REQUIRED FOR DISPATCH		
14. ***			4. REMARKS OR EXCEPTIONS		
Main Deck Cargo Compartment Fire Detection/ Suppression Systems (737C/QC/-700C/-700 Combi, STCs ST01566LA, -400C ST00235BO, -400 Combi ST00248BO, SA2970SO, ST01827LA, ST00283AT, and ST01961SE) (Cont'd)					
4) Smoke Detectors (737C/QC/-700C/-700 Combi, STC's ST01566LA, ST00235BO, SA2970SO, -400 Combi ST00248BO, ST01827LA, and ST01961SE) (Cont'd)					
f) (STC ST01827LA Only)					
(1) -300		C	12	10	Two detectors may be inoperative provided: a) Inoperative detectors are not in adjacent locations, and b) Detector #1, most forward detector, operates normally.
(2) -400		C	12	10	(M) Two detectors may be inoperative provided they are not in adjacent locations.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

26-12

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED		
26 - FIRE PROTECTION				3. NUMBER REQUIRED FOR DISPATCH	
				4. REMARKS OR EXCEPTIONS	
14. ***	Main Deck Cargo Compartment Fire Detection/ Suppression Systems (737C/QC/ -700C/-700 Combi, STCs ST01566LA, - 400C ST00235BO, - 400 Combi ST00248BO, SA2970SO, ST01827LA, ST00283AT, and ST01961SE) (Cont'd) 4) Smoke Detectors (737C/QC/ -700C/-700 Combi, STC's ST01566LA, ST00235BO, SA2970SO, -400 Combi ST00248BO, ST01827LA, and ST01961SE) (Cont'd) g) (STC ST01961SE Only)				(O) One detector (bus A or B) may be inoperative provided remaining detectors are verified to operate normally before each departure. (M) May be inoperative provided: a) All CCP smoke detector lights operate normally, and b) System integrity is verified to operate normally before each departure. (M) May be inoperative provided: a) DET FAULT Light operates normally, and b) System integrity is verified to operate normally before each departure. (Continued)
	(1) Smoke Detectors	C	20	19	
	(2) DET FAULT Light	C	1	0	
	(3) CCP Smoke Detector Lights	C	20	0	

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST	
FEDERAL AVIATION ADMINISTRATION					
AIRCRAFT: B-737		REVISION NO : 54a DATE: 05/12/2010		PAGE: 26-13	
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
26 - FIRE PROTECTION				3.	NUMBER REQUIRED FOR DISPATCH
				4.	REMARKS OR EXCEPTIONS
14. ***	Main Deck Cargo Compartment Fire Detection/ Suppression Systems (737C/QC/ -700C/-700 Combi, STCs ST01566LA, - 400C ST00235BO, - 400 Combi ST00248BO, SA2970SO, ST01827LA, ST00283AT, and ST01961SE) (Cont'd)				
	4) Smoke Detectors (737C/QC/ -700C/-700 Combi, STC's ST01566LA, ST00235BO, SA2970SO, -400 Combi ST00248BO, ST01827LA, and ST01961SE) (Cont'd)				
	g) (STC ST01961SE Only) (Cont'd)				
	(4) CARGO Light	C	1	0	((M)May be inoperative provided: a) DEPRESS Light operates normally, b) All Class E shut-Off valves are verified to be closed after pressing the DEPRESS switch, and c) The forward outflow valve is verified to be open after pressing the DEPRESS switch.
	(2) DEPRESS Light	C	1	0	(M) May be inoperative provided: a) All Class E shut-Off valves are verified to be closed after pressing the DEPRESS switch, and b) The forward outflow valve is verified to be open after pressing the DEPRESS switch.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

26-14

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3.	4. REMARKS OR EXCEPTIONS
26 - FIRE PROTECTION							
15.	Lavatory Fire Extinguisher Systems						
	1) Passenger Configuration	C	-	0			For each lavatory, lavatory fire extinguisher system may be inoperative provided associated lavatory smoke detection system operates normally.
		C	-	0			(M)(O) For each lavatory, lavatory fire extinguisher system may be inoperative provided: a) Lavatory waste receptacle is empty, b) Associated lavatory door is locked closed and placarded: "INOPERATIVE – DO NOT ENTER", and c) Lavatory is used only by crewmembers. NOTE: These provisions are not intended to prohibit lavatory use or inspection by crewmembers.
	2) Cargo Configuration	D	-	0			
16.	Lavatory Smoke Detection System						
	1) Passenger Configuration	C	-	0			(M)(O) For each lavatory, lavatory smoke detection system may be inoperative provided: a) Lavatory waste receptacle is empty, b) Associated lavatory door is locked closed and placarded: "INOPERATIVE – DO NOT ENTER", and c) Lavatory is used only by crewmembers. NOTE: These provisions are not intended to prohibit lavatory use or inspection by crewmembers.
	2) Cargo Configuration	D	-	0			
***	3) Lavatory Smoke Detector SELF TEST Switch	C	-	0			(M) May be inoperative provided associated lavatory smoke detector is verified to operate normally.
***	4) Lavatory Smoke Detector TEST Switch on Flight Attendant's Panel	C	-	0			(M) May be inoperative provided each lavatory smoke detector is verified to operate normally.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT: B-737			REVISION NO : 54a		PAGE: 26-15		
			DATE: 05/12/2010				
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED		
					3. NUMBER REQUIRED FOR DISPATCH		
					4. REMARKS OR EXCEPTIONS		
26 - FIRE PROTECTION							
17. ***	Engine Fire Extinguisher Thermal/Discharge Discs (-100/-200)						
	1) Discharge (Yellows) Discs	C	2	0	(M) May be missing provided indicator readings or other acceptable means are used to verify adequate charge.		
	2) Thermal (Red) Discs	C	2	0	(M) May be missing provided indicator readings or other acceptable means are used to verify adequate charge.		
18.	Wing-Body Overheat Test System						
	1) Flight Deck Test Feature	C	1	0	(M) May be inoperative provided system integrity is verified by an acceptable procedure once each flight day.		
19. ***	Lower Cargo Compartment Fire Detection/Suppression Systems (All models and STC's)	C	-	0	(O) May be inoperative provided procedures are established and used to ensure associated compartment remains empty, or is verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits.		
					NOTE 1: Operator MELs must define which items are approved for inclusion in Fly Away Kits and which materials can be used as ballast.		
					NOTE 2: Class E cargo compartments require only installation of smoke or fire detection systems (not suppression)		
					(Continued)		

MASTER MINIMUM EQUIPMENT LIST

PAGE:

26-16

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED		3.	NUMBER REQUIRED FOR DISPATCH	4.	REMARKS OR EXCEPTIONS
26 - FIRE PROTECTION										
19.	Lower Cargo									
***	Compartment Fire									
	Detection/									
	Suppression									
	Systems (All models									
	and STC's) (Cont'd)									
	1) Fwd/Aft									
	Detection Loops									
***	a) Boeing	C	4	2						(O) One loop (A or B) in each compartment may be
	installed									inoperative provided opposite loop is checked to
	system, STC									operate normally.
	ST00749LA-D,									
	ST00763LA-D,									
	ST01184LA,									
	ST01674AT,									
	ST01424LA,									
	ST10153T,									
	ST01804LA,									
	ST01114WI									
	Only									
***	b) STC	C	-	2						(O) May be inoperative provided one loop in each
	ST00405LA-D									compartment is checked to operate normally.
	Only									
***	2) Extinguisher									
	Bottles									
***	a) No. 1 (STC	C	1	0						(O) May be inoperative provided associated cargo
	ST01424LA,									compartment remains empty.
	ST01457LA,									
	and									
	ST01804LA									
	Only)									
***	b) No. 2 (Boeing	C	1	0						(M)(O) Except for ER operations, may be inoperative
	installed									with cargo carried in compartment.
	system and									
	STC									
	ST01184LA									
	Only)									
(Continued)										

U.S. DEPARTMENT OF TRANSPORTATION									
FEDERAL AVIATION ADMINISTRATION					MASTER MINIMUM EQUIPMENT LIST				
AIRCRAFT:				REVISION NO :			54a		PAGE:
B-737				DATE:			05/12/2010		26-17
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2. NUMBER INSTALLED					
				3. NUMBER REQUIRED FOR DISPATCH					
							4. REMARKS OR EXCEPTIONS		
26 - FIRE PROTECTION									
19.	Lower Cargo								
***	Compartment Fire								
	Detection/								
	Suppression								
	Systems (All models								
	and STC's) (Cont'd)								
***	2) Extinguisher								
	Bottles (Cont'd)								
***	c) No. 2 (STC	C	1	0	(M)(O) May be inoperative provided associated cargo				
	ST01424LA				compartment remains empty.				
	ST01457LA,								
	and								
	ST01804LA								
	Only)								
***	d) No. LRD2	C	1	0	(O) Except for ER operations, may be inoperative with				
	(STC				cargo carried in compartment.				
	ST00405LA-D								
	Only)								
***	3) Squib Lights	C	2	0	(O) May be inoperative provided associated cargo				
	(STC				compartment remains empty.				
	ST01424LA, and								
	ST01457LA								
	Only)								
***	4) DISCH Light(s)								
***	a) Boeing	C	1	0	(M) May be inoperative provided associated				
	installed				extinguisher bottle(s) is verified to have an adequate				
	system, STC				charge once each flight day.				
	ST01184LA								
	and								
	ST00405LA-D								
	Only								
***	b) STC	C	2	0	May be inoperative provided associated compartment				
	ST01424LA,				remains empty.				
	ST01457LA,								
	and								
	ST01804LA								
	Only								
							(Continued)		

U.S. DEPARTMENT OF TRANSPORTATION					MASTER MINIMUM EQUIPMENT LIST	
FEDERAL AVIATION ADMINISTRATION						
AIRCRAFT:			REVISION NO :		54a	PAGE:
B-737			DATE:		05/12/2010	26-18
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
26 - FIRE PROTECTION						
19. ***	Lower Cargo Compartment Fire Detection/ Suppression Systems (All models and STC's) (Cont'd)					
***	5) Extinguisher Bottle Pressure Switch (Boeing installed system Only)	C	-	0		(M) May be inoperative provided associated extinguisher bottle(s) is verified to have an adequate charge once each flight day.
***	6) EXT Lights (FWD and AFT) (Boeing installed system, STC ST01184LA and ST00405LA-D Only)	C	2	0		(M) May be inoperative provided: a) Failure is verified to be in squib light circuit, and b) Squib circuit is verified to operate normally once each flight day.
***	7) Fault(s) Indicated by Illumination of MX Indicator (STC ST00511LA, ST00404LA-D, ST00740LA-D, ST00745LA-D, ST00751LA-D, and ST00990LA-D Only)	B	-	-		Dispatch with MX indicator illuminated is permitted provided green SYS OK indicator remains illuminated. NOTE: This is a fault tolerant system and unit will continue to perform its intended function as long as green SYS OK indicator remains illuminated.
	a) Display of FWD INOP and/or AFT INOP Message(s)	B	-	-		(O) May be displayed provided green SYS OK indicator remains illuminated and indicated cargo bay remains empty.
	b) Smoke Detector(s)	C	-	-		(O) One smoke detector may be inoperative in each compartment provided SYS OK indicator on CDU remains illuminated. NOTE: MX indicator on CDU will remain illuminated.
						(Continued)

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010		PAGE: 26-19		
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2. NUMBER INSTALLED			
26 - FIRE PROTECTION				3. NUMBER REQUIRED FOR DISPATCH			
				4. REMARKS OR EXCEPTIONS			
19. ***	Lower Cargo Compartment Fire Detection/ Suppression Systems (All models and STC's) (Cont'd)						
***	8) Control Panel ALARM OFF Switch (STC ST00749LA-D and ST00763LA-D Only)	C	1	0	(O) May be inoperative provided Fire Bell cutout switch silences Cargo Bay Fire Protection Fire Warning Bell.		
***	9) DET Lights (STC ST01674AT, and ST01114WI Only)	C	4	2	(O) One light in each compartment may be inoperative provided remaining loop in associated compartment is checked to operate normally before each departure.		
***	10) FAIL Lights (STC ST01674AT, and ST01114WI Only)	C	4	2	(O) One light in each compartment may be inoperative provided remaining loop in associated compartment is checked to operate normally before each departure.		
***	11) Smoke Detectors						
***	a) STC ST01674AT, and ST01114WI Only	C	-	-	(M) One detector in each detector enclosure may be inoperative provided remaining detector in associated detector enclosure is verified to operate normally before each departure.		
***	b) STC ST01424LA, and ST01804LA (-300) Only	C	12	6	(M) May be inoperative provided 2 FWD and 4 AFT in same loop are functional.		
***	c) STC ST01457LA, and ST01804LA (-400) Only	C	14	7	(M) May be inoperative provided 3 FWD and 4 AFT in same loop are functional.		
						(Continued)	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

26-20

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
26 - FIRE PROTECTION							
19.	Lower Cargo						
***	Compartment Fire						
	Detection/						
	Suppression						
	Systems (All models						
	and STC's) (Cont'd)						
***	11) Smoke Detectors						
	(Continued)						
***	d) STC	C	10	5		(M) May be inoperative provided 2 FWD and 3 AFT in	
	ST01804LA					same loop are functional.	
	(-200)						
***	12) Fault Panel (E &	D	1	0			
	E Compartment,						
	STCs						
	ST01674AT, and						
	ST01114WI						
	Only)						
***	13) DETECTOR	C	1	0		(O) May be inoperative provided the cargo fire TEST	
	FAULT Light					switch is used to check for faults in the cargo fire	
	(Boeing					detection and suppression system before each flight.	
	Installed						
	System Only)						
20.	Lower Cargo					Incorporated into item 26-19 in Revision 39.	
	Compartment Fire						
	Extinguisher						
	System						
21.	Cabin Configuration	C	2	0		(M) May be inoperative provided:	
	Test Panel CARGO/					a) EE Bay Mode Selector Switch is verified to be	
	PASSENGER					in appropriate position for intended airplane	
	Lights (-700C and					configuration before each departure, and	
	-700 Combi)					b) Passenger Oxygen Shutoff Valve is verified to	
						be in appropriate position for intended airplane	
						configuration before each departure.	
22.	Galley Fire	C	1	0		(M) May be inoperative provided procedures are	
***	Detection System					established and used to deactivate cooktop.	
	(Jet Aviation						
	Engineering						
	Services, (JAES)						

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

26-21

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

26 - FIRE PROTECTION

23.
***Galley Vent Fire
Extinguisher
System (STC
ST09977)

C

1

0

(M) May be inoperative provided procedures are
established and used to deactivate cooktop and vent
fan.24.
***Smoke Detectors
(Jet Aviation
Engineering
Services, (JAES))

C

-

0

(M) May be inoperative provided all equipment in
cabinet is deactivated off.1) Equipment
Cabinets

C

-

0

(O) May be inoperative provided associated cabin
compartment remains open and is continuously
monitored.2) Cabin
Compartments

MASTER MINIMUM EQUIPMENT LIST

PAGE:

27-1

SYSTEM & SEQUENCE NUMBER			1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
27 - FLIGHT CONTROLS						
1.	Stabilizer Main Electrical Trim Operating Light (-100/-200)	C	1	0		
2.	Takeoff Warning Horn system					Deleted prior to Revision 27.
3.	Wing trailing Edge Flap Position Indication System					
	1) Mechanical Asymmetry Protection (-100/-200)	C	1	1		(O) Left Flap position indication may be inoperative provided proper flap operation is verified prior to each takeoff.
4.	Leading Edge Flap/Slat Position Light Systems	C	2	1		(M) Forward panel lights or annunciator panel may be inoperative. If forward panel lights are inoperative, annunciator panel must be used to verify proper LED position, and a placard must be installed to indicate proper positions for flap configuration in use.
	1) Leading Edge Slat Indications (-100/-200)	C	6	5		(O) Indication lights on forward panel and in addition indication lights for one leading edge slat on overhead annunciator panel may be inoperative provided: <ul style="list-style-type: none"> a) Normal operation is verified by flight crew before each takeoff and landing, b) Maximum speed is limited to 300 KIAS at/below FL 200 or .65 Mach above FL 200, and c) All remaining indications on overhead annunciator panel operate normally.
	2) Leading Edge Slat Indications (-300/-500)	C	6	5		(M)(O) Indication lights on forward panel and in addition indication lights for one leading edge slat on overhead annunciator panel may be inoperative provided: <ul style="list-style-type: none"> a) Normal operation is verified by flight crew before each takeoff and landing, b) Maximum speed is limited to 300 KIAS at/below FL 200 or .65 Mach above FL 200, c) All remaining indications on overhead annunciator panel operate normally, and d) Stall warning operation of both systems is verified to operate normally.
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

27-2

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
			3.	NUMBER REQUIRED FOR DISPATCH
				4. REMARKS OR EXCEPTIONS
27 - FLIGHT CONTROLS				
4.	Leading Edge Flap/ Slat Position Light Systems (Cont'd)			
	3) Leading Edge Slat Indications (-400)	C 6	5	(M)(O) Indication lights on forward panel and in addition indication lights for one leading edge slat, except for slats 3 and 4, on overhead annunciator panel may be inoperative provided: a) Normal operation is verified by flight crew before each takeoff and landing, b) Maximum speed is limited to 300 KIAS at/below FL 200 or .65 Mach above FL 200, c) All remaining indications on overhead annunciator panel operate normally, and d) Stall warning operation of both systems is verified to operate normally.
	4) Leading Edge Slat Indications (-600/-700)	C 8	7	(M)(O) Indication lights on forward panel and in addition indication lights for one leading edge slat, except for slats 4 and 5, on overhead annunciator panel may be inoperative provided: a) Normal operation is verified by flight crew before each takeoff and landing, b) Maximum speed is limited to 300 KIAS at/below FL 200 or .65 Mach above FL 200, c) All remaining indications on overhead annunciator panel operate normally, and d) Stall warning operation of both systems is verified to operate normally.
	5) Leading Edge Slat Indications (-800)	C 8	7	(M)(O) Indication lights on forward panel and in addition indication lights for one leading edge slat, except for slats 3, 4, 5 and 6, on overhead annunciator panel may be inoperative provided: a) Normal operation is verified by flight crew before each takeoff and landing, b) Maximum speed is limited to 300 KIAS at/below FL 200 or .65 Mach above FL 200, c) All remaining indications on overhead annunciator panel operate normally, and d) Stall warning operation of both systems is verified to operate normally.
(Continued)				

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT: B-737			REVISION NO : 53 DATE: 08/01/2009		PAGE: 27-3		
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED		
27 - FLIGHT CONTROLS					3. NUMBER REQUIRED FOR DISPATCH		
					4. REMARKS OR EXCEPTIONS		
4.	Leading Edge Flap/ Slat Position Light Systems (Cont'd)						
	6) Leading Edge Slat Indications (-900)	C	8	7	(M)(O) Indication lights on forward panel and in addition indication lights for one leading edge slat, except for slats 2, 3, 4, 5, 6 and 7, on overhead annunciator panel may be inoperative provided: a) Normal operation is verified by flight crew before each takeoff and landing, b) Maximum speed is limited to 300 KIAS at/below FL 200 or .65 Mach above FL 200, c) All remaining indications on overhead annunciator panel operate normally, and d) Stall warning operation of both systems is verified to operate normally.		
5.	Flight Control Low Pressure Lights (A and B) Systems (-100/-200)	C	2	0	May be inoperative provided warning lights, hydraulic pressure and quantity indicators operate normally.		
6.	Mach Trim System				Moved to Item 22-5 prior to Revision 27.		
7. ***	Auto Speed Brake System						
	1) (All Models except -800 with Short Field Performance (SFP) Option and -900ER)	C	1	0	(M)(O) May be inoperative provided: a) System is deactivated, and b) Operations are conducted in accordance with AFM, and c) For models with Blended Winglet with Speed Brake Load Alleviation System, Speed Brake Load Alleviation System is considered inoperative.		
	2) (-800SFP and -900ER)	C	1	0	(M)(O) May be inoperative provided: a) System is deactivated, b) Appropriate performance adjustments are applied, and c) For models with Blended Winglet with Speed Brake Load Alleviation System, Speed Brake Load Alleviation System is considered inoperative.		

MASTER MINIMUM EQUIPMENT LIST

PAGE:

27-4

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	3.	NUMBER REQUIRED FOR DISPATCH	4.	REMARKS OR EXCEPTIONS
27 - FLIGHT CONTROLS									
8.	Flap Load Limiter System								
***	1) -100/-200	C	1	0					(M) May be inoperative provided: a) Flaps are verified to operate normally throughout their full range before each departure, and b) Flaps are not extended beyond Flaps 30 at gross weights above 98,000 lbs. (44,453 kg).
	2) -300/-400/-500	C	1	0					May be inoperative provided flaps are not extended beyond Flaps 30.
	3) -600	C	1	0					May be inoperative provided: a) Flaps are not extended beyond Flaps 30 at gross weights above 93,830 lbs. (42,560 kg), and b) Flaps are not extended beyond Flaps 15 at gross weights above 105,040 lbs. (47,645 kg).
	4) -700	C	1	0					May be inoperative provided: a) Flaps are not extended beyond Flaps 30 at gross weights above 93,480 lbs. (42,401 kg), and b) Flaps are not extended beyond Flaps 15 at gross weights above 104,403 lbs. (47,356 kg).
	5) -800 without Short Field Performance (SFP) Option	C	1	0					May be inoperative provided: a) Flaps are not extended beyond Flaps 30 at gross weights above 93,995 lbs. (42,635 kg), and b) Flaps are not extended beyond Flaps 15 at gross weights above 104,875 lbs. (47,570 kg).
	6) -800 with Short Field Performance (SFP) Option	C	1	0					May be inoperative provided: a) Flaps are not extended beyond Flaps 30 at gross weights above 95,800 lbs. (43,454 kg), b) Flaps are not extended beyond Flaps 15 at gross weights above 105,000 lbs. (47,627 kg), and c) Flaps are not extended beyond Flaps 10 at gross weights above 135,800 lb (61,597 kg).
(Continued)									

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737		REVISION NO : 53 DATE: 08/01/2009		PAGE: 27-5	
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
27 - FLIGHT CONTROLS					
8.	Flap Load Limiter System (Cont'd)				
	7) -900	C	1	0	May be inoperative provided: a) Flaps are not extended beyond Flaps 30 at gross weights above 94,760 lbs. (42,982 kg), and b) Flaps are not extended beyond Flaps 15 at gross weights above 105,130 lbs. (47,686 kg).
	8) -900ER	C	1	0	May be inoperative provided: a) Flaps are not extended beyond Flaps 30 at landing gross weights above 105,800 lbs. (47,990 kg), b) Flaps are not extended beyond Flaps 15 at landing gross weights above 113,400 lbs. (51,437 kg), c) Flaps are not extended beyond Flaps 10 at landing gross weights above 135,600 lb (61,507 kg), d) Flaps are not extended beyond Flaps 15 at takeoff gross weights above 155,600 lbs. (70,578 kg), and e) Flaps are not extended beyond Flaps 5 at takeoff gross weights above 176,000 lbs (79,832 kg).
9.	Control Wheel Trim Switch Systems	B	2	1	One may be inoperative on non-flying pilot's side provided stabilizer trim system operates normally on flying pilot's side.
10.	FEEL DIFF PRESS Light System	B	1	0	(M) May be inoperative provided: a) Elevator Feel system is verified to operate normally, and b) Verification is repeated each flight day.
11.	Auto Slat Fail Light System (-300/-400/-500/-600/-700/-800/-900)	C	1	0	(M) May be inoperative provided: a) Auto slat systems are verified to operate normally, and b) Verification is repeated every two flight days.
12.	Auto Slat Systems (-300/-400/-500/-600/-700/-800/-900)	C	2	1	(O) One system may be inoperative provided: a) Remaining auto slat system is checked to operate normally, and b) Auto slat fail light operates normally.

U.S. DEPARTMENT OF TRANSPORTATION				FEDERAL AVIATION ADMINISTRATION		MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : 53		PAGE: 27-6		
SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED			
				3. NUMBER REQUIRED FOR DISPATCH			
				4. REMARKS OR EXCEPTIONS			
27 - FLIGHT CONTROLS							
13.	Stall Warning Systems						
	1) (-200/-300/400/-500/-600/-700/-800/-900 without Blended Winglet) (-300/-500 with Blended Winglet)	C	-	1	(M) One may be inoperative provided remaining system is verified to operate normally before each departure.		
	2) (-700/800/-900 with Blended Winglet without Speedbrake Load Alleviation System)	C	2	1	(M) One may be inoperative provided remaining system is verified to operate normally before each departure.		
	3) (-700/-800/-900ER with Blended Winglet with Speedbrake Load Alleviation System)	C	2	1	(M) No. 1 SMYD may be inoperative provided remaining stall warning system is verified to operate normally before each departure.		
	a) (-700)	C	2	1	(M) No. 2 SMYD may be inoperative provided: <ul style="list-style-type: none"> a) Remaining stall warning system is verified to operate normally before each departure, b) Speedbrake handle forces are normal from full down position to full up position, c) Airspeed does not exceed 265 KIAS when inflight gross weight is in excess of 143,000 lbs (64,863 kg.), and d) Severe turbulent air penetration speed is 265 KIAS or 0.76 Mach, whichever is lower, when inflight gross weight is in excess of 143,000 lbs (64,863 kg.). 		
(Continued)							

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST	
FEDERAL AVIATION ADMINISTRATION					
AIRCRAFT: B-737			REVISION NO : 53 DATE: 08/01/2009		PAGE: 27-7
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH	
27 - FLIGHT CONTROLS				4. REMARKS OR EXCEPTIONS	
13.	Stall Warning Systems (Cont'd) 3) (-700/-800/-900ER with Blended Winglet with Speedbrake Load Alleviation System) (Con't)				
	a) (-700) (Cont'd)	C	2	1	(M) No. 2 SMYD may be inoperative provided: a) Remaining stall warning system is verified to operate normally before each departure, b) Speedbrake handle forces are normal from full down position to full up position, and c) Takeoff weight does not exceed 144,500 lbs. (65,544 kg).
	b) (-800)	C	2	1	(M) No. 2 SMYD may be inoperative provided: a) Remaining stall warning system is verified to operate normally before each departure, b) Speedbrake handle forces are normal from full down position to full up position, c) Airspeed does not exceed 265 KIAS when inflight gross weight is in excess of 155,000 lbs (70,306 kg.), and d) Severe turbulent air penetration speed is 265 KIAS or 0.76 Mach, whichever is lower, when inflight gross weight is in excess of 155,000 lbs (70,306 kg.).
		C	2	1	(M) No. 2 SMYD may be inoperative provided: a) Remaining stall warning system is verified to operate normally before each departure, b) Speedbrake handle forces are normal from full down position to full up position, and c) Takeoff weight does not exceed 156,500 lbs. (70,987 kg).
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

27-8

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
			3. NUMBER REQUIRED FOR DISPATCH	
27 - FLIGHT CONTROLS			4. REMARKS OR EXCEPTIONS	
13.	Stall Warning Systems (Cont'd)			
	3) (-700/-800/- 900ER with Blended Winglet with Speedbrake Load Alleviation System) (Con't)			
	c) (-900ER)	C 2	1	(M) No. 2 SMYD may be inoperative provided: a) Remaining stall warning system is verified to operate normally before each departure, b) Speedbrake handle forces are normal from full down position to full up position, c) Airspeed does not exceed 265 KIAS when inflight gross weight is in excess of 170,000 lbs (77,110 kg.), and d) Severe turbulent air penetration speed is 265 KIAS or 0.76 Mach, whichever is lower, when inflight gross weight is in excess of 170,000 lbs (77,110 kg.).
		C 2	1	(M) No. 2 SMYD may be inoperative provided: a) Remaining stall warning system is verified to operate normally before each departure, b) Speedbrake handle forces are normal from full down position to full up position, and c) Takeoff weight does not exceed 171,500 lbs (77,791 kg.).

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT: B-737			REVISION NO : 53 DATE: 08/01/2009		PAGE: 27-9		
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED		
					3.	NUMBER REQUIRED FOR DISPATCH	
						4. REMARKS OR EXCEPTIONS	
27 - FLIGHT CONTROLS							
14.	Rudder Trim Indicator						
	1) (-600/-700/-800/-900)	C	1	0	(O) May be inoperative provided: a) Control Surface Position Indicating System is installed and operates normally, b) Rudder trim actuator is checked to operate normally, and c) Rudder trim is checked to be centered before each departure.		
	2) (All models, upon incorporation Boeing Service bulletin 737-27-1252, 737-27-1253, or 737-27-1255, or production equivalent)	C	1	0	(O) May be inoperative provided: a) Rudder trim actuator is checked to operate normally, and b) Rudder trim is checked centered before each departure.		
15. ***	Mechanical Flaps Position 30 Stop (100/-200 Modified by STC ST00131SE)	C	1	0			
16.	SPEED BRAKE/ SPEEDBRAKES EXTENDED Light						
***	1) (-300/-400/-500)	D	1	0			
	2) (-600/-700/-800/-900)	C	1	0	(M) May be inoperative provided speedbrakes are verified to operate normally.		
17.	Wheel to Rudder Interconnect System (WTRIS) (-600/-700/-800/-900)	C	1	0			
18. ***	Control Surface Position Indicating System	C	1	0			

MASTER MINIMUM EQUIPMENT LIST

PAGE:

27-10

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
27 - FLIGHT CONTROLS						
19.	Rudder Pressure Reducer (RPR) System (-100/-200/-300/-400/-500)	C	1	0	(M)(O) May be inoperative provided: a) RPR system is deactivated, and b) RPR valve is verified to provide high pressure output.	
20. ***	Speed Brake Load Alleviation System					
	1) -700/-800 with Blended Winglet STC ST00830SE					
	a) -700	C	1	0	(M)(O) May be inoperative provided: a) Speedbrake handle forces are normal from full down to full up position, b) Airspeed does not exceed 265 KIAS when inflight gross weight is in excess of 143,000 lbs (64,863 kg), c) Severe turbulent air penetration speed is 265 KIAS or 0.76 Mach whichever is lower, when inflight gross weight is in excess of 143,000 lbs (64,863 kg), and d) Automatic Speed Brake System is considered inoperative.	
		C	1	0	(M) May be inoperative provided: a) Speedbrake handle forces are normal from full down to full up position, and b) Takeoff weight does not exceed 143,500 lbs. (65,090 kg).	
	b) -800	C	1	0	(M)(O) May be inoperative provided: a) Speedbrake handle forces are normal from full down to full up position, b) Airspeed does not exceed 265 KIAS when inflight gross weight is in excess of 155,000 lbs (70,306 kg), c) Severe turbulent air penetration speed is 265 KIAS or 0.76 Mach whichever is lower, when inflight gross weight is in excess of 155,000 lbs (70,306 kg), and d) Automatic Speed Brake System is considered inoperative.	
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

27-11

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

27 - FLIGHT CONTROLS

20.
***Speed Brake Load
Alleviation System
(Cont'd)1) -700/-800 with
Blended Winglet
STC ST00830SE
(Cont'd)

b) -800 (Cont'd)

C

1

0

(M) May be inoperative provided:

- a) Speedbrake handle forces are normal from full down to full up position, and
- b) Takeoff weight does not exceed 155,500 lbs. (70,533 kg).

2) -300/-500 with
Blended Winglet
STC ST01219SE

C

1

0

(M) May be inoperative provided:

- a) Speedbrake handle forces are normal from full down to the full up position,
- b) Airspeed does not exceed 265 KIAS when inflight gross weight is in excess of 125,000 lbs (56,699 kg.), and
- c) Severe turbulent air penetration speed is 265 KIAS or 0.73 Mach whichever is lower, when inflight gross weight is in excess of 125,000 lbs (56,699 kg.).

C

1

0

(M) May be inoperative provided:

- a) Speedbrake handle forces are normal from full down to full up position, and
- b) Takeoff weight does not exceed 126,500 lbs. (57,380 kg).

(Continued)

MASTER MINIMUM EQUIPMENT LIST

PAGE:

27-12

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
27 - FLIGHT CONTROLS						
20. ***	Speed Brake Load Alleviation System (Cont'd)					
	3) -900ER with Blended Winglet	C	1	0	(O)(M) May be inoperative provided:	
					a) Speedbrake handle forces are normal from full down to full up position,	
					b) Airspeed does not exceed 265 KIAS when inflight gross weight is in excess of 170,000 lbs (77,110 kg.), and	
					c) Severe turbulent air penetration speed is 265 KIAS or 0.76 Mach whichever is lower, when inflight gross weight is in excess of 170,000 lbs (77,110 kg.), and	
					d) Automatic Speedbrake System is considered inoperative.	
		C	1	0	(M) May be inoperative provided:	
					a) Speedbrake handle forces are normal from full down to full up position, and	
					b) Takeoff weight does not exceed 170,500 lbs (77,337 kg.).	
21. ***	STBY RUD ON light (Boeing Service Bulletin 737-27A-1279, 737-27-1252R3, 737-27-1253R3, 737-27-1255R3, or production equivalent incorporated)	C	1	0	(M)(O) May be inoperative provided:	
					a) Rudder is verified to operate normally on hydraulic systems A and B independently,	
					b) Standby hydraulic pump is verified to operate normally, and	
					c) Rudder force fight monitor is deactivated.	
22. ***	Quiet Wing Flaps 1* System (STC ST01535SE Only)					
	1) -200	C	1	0	May be inoperative provided:	
					a) Flaps 1* control switch is positioned in UP position,	
					b) System is deactivated by pulling and collaring circuit breaker MS3320-3,	
					c) Appendix QWS001 "Flaps 1* High Altitude Kit" is not used, and	
					d) All other aspects of QWS supplement are followed.	
NOTE: c/b MS3320-3 is located on P6-2 panel.						

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737	REVISION NO : DATE:	54a 05/12/2010	PAGE: 28-1
---------------------------	-------------------------------	-------------------	----------------------

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	
					3.	NUMBER REQUIRED FOR DISPATCH
					4.	REMARKS OR EXCEPTIONS
28 - FUEL						
1.	Fuel Boost Pumps (Main Tanks)					
	1) (-100/-200/-300/-400/-500) (All pumps except Plessey 8240 MK I & MK II)					
	a) Aft Pumps	C	2	1	(O) One may be inoperative provided: a) Both main tank forward pumps operate normally, b) At start of takeoff, fuel quantity in associated tank is not less than 7,500 lbs (3,402 kg.), and c) A minimum fuel quantity of 2,500 lbs. (1,134 kg.) is maintained in associated tank.	
	b) Forward Pumps	C	2	1	(O) One may be inoperative provided: a) Both main tank aft pumps operate normally, b) At start of takeoff, fuel quantity in associated tank is not less than 4,800 lbs (2,177 kg.), and c) A minimum fuel quantity of 1,800 lbs. (817 kg.) is maintained in associated tank.	
	2) (-100/-200/-300) (Plessey 8240 MK I & MK II)					
	a) Aft Pumps	C	2	1	(O) Except for ER operations, one may be inoperative provided: a) Both main tank forward pumps operate normally, b) At start of takeoff, fuel quantity in associated tank is not less than 7,500 lbs (3,402 kg.), and c) A minimum fuel quantity of 2,500 lbs. (1,134 kg.) is maintained in associated tank.	
	b) Forward Pumps	C	2	1	(O) Except for ER operations, one may be inoperative provided: a) Both main tank aft pumps operate normally, b) At start of takeoff, fuel quantity in associated tank is not less than 4,800 lbs (2,177 kg.), and c) A minimum fuel quantity of 1,800 lbs. (817 kg.) is maintained in associated tank.	
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-2

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED	
			3.	NUMBER REQUIRED FOR DISPATCH
				4. REMARKS OR EXCEPTIONS
28 - FUEL				
1.	Fuel Boost Pumps (Main Tanks) (Cont'd)			
	3) (-600/-700/-800/ -900)			
	a) Aft Pumps	C 2	1	(O) Except for ER operations beyond 120 minutes, one may be inoperative provided: a) Both main tank forward pumps operate normally, b) At start of takeoff, fuel quantity in associated tank is not less than 7,500 lbs (3,402 kg.), and c) A minimum fuel quantity of 2,500 lbs. (1,134 kg.) is maintained in associated tank.
	b) Forward Pumps	C 2	1	(O) Except for ER operations beyond 120 minutes, one may be inoperative provided: a) Both main tank aft pumps operate normally, b) At start of takeoff, fuel quantity in associated tank is not less than 4,800 lbs (2,177 kg.), and c) A minimum fuel quantity of 1,800 lbs. (817 kg.) is maintained in associated tank.
2.	Fuel Boost Pumps (Center Tank)	C 2	1	May be inoperative provided tank remains empty.
		C 2	1	(O) May be inoperative with center tank fueled provided: a) Fuel quantity remaining in main wing tanks is adequate to reach a suitable airport if remaining center pump fails at any time, b) Zero fuel weight calculations are adjusted by weight of center tank fuel, c) Effect on airplane balance, in event fuel cannot be used is accounted for, d) LOW PRESSURE light of operating center fuel tank pump operates normally, and e) Center tank quantity indication operates normally. (Continued)

MASTER MINIMUM EQUIPMENT LIST

PAGE:

28-3

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	
					3.	NUMBER REQUIRED FOR DISPATCH
					4.	REMARKS OR EXCEPTIONS
28 - FUEL						
2.	Fuel Boost Pumps (Center Tank) (Cont'd)	C	2	0		May be inoperative provided: a) Center tank quantity indication operates normally, and b) Center tank remains empty, or zero fuel weight calculations are adjusted by weight of center tank fuel. NOTE: AFM limitations for fuel loading must be observed.
	1) Universal Fault Interrupter (UFI) (STC ST01844LA, - 300 Only)	C	2	0		May be inoperative provided associated Center Tank Boost Pump is considered inoperative.
3.	Fuel Boost Pump Low Pressure Warning Light Systems					
	1) Main Tank Pump Low Pressure Warning Light Systems	C	4	3	(M)(O)	May be inoperative provided: a) Associated fuel pump is not used, and b) MASTER CAUTION lights and FUEL system annunciator light are verified to operate normally.
	a) Main Tank Pump Lights	C	4	3		May be inoperative provided: a) Both pumps in associated tank operate normally. b) Associated tank quantity indicator operates normally.
		C	4	3		May be inoperative for an associated inoperative pump.
	2) Center Tank Pump Low Pressure Warning Light Systems	C	2	1	(M)(O)	May be inoperative provided: a. Associated fuel pump is not used. b. MASTER CAUTION lights and FUEL system annunciator light are verified to operate normally.
		C	2	0		May be inoperative provided: a) Center tank fuel is not required for flight, b) Center tank fuel boost pumps are turned off, and c) Center tank remains empty, or zero fuel weight calculations are adjusted by weight of center tank fuel.
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-4

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

28 - FUEL

3. Fuel Boost Pump
Low Pressure
Warning Light
Systems (Cont'd)

2) CenterTank
Pump Low
Pressure
Warning Light
Systems (Cont'd)

a) Center Tank
Pump Lights

C

2

0

(M)(O) May be inoperative provided:
a) Center Tank Fuel Quantity Indicator operates normally, and
b) MASTER CAUTION lights and FUEL system annunciator light are verified to operate normally.

4. APU Fuel Valve

C

1

0

(M)(O) Except for ER operations, may be inoperative provided:
a) APU is not used, and
b) Valve is deactivated closed.

5. Crossfeed VALVE
OPEN Light

C

1

0

(M) Except for ER operations, may be inoperative provided:
a) Crossfeed valve is verified to operate normally,
b) Fuel quantity indication for both main tanks operates normally.

6. Flight Deck Fuel
Quantity Indicators
(Main Tanks)

C

2

1

(M)(O) Except for ER operations, one may be inoperative provided:
a) All boost pumps in associated tank operate normally,
b) Fuel flow meters operate normally,
c) Center tank indicator operates normally,
d) Flight crew periodically computes fuel remaining, or checks fuel remaining against a pre-computed fuel burn chart, and
e) Fuel quantity in associated main tank is verified by an acceptable procedure.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-5

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**

28 - FUEL

7.

Flight Deck Fuel
Quantity Indicator
(Center Tank)

C

1

0

1) (-100 and 600/
-700/-800/-900)

May be inoperative provided:

- a) One center tank boost pump operates normally, and
- b) Center tank remains empty.

2) (-200/-300/-400/
-500)

C

1

0

(M) May be inoperative provided:

- a) One center tank boost pump operates normally, and
- b) Center tank remains empty.

3) (-100/-200/-300/-
400/-500)

C

1

0

(M) Except for ER operations, may be inoperative provided:

- a) Both center tank boost pumps operate normally, and
- b) Fuel quantity in center tank is verified by an acceptable procedure.

4) (-600/-700/-800/
-900 with Boeing
Service Bulletin
737-28A1206 or
production
equivalent
installed)

C

1

0

(M) Except for ER operations, may be inoperative provided:

- a) Both center tank boost pumps operate normally, and
- b) Fuel quantity in center tank is verified by an acceptable procedure.

8.

Fuel Temperature
Indicator

C

1

0

May be inoperative provided Total Air Temperature or Ram Air Temperature is substituted as an indication of fuel temperature.

9.
***Fuel Quantity
Totalizer

C

1

0

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-6

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
28 - FUEL						
10.	Pressure Fueling System	C	1	0		(M) May be inoperative provided alternate procedures are established and used.
	1) Fueling Manifold Check Valves	C	-	0		(M) May be inoperative provided associated Fueling Shutoff Valve is verified to operate normally.
	2) Fueling Shutoff Valves	C	-	0		(M) May be inoperative closed provided: a) Associated Fueling Manifold Check Valve operates normally, and b) Alternate procedures are established and used.
	3) Refuel Panel Fueling Power Control Switch	C	1	0		May be inoperative off provided refuel panel indicator test switch operates normally in AUX FUELING POWER CONTROL position or FUEL DOOR SWITCH BYPASS position as applicable.
11. ***	Fueling Bay Fuel Cap	D	1	0		
12.	Refueling Control Panel Quantity Indicators	C	-	0		(M) May be inoperative provided fuel quantity is verified by an acceptable procedure.
13.	Manually Operated De-fueling Valve					Deleted prior to Revision 27.
14. ***	Aft Auxiliary Fuel Tank Boost Pumps (Boeing Aux Tank)	C	2	1		(O) One may be inoperative provided: a) Fuel quantity in other tanks is adequate to reach an alternate destination if remaining pump fails at any time, and b) Fuel in tank is included as part of zero fuel weight.
		C	2	0		May be inoperative provided tank remains empty.
		C	2	0		May be inoperative provided fuel in tank is included as part of zero fuel weight.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-7

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
28 - FUEL							
15. ***	Flight Deck Fuel Quantity Indicators (Aft Auxiliary Tank)						
	1) Boeing Tank Indicator (Boost Pump Transfer System)	C	1	0		(M)(O) May be inoperative provided both boost pumps operate normally when tank is fueled.	
		C	1	0		May be inoperative provided tank remains empty.	
	2) Rogerson/PATS Tank Indicator (Pressurized Transfer System)	C	1	0		(M)(O) May be inoperative provided:	
						a) Both auxiliary fuel transfer systems operate normally,	
						b) Flight deck center tank fuel quantity indicator operates normally,	
						c) Tank is emptied and serviced with a known quantity of fuel, and	
						d) AFM normal procedures are used for in-flight fuel transfer.	
		C	1	0		May be inoperative provided tank remains empty.	
16.	Fuel Measuring Sticks/Dripsticks	C	-	0		(M) May be inoperative or broken/missing provided fuel quantity is determined by other acceptable means.	
17. ***	Fuel Scavenge System	C	1	0		May be inoperative with fuel scavenge shutoff valve closed.	
		C	1	0		(O) May be inoperative with fuel scavenge shutoff valve open provided No. 1 Main Fuel Tank forward boost pump remains off.	
		C	1	0		May be inoperative with fuel scavenge shutoff valve open provided center tank remains empty.	
18. ***	Aft Auxiliary Tank Pressurized Transfer System (Rogerson/PATS Aux Tank)	C	2	1		(O) One may be inoperative provided:	
						a) Remaining transfer system operates normally,	
						b) Fuel quantity in other tanks is adequate to reach an alternate destination if remaining valve fails at any time, and	
						c) Fuel in tank is included as part of zero fuel weight.	
		C	2	0		May be inoperative provided tank remains empty.	
		C	2	0		May be inoperative provided fuel in tank is included as part of zero fuel weight.	

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST		
AIRCRAFT: B-737			REVISION NO : DATE:		54a 05/12/2010	PAGE: 28-8
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
28 - FUEL						
19.***	Aft Auxiliary Tank Refueling Valves (Rogerson Aux Tank)	C	2	1	(O) One may be inoperative provided: a) Remaining refueling valve operates normally, and b) Automatic refueling shutoff system operates normally.	
20***	Aft Auxiliary Tank LOW PRESSURE/ TRANSFER Lights (Rogerson Aux Tank)	C	2	1	(O) One may be inoperative provided: a) Auxiliary fuel tank indicator operates normally, and b) Automatic transfer system operates normally.	
		C	2	0	(O) May be inoperative for an associated inoperative fuel transfer system.	
21.	Fuel Quantity Test Switches					
	1) Digital System	C	-	0		
	2) Analog System (-100/-200/-300)					
	a) Flight Deck	C	1	0	(M) May be inoperative provided associated fuel quantity indicators are verified to operate normally once each flight day.	
	b) Fueling Panel	C	-	0	(M) May be inoperative provided associated fuel quantity is verified by an acceptable procedure.	
22.	FUEL/SPAR VALVE CLOSED Lights					
	1) FUEL VALVE CLOSED Lights (-100/-200/-300/-400/-500)	C	2	0	(M) May be inoperative provided: a) Associated valve is verified to operate normally, and b) Crossfeed VALVE OPEN light operates normally.	
	2) SPAR VALVE CLOSED Lights (-600/-700/-800/-900	C	2	0	(M) May be inoperative provided: a) Associated valve is verified to operate normally, and b) Crossfeed VALVE OPEN light operates normally.	

MASTER MINIMUM EQUIPMENT LIST

PAGE:

28-9

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2. NUMBER INSTALLED					
				3. NUMBER REQUIRED FOR DISPATCH					
					4. REMARKS OR EXCEPTIONS				
28 - FUEL									
23. ***	Fuel Summation Unit (FSU) (-200/-300/-400/-500)								
	1) PDCS	C	1	0	(M)(O) May be inoperative provided PDCS functions requiring gross weight are not used.				
	2) FMCS (Software Update 7.4 and prior)	C	1	0	(M)(O) May be inoperative provided: a) FMCS functions requiring gross weight are not used, and b) AFDS VNAV mode is not used.				
	3) FMCS (Software Updates 7.5, 8.5, and 10.x)	C	1	0	(M)(O) May be inoperative provided alternate procedures are established and used.				
24.	Refuel Panel Fueling Power Control Switch				Incorporated as a sub-item in 28-10 Rev 47a.				
25. ***	Center Tank Fuel Boost Pump Automatic Shut Off System (Service Bulletin 737-28A1228, 737-28A1216, 737-28A1206, or Equivalent Installed)								
	1) All Models	C	2	0	May be inoperative provided associated center tank fuel boost pump is considered inoperative.				
		C	2	0	May be inoperative provided center tank remains empty.				
(Continued)									

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-10

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

28 - FUEL

25.

Center Tank Fuel
Boost Pump
Automatic Shut Off
System (Service
Bulletin 737-
28A1228, 737-
28A1216,
737-28A1206, or
Equivalent Installed)
(Cont'd)

2) -100/-200/-300/
-400/-500

C

2

0

May be inoperative with center tank fueled provided:

- a. Both center tank fuel boost pump Low Pressure Warning Light Systems operate normally,
- b. Center tank fuel quantity indication operates normally,
- c. Center tank fuel boost pump switches must not be ON unless personnel are available in the flight deck to monitor low pressure lights,
- d. For ground operations, center tank fuel boost pump switches must not be positioned to ON unless the center tank fuel quantity exceeds 1,000 pounds (453 kg), except when defueling or transferring fuel,
- e. Both center tank fuel boost pumps are positioned OFF at first indication of fuel pump low pressure, and
- f. Center tank fuel boost pumps may be positioned ON when established in cruise flight if the center tank contains fuel.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-11

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

28 - FUEL

25.
***Center Tank Fuel
Boost Pump
Automatic Shut Off
System (Service
Bulletin 737-
28A1228, 737-
28A1216,
737-28A1206, or
Equivalent Installed)
(Cont'd)3) -600/-700/-800/
-900

C

2

0

May be inoperative with center tank fueled provided:

- a. Both center tank fuel boost pump Low Pressure Warning Light Systems operate normally,
- b. Center tank fuel quantity indication operates normally,
- c. Center tank fuel boost pumps must not be ON unless personnel are available in the flight deck to monitor low pressure lights,
- d. For ground operations, center tank fuel boost pump switches must not be positioned to ON unless the center tank fuel quantity exceeds 1,000 pounds (453 kg), except when defueling or transferring fuel,
- e. Center tank fuel boost pumps are OFF for takeoff if center tank fuel is less than 5,000 pounds (2,300 kg) with airplane readied for initial taxi,
- f. Both center tank fuel boost pumps are selected OFF when center tank fuel quantity reaches 1,000 pounds (500 kg) of fuel during climb and cruise,
- g. Both center tank fuel boost pumps are selected OFF when center tank fuel quantity reaches 3,000 pounds (1,400 kg) of fuel during descent and landing,
- h. Both center tank fuel boost pumps are positioned OFF at first indication of fuel pump low pressure,
- i. Center tank fuel boost pumps may be positioned ON when established in cruise flight if the center tank contains more than 1000 lbs (500 Kg) of fuel,

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-12

SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
28 - FUEL					
25. ***	Center Tank Fuel Boost Pump Automatic Shut Off System (Service Bulletin 737-28A1228, 737-28A1216, 737-28A1206, or Equivalent Installed) (Cont'd)				<p>j. If the main tanks are not full, the zero fuel gross weight of the airplane plus the weight of center tank fuel may exceed the maximum zero fuel weight by up to 5,000 pounds (2,300 kg) for takeoff, climb and cruise and up to 3,000 pounds (1,400 kg) for descent and landing, provided that the effects of balance (CG) have been considered, and</p> <p>k. Defueling with passengers on board is prohibited.</p>
100.	Forward Auxiliary Fuel System Transfer Valves (PATS, -600/-700/-800)	B	2	1	(M)(O) One may be inoperative provided: <p>a) Inoperative Fwd Aux tank transfer valve is verified "closed" and remains closed,</p> <p>b) Remaining Fwd Aux tank transfer valve operates normally,</p> <p>c) Fuel quantity in main tanks is adequate to reach an alternate destination if remaining transfer valve fails at any time, and</p> <p>d) Fuel in tank is included as part of zero fuel weight.</p>
		C	2	0	May be inoperative provided Fwd Aux tank remains empty.
		C	2	0	May be inoperative provided fuel in Fwd Aux tank is included as part of zero fuel weight.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-13

SYSTEM & SEQUENCE NUMBER		1.	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
28 - FUEL						
101.	Forward Auxiliary Fuel System Vent Valves (PATS, -600/-700/-800)	B	2	1	(M)(O) One may be inoperative provided: a) Remaining Fwd Aux tank vent valve operates normally, b) Fuel quantity in main tanks is adequate to reach an alternate destination if remaining vent valve fails at any time, and c) Fuel in tank is included as part of zero fuel weight.	
		C	2	0	May be inoperative provided Fwd Aux tank remains empty.	
		C	2	0	May be inoperative provided fuel in Fwd Aux tank is included as part of zero fuel weight.	
102.	Forward Auxiliary Fuel System Bleed Air Valve (PATS, -600/-700/-800)	C	1	0	May be inoperative provided: a) Both air conditioning packs operate normally, b) Cabin pressure control system operates normally, and c) Fwd Aux fuel quantity indicator operates normally.	
		C	1	0	May be inoperative provided Fwd Aux tank remains empty.	
		C	1	0	May be inoperative provided fuel in Fwd Aux tank is included as part of zero fuel weight.	
103.	Aft Auxiliary Fuel System Transfer Valves (PATS, -600/-700/-800)	B	2	1	(M)(O) One may be inoperative provided: a) Inoperative Aft Aux tank transfer valve is verified "closed" and remains closed, b) Remaining Aft Aux tank transfer valve operates normally, c) Fuel quantity in main tanks is adequate to reach an alternate destination if remaining transfer valve fails at any time, and d) Fuel in Aft Aux tank is included as part of zero fuel weight.	
		C	2	0	May be inoperative provided Aft Aux tank remains empty.	
		C	2	0	May be inoperative provided fuel in Aft Aux tank is included as part of zero fuel weight.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-14

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3.	4. REMARKS OR EXCEPTIONS
28 - FUEL							
104.	Aft Auxiliary Fuel System Vent Valves (PATS, -600/-700/-800)	B	2	1		(M)(O) One may be inoperative provided:	
						a) Remaining Aft Aux tank vent valve operates normally,	
						b) Fuel quantity in main tanks is adequate to reach an alternate destination if remaining vent valve fails at any time, and	
						c) Fuel in tank is included as part of zero fuel weight.	
		C	2	0		May be inoperative provided Aft Aux tank remains empty.	
		C	2	0		May be inoperative provided fuel in Aft Aux tank is included as part of zero fuel weight.	
105.	Aft Auxiliary Fuel System Bleed Air Valve (PATS, -600/-700/-800)	C	1	0		May be inoperative provided:	
						a) Both air conditioning packs operate normally,	
						b) Cabin pressure control system operates normally, and	
						c) Aft Aux fuel quantity indicator operates normally.	
		C	1	0		May be inoperative provided Aft Aux tank remains empty.	
		C	1	0		May be inoperative provided fuel in Aft Aux tank is included as part of zero fuel weight.	
106.	Auxiliary Fuel System Isolation Valve Open Light (PATS, -600/-700/-800)	C	1	0		(M) May be inoperative provided isolation valve is visually verified open before each flight.	
107.	Auxiliary Fuel System Isolation Valve Closed Light (PATS, -600/-700/-800)	C	1	0		(M) May be inoperative provided isolation valve is visually verified closed before each auxiliary refueling.	
108.	Auxiliary Fuel System Isolation Valve (PATS, -600/-700/-800)	C	1	0		(M) May be inoperative provided:	
						a) Isolation valve is safety wired in open position, and	
						b) Electrical connector is capped for flight.	
							NOTE: Fuel remaining in auxiliary tanks may be used for flight.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-15

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**

28 - FUEL

109.

Auxiliary Tank
Fueling Valves
(PATS, -600/-700/
-800)1) Forward Auxiliary
Refueling Valve

C

1

0

(M) May be inoperative provided forward refueling
valve is verified "closed".NOTE 1: Auxiliary Fuel Tanks shall not be fueled until
refueling valve has been verified to operate
normally.

NOTE 2: Fuel remaining in tank may be used for flight.

2) Aft Auxiliary
Refueling Valve

C

1

0

(M) May be inoperative provided Aft Refueling valve is
verified "closed".NOTE 1: Auxiliary Fuel Tanks shall not be fueled until
refueling valve has been verified to operate
normally.NOTE 2: Fuel remaining in tank may be used for
flight.

110.

Auxiliary Fuel
System Alert
Message Display
(PATS, -600/-700/
-800)

C

2

1

(M) One may be inoperative provided transfer system
is verified to operate normally.

C

2

0

May be inoperative provided auxiliary tanks remain
empty.

C

2

0

May be inoperative provided fuel auxiliary tanks is
included as part of zero fuel weight.

111.

Auxiliary Fuel
Control Unit (PATS,
-600/-700/-800)

C

1

0

(O) May be inoperative provided auxiliary fuel tanks
remain empty.

MASTER MINIMUM EQUIPMENT LIST

PAGE:

28-16

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3.	NUMBER REQUIRED FOR DISPATCH	4.	REMARKS OR EXCEPTIONS
28 - FUEL									
112.	Auxiliary Fuel Low Level Float Switches (PATS, -600/-700/-800)								
	1) Forward Tank System	C	2	1					(O) One low level switch may be inoperative provided fuel quantity indicators operate normally.
		C	2	0					(O) May be inoperative provided tank remains empty.
		C	2	0					(O) May be inoperative provided fuel in tank is included as part of zero fuel weight.
	2) Aft Tank System	C	2	1					(O) One low level switch may be inoperative provided fuel quantity indicators operate normally.
		C	2	0					(O) May be inoperative provided tank remains empty.
		C	2	0					(O) May be inoperative provided fuel in tank is included as part of zero fuel weight.
113.	Auxiliary Fuel Processor (PATS, -600/-700/-800)	C	1	0					(O) May be inoperative provided auxiliary fuel tank remains empty.
114.	Auxiliary Fuel Pressure Switches (PATS, -600/-700/-800)								
	1) Forward Tank Pressure Switches	C	2	1					(M) One may be inoperative provided: a) Failed pressure switch indicates low pressure, b) Pressurization system operates normally, and c) Air conditioning packs operate normally.
		C	2	0					May be inoperative provided tank remains empty.
		C	2	0					May be inoperative provided fuel in tank is included as part of zero fuel weight.
(Continued)									

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-17

SYSTEM & SEQUENCE NUMBER		1.	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
28 - FUEL						
114.	Auxiliary Fuel Pressure Switches (PATS, -600/-700/-800) (Cont'd)					
	2) Aft Tank Pressure Switches	C	2	1	(M) One may be inoperative provided: a) Failed pressure switch indicates low pressure, b) Pressurization system operates normally, and c) Air conditioning packs operate normally.	
		C	2	0	May be inoperative provided tank remains empty.	
		C	2	0	May be inoperative provided fuel in tank is included as part of zero fuel weight.	
115.	Auxiliary Fuel Center Tank Float Switches (PATS, -600/-700/-800)	C	2	0	(O) May be inoperative provided auxiliary fuel tanks remain empty.	
		C	2	0	(O) May be inoperative provided fuel in tank is included as part of zero fuel weight.	
116.	Auxiliary Fuel Maintenance Switches (PATS, -600/-700/-800)	C	2	1	(M) One may be inoperative provided: a) Affected maintenance switch/indicator is failed in an open condition, and b) Remaining maintenance switch/indicator is verified to operate normally.	
		C	2	0	May be inoperative provided auxiliary fuel tanks remain empty.	
		C	2	0	May be inoperative provided fuel in tank is included as part of zero fuel weight.	
117.	Auxiliary Fuel Alert Switches (PATS, -600/-700/-800)	C	2	1	(M) One may be inoperative provided: a) Affected alert switch/indicator is failed in an open condition, and b) Remaining alert switch/indicator is verified to operate normally.	
		C	2	0	May be inoperative provided auxiliary fuel tanks remain empty.	
		C	2	0	May be inoperative provided fuel in tank is included as part of zero fuel weight.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

28-18

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		
				3.	NUMBER REQUIRED FOR DISPATCH	
					4.	REMARKS OR EXCEPTIONS
28 - FUEL						
118.	Auxiliary Fuel Test Switches (PATS, -600/-700/-800)	C	2	0	(M) May be open provided: a) Associated fuel quantity indicator display is verified to operate normally before each flight, and b) Alert message displays are verified to operate normally before each flight.	
119.	Flight Deck Fuel Quantity Indicators (Auxiliary Tanks) (PATS, -600/-700/-800)					
	1) Aft Auxiliary Tank System	C	2	1	(O) One may be inoperative provided transfer system operates normally and total fuel quantity on the FMC is verified to be correct.	
		C	2	0	May be inoperative provided auxiliary fuel tanks remain empty.	
		C	2	0	May be inoperative provided fuel in tank is included as part of zero fuel weight.	
	2) Forward Auxiliary Tank System	C	2	1	(O) One may be inoperative provided transfer system operates normally and total fuel quantity on the FMC is verified to be correct.	
		C	2	0	May be inoperative provided auxiliary fuel tanks remain empty.	
		C	2	0	May be inoperative provided fuel in tank is included as part of zero fuel weight.	

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : 53 DATE: 08/01/2009		PAGE: 29-1
SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED	
				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
29 - HYDRAULICS					
1.	Ground Inter-connect Valve (System A and B) (-100/-200)	C	1	0	(M) May be inoperative provided valve remains closed.
2.	System B Pumps				
	1) (-100/-200)	C	2	1	Except for ER operations, one may be inoperative provided: a) Pressure indicator operates normally, and b) Thrust reversers operate normally.
	2) Engine Driven Hydraulic Pump Depressurization Function (-300/-400/-500/-600/-700/-800/-900)	C	1	0	
3.	System Pressure Indications (A and B)				
	1) (-100/-200)	C	2	0	(O) May be inoperative provided: a) System pressure is checked from brake pressure indicator before each departure, and b) All hydraulic low pressure lights operate normally.
	2) (-300/-400/-500/-600/-700/-800/-900)	C	2	1	(O) One may be inoperative provided: a) System pressure is checked before each departure, and b) All hydraulic low pressure lights operate normally.
4.	System A Pump Low Pressure Indication Systems	C	2	1	(O) One may be inoperative provided output of associated pump is checked before each departure.
5.	System B Pump Low Pressure Indication Systems	C	2	1	(O) One may be inoperative provided output of associated pump is checked before each departure.
6.	Hydraulic Brake Pressure Indicator				Moved to item 32-13, Revision 33.

MASTER MINIMUM EQUIPMENT LIST

PAGE:

29-2

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		
				3. NUMBER REQUIRED FOR DISPATCH		
				4. REMARKS OR EXCEPTIONS		
29 - HYDRAULICS						
7.	System A and B Overheat Lights					
***	1) System A Overheat Lights (-100/-200)	D	2	0		
	2) System B Overheat Lights (-100/-200)	C	2	0	May be inoperative provided associated system B Low Pressure light operates normally.	
	3) (-300/-400/-500/-600/-700/-800/-900)	C	2	0	May be inoperative provided associated Low Pressure light operates normally.	
8.	Hydraulic Quantity Low Level Light System B (-100/-200)	C	1	0	(M) May be inoperative provided quantity is verified adequate before each departure.	
9.	Hydraulic Quantity Low Level Light System (Standby System)	C	1	0	(M) May be inoperative provided quantity is verified adequate before each departure.	
10.	System A Pumps					
	1) Engine Driven Hydraulic Pump Depressurization Function	C	-	0		
11.	System A Quantity Indication System (Flight Deck)					
	1) -100/-200	C	1	0	(M) May be inoperative provided: a) Quantity is verified adequate before each departure, b) System A pressure indicator operates normally, and c) System B and Standby systems low quantity lights operate normally.	
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST		
FEDERAL AVIATION ADMINISTRATION						
AIRCRAFT:		REVISION NO :		53		PAGE:
B-737		DATE:		08/01/2009		29-3
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED		
29 - HYDRAULICS				3.	NUMBER REQUIRED FOR DISPATCH	
					4. REMARKS OR EXCEPTIONS	
11.	System A Quantity Indication System (Flight Deck) (Cont'd)					
	2) (-300/-400/-500/-600/-700/-800/-900)	C	1	0	(M) May be inoperative provided: a) Quantity is verified adequate before each departure, b) System pressure indication operates normally, and c) Pump low pressure lights operate normally.	
12.	Standby System Low Pressure Light	C	1	0	(M) May be inoperative provided: a) Standby system low quantity light operates normally, b) Output of standby pump is verified before each departure, and c) Both System B pumps operate normally.	
13.	Hydraulic Reservoir Pressurization System Sources	C	-	1	(M) May be inoperative provided reservoir can be pressurized.	
14.	System A Overheat Lights				Incorporated into Item 29-7, in Revision 39.	
15.	System B Quantity Indication System (Flight Deck) (-300/-400/-500/-600/-700/-800/-900)	C	1	0	(M) May be inoperative provided: a) Quantity is verified adequate before each departure, b) System pressure indication operates normally, and c) Pump low pressure lights operate normally.	
16. ***	Hydraulic Reservoir Air Pressure Indicator (Wheel Well)	C	-	0		
17.	Hydraulic Reservoir Quantity Indicator (Wheel Well)	C	-	0		
18.	Hydraulic Reservoir Fill System (Wheel Well)	C	1	0		

MASTER MINIMUM EQUIPMENT

PAGE:

30-1

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION
LIST

MASTER MINIMUM EQUIPMENT

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

30-2

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3.

NUMBER REQUIRED FOR DISPATCH

4.

REMARKS OR EXCEPTIONS

30 - ICE AND RAIN PROTECTION

3. Engine and Nose
Cowl Anti-Ice
Valves1) (-100/-200)
(Cont'd)

C

6

5

(M)(O) One may be inoperative open provided:

- a) All thrust rating limits on associated engine, except for takeoff and go-around, are reduced by .03 EPR,
- b) Enroute climb limited weight is reduced by 3,000 lbs. (1,361 Kg),
- c) At temperatures greater than 50 degrees F (10 degrees C),
 - (1) Takeoff and go-around thrust limits on associated engine are reduced by .03 EPR,
 - (2) Takeoff and landing performance limited weight is reduced by 3,000 lbs. (1,361 Kg),
- d) All remaining valves operate normally,
- e) Operating temperature for cowl valves is limited to 50 degrees F (10 degrees C) maximum (ambient or total air temperature) unless S/B 71-1045 or 71-1046 "Nose Cowl TAI Spray Ring Modification" or production equivalent has been incorporated, and
- f) For JT8D-15/15A, JT8D-17/17A engine installations, the following adjustments must be applied when dispatching with anti-ice OFF, and the following conditions exists:

-TAKEOFF-

(15/15A) Pressure altitude between 3,000 and 10,000 feet, ambient temperature below 0 degrees F (-18 degrees C).

(-17/17A) Pressure altitude between 3,000 and 10,000 feet, ambient temperature below 15 degrees F (-10 degrees C).

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION
LIST

MASTER MINIMUM EQUIPMENT

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

30-3

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

30 - ICE AND RAIN PROTECTION

3. Engine and Nose
Cowl Anti-Ice
Valves (Cont'd)1) (-100/-200)
(Cont'd)

-GO-AROUND-

(-15/15A) Pressure altitude between 3,000 and
10,000 feet, ambient temperature below 0
degrees F (-18 degrees C).(-17/17A) Pressure altitude between 3,000 and
10,000 feet, ambient temperature below 15
degrees F (-10 degrees C).(1) Takeoff and go-around thrust limits on
associated engine are reduced by .03
EPR,(2) Takeoff and landing performance
weight is reduced by 3,000 lbs. (1,361
Kg).

2) (-300/-400/-500)

C

2

1

(M) One may be inoperative closed provided airplane
is not operated in known or forecast icing conditions.

C

2

1

(M)(O) One may be inoperative locked open provided:
a) Associated High Stage Valve is considered
inoperative,
b) Ambient temperature is below 100 degrees F
(38 degrees C),
c) A minimum of 60% N1 is maintained on
associated engine during flight in icing
conditions,
d) All thrust rating limits on affected engine,
except Takeoff and Go-Around, are reduced
by 0.8% N1,
e) Enroute climb limited weight is reduced by
4,500 lbs. (2,040 Kg),

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION
LIST

MASTER MINIMUM EQUIPMENT

AIRCRAFT: B-737	REVISION NO : 53 DATE: 08/01/2009	PAGE: 30-4
---------------------------	--	----------------------

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED																	
			3. NUMBER REQUIRED FOR DISPATCH																	
			4. REMARKS OR EXCEPTIONS																	
30 - ICE AND RAIN PROTECTION																				
3.	Engine and Nose Cowl Anti-Ice Valves (Cont'd)																			
	2) (-300/-400/-500) (Cont'd)				f) At temperatures greater than 50 degrees F (10 degrees C), Takeoff and Go-Around thrust limits on associated engine and takeoff and landing performance limited weights are reduced by:															
					<table><tr><td>RATING</td><td>%N1</td><td>WEIGHT</td></tr><tr><td>18.5 K</td><td>0.8</td><td>3650 lbs. (1670 Kg)</td></tr><tr><td>20.0 K</td><td>0.8</td><td>3900 lbs. (1770 Kg)</td></tr><tr><td>22.0 K</td><td>0.8</td><td>3900 lbs. (1770 Kg)</td></tr><tr><td>23.5 K</td><td>1.1</td><td>4650 lbs. (2110 Kg)</td></tr></table>	RATING	%N1	WEIGHT	18.5 K	0.8	3650 lbs. (1670 Kg)	20.0 K	0.8	3900 lbs. (1770 Kg)	22.0 K	0.8	3900 lbs. (1770 Kg)	23.5 K	1.1	4650 lbs. (2110 Kg)
RATING	%N1	WEIGHT																		
18.5 K	0.8	3650 lbs. (1670 Kg)																		
20.0 K	0.8	3900 lbs. (1770 Kg)																		
22.0 K	0.8	3900 lbs. (1770 Kg)																		
23.5 K	1.1	4650 lbs. (2110 Kg)																		
	3) (-600/-700/-800/-900)	C	2	1	(M) Except for ER operations beyond 120 minutes, one may be inoperative closed provided airplane is not operated in known or forecast icing conditions.															
		C	2	1	(M)(O) One may be inoperative locked open provided: (a) Associated High Stage Valve is considered inoperative, (b) Ambient temperature is below 100 degrees F (38 degrees C), (c) A minimum of 60% N1 is maintained on associated engine during flight in icing conditions, (d) All thrust rating limits on affected engine, except Takeoff and Go-Around, are reduced by 1.1% N1, (e) Enroute climb limited weight is reduced by 4,000 lbs. (1,810 Kg),															
(Continued)																				

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION
LIST

MASTER MINIMUM EQUIPMENT

AIRCRAFT: B-737	REVISION NO : DATE:	53 08/01/2009	PAGE: 30-5
---------------------------	-------------------------------	------------------	----------------------

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
30 - ICE AND RAIN PROTECTION						
3.	Engine and Nose Cowl Anti-Ice Valves (Cont'd)					
	3) (-600/-700/-800/-900) (Cont'd)					f) At temperatures greater than 50 degrees F (10 degrees C); (1) Takeoff and go-Around thrust limits on associated engine are reduced by 1.4% N1, and (2) Takeoff and landing performance limited weights are reduced by 4500 lbs. (2040 Kg). g) For temperatures at or below 50 degrees F (10 degrees C), base performance limited weights on Engine Anti-Ice ON.
4.	Engine and Nose Cowl Anti-Ice Valve Position Lights or TAI Indications					
	1) (-100/-200)	C	-	0		(M) May be inoperative provided valve is verified to operate normally before each departure.
	2) (-300/-400/-500/-600/-700/-800/-900)	C	-	0		(O) May be inoperative provided valve is verified to operate normally before each departure.
	3) (-600/-700/-800/-900)	C	4	2		One valve position indication (either COWL VALVE OPEN light or TAI indication) for each engine may be inoperative provided other valve position indication for that engine operates normally.
	4) (All Models)	C	-	-		May be inoperative provided associated valve is considered inoperative.

MASTER MINIMUM EQUIPMENT

PAGE:

30-6

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
30 - ICE AND RAIN PROTECTION						
5.	Pitot/Static Probe Heaters					
	1) (-100/-200/-300/-400/-500)					
	a) No. 1 Aux Pitot/Static Heater (Right Lower Probe)	B	1	0	May be inoperative provided:	a) No. 2 Aux Pitot Static heater operates normally, b) RVSM operations are not conducted, and c) Airplane is not operated in known or forecast icing conditions.
	b) No. 2 Aux Pitot/Static Heater (Left Lower Probe)	B	1	0	May be inoperative provided:	a) No. 1 Aux Pitot Static heater operates normally, b) RVSM operations are not conducted, and c) Airplane is not operated in known or forecast icing conditions.
		B	1	0	May be inoperative provided:	a) No.1 Aux Pitot Static heater operates normally, and b) Dispatch deviations for associated equipment are observed.
	c) Pitot/Static Heaters (Upper Probes)	B	2	1	Pilot's or copilot's may be inoperative for day VMC provided airplane is not operated in visible moisture, or in known or forecast icing conditions.	
	2) (-600/-700/-800/-900)					
	a) Left/Right Pitot Heaters	B	2	1	Except for ER operations beyond 120 minutes, one may be inoperative for day VMC provided:	a) Aux Pitot heater operates normally, b) Airplane is not operated in visible moisture, and c) Airplane is not operated in known or forecast icing conditions.
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION
LIST

MASTER MINIMUM EQUIPMENT

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

30-7

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

30 - ICE AND RAIN PROTECTION

5. Pitot/Static Probe
Heaters2) (-600/-700/-800/
-900) (Cont'd)b) Aux Pitot
Heater (Right
Lower Probe)

B

1

0

Except for ER operations beyond 120 minutes, may
be inoperative provided:

- a) Both Left and Right Pitot heaters operate normally, and
- b) Airplane is not operated in known or forecast icing conditions.

6. Vertical Stabilizer
*** Pitot Heaters
(Elevator and
Rudder Feel
Systems)

B

2

1

Except for ER operations beyond 120 minutes, one
may be inoperative provided airplane is not operated
in known or forecast icing conditions.7. Total Air
Temperature Probe
Heater

C

-

0

Except for ER operations beyond 120 minutes, may
be inoperative provided airplane is not operated in
known or forecast icing conditions.

C

-

0

(O) May be inoperative provided an alternate
temperature indicator system is installed and
operating normally (i.e., Ram Air or Static Air
Temperature).8. Angle of Attack
Sensor Heater(s)/
Stall Warning
System Sensor
Heater(s)/Alpha
Vane Heater(s)

C

-

0

Except for ER operations beyond 120 minutes, may
be inoperative provided airplane is not operated in
known or forecast icing conditions.

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION
LIST

MASTER MINIMUM EQUIPMENT

AIRCRAFT: B-737	REVISION NO : 53	PAGE: 30-8
	DATE: 08/01/2009	

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3.	NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
30 - ICE AND RAIN PROTECTION								
9.		Pitot, Pitot/Static and Temperature Probe Heater Lights						
***		1) Green (Heater On) Lights (-100/-200)						
		a) Pitot and Pitot/Static	B	-	-			(M) One may be inoperative provided: a) Required heater function is verified before each departure, and b) HEATER OFF light operates normally.
		b) Temperature	C	1	0			(M) May be inoperative provided associated heater function is verified to operate normally before each departure.
			C	1	0			May be inoperative provided associated heater is inoperative.
***		2) Amber (Heater Off) Lights						
		a) Pitot and Pitot/Static	B	-	0			(M) Except for ER operations beyond 120 minutes, may be inoperative provided: a) Associated heater function is verified to operate normally, and b) Airplane is not operated in known or forecast icing conditions.
		b) Temperature	C	-	1			
			C	-	0			(M) May be inoperative provided associated heater function is verified to operate normally before each departure.
			C	-	0			May be inoperative provided associated heater is inoperative.
10.		Wing Anti-Ice Duct Overheat System						
***		1) Ground Test Feature	C	1	0			

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION
LIST

MASTER MINIMUM EQUIPMENT

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

30-9

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS****30 - ICE AND RAIN PROTECTION**11. Electrically Heated
Windshields1) No.1 or No. 2
Window

C

4

3

Except for ER operations beyond 120 minutes, one
No. 1 or No. 2 window heater may be inoperative
provided:

- a) Airplane is not operated in known or forecast
icing conditions,
- b) Windshield de-fog system operates normally,
and
- c) Airspeed is limited to 250 KIAS below 10,000
feet MSL.

*** 2) No. 4 or No. 5
Window

C

4

0

No. 4 and No. 5 window heat may be inoperative
provided airspeed is limited to 250 KIAS below 10,000
feet MSL.*** 3) No. 3 Window
Heat System(s)

D

2

0

12. De-Fog System

C

1

0

13. Windshield Wiper
System(s)

C

2

0

May be inoperative provided airplane is not operated
in precipitation within 5 nautical miles of airport of
takeoff or intended landing.

1) Park Function

C

2

0

May be inoperative for all flight conditions provided
blade(s) can be positioned in a location that will not
obstruct forward vision.*** 2) Intermittent
Speed Function
(-300/-400/-500/
-600/-700/-800/
-900)

D

2

0

3) Low Speed
Function

C

2

0

May be inoperative provided both high speed
functions operate normally.

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION
LIST

MASTER MINIMUM EQUIPMENT

AIRCRAFT: B-737	REVISION NO : 53 DATE: 08/01/2009	PAGE: 30-10
---------------------------	--	-----------------------

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
30 - ICE AND RAIN PROTECTION							
13.	Windshield Wiper System(s) (cont'd)						
	4) High Speed Function	C	2	1		One may be inoperative provided associated low speed function operates normally.	
		C	2	0		May be inoperative provided both low speed functions operate normally and rain intensity is less than moderate.	
						DELETED REVISION 53.	
14. ***	RainBoe Rain Repellent System (-100/-200/-300/-400/-500)	D	1	0			
15. ***	Windshield Perimeter Heater(s)	C	2	0			
16. ***	HEATER OFF Light (-100/-200)	B	1	0		(O) May be inoperative provided: a) Remaining components of pitot heat system are verified to operate normally, and b) Airplane is not operated in known or forecast icing conditions.	
17.	COWL ANTI-ICE Lights (-300/-400/-500/-600/-700/-800/-900)	C	2	1		Except for ER operations beyond 120 minutes, one may be inoperative provided airplane is not operated in known or forecast icing conditions.	
		C	2	1		(M)(O) One may be inoperative provided associated Cowl anti-ice valve is locked open.	
18. ***	Alpha Vane Heater Light Systems	C	2	0		(M) May be inoperative provided associated heater function is verified to operate normally before each departure.	
		C	2	0		May be inoperative provided associated heater is considered inoperative.	
19. ***	Drain Mast Heaters	C	2	0		(M) May be inoperative provided water supply to associated components is secured off.	
20. ***	Ice Detection System	D	1	0			

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION
LIST

MASTER MINIMUM EQUIPMENT

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

30-11

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

30 - ICE AND RAIN PROTECTION

21.
***Control Stand Wing
Anti-Ice Switches

C

2

0

(O) May be inoperative closed.

C

2

0

(O) May be inoperative open.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

31-1

SYSTEM & SEQUENCE NUMBER			1.	2. NUMBER INSTALLED	
				3. NUMBER REQUIRED FOR DISPATCH	
				4. REMARKS OR EXCEPTIONS	
31 - INDICATING / RECORDING SYSTEMS					
1.	Clocks	C	2	1	One may be inoperative at either pilot or copilot station.
***	1) Automatic UTC Update Function	C	2	0	(O) May be inoperative provided manual mode is set and operates normally.
2.	Flight Data Recorder System (FDR)	C	-	-	Any in excess of those required by 14 CFR may be inoperative.
		A	-	0	May be inoperative provided: a) Cockpit Voice Recorder (CVR) operates normally, b) Airplane is not dispatched from a designated airport as listed in operator's MEL unless; 1) FDR failure occurs after pushback but prior to takeoff, or 2) FDR repair was attempted but was not successful. c) In those cases where repair is attempted but not successful, aircraft may be dispatched on a flight or series of flights until next designated airport where repair must be accomplished prior to dispatch, and d) Repairs are made within three flight days.
	1) FDR Recording Parameters required by 14 CFR	A	-	-	May be inoperative provided: a) Cockpit Voice Recorder (CVR) operates normally, and b) Repairs are made within 20 calendar days.
	2) FDR Recording Parameters not required by 14 CFR	A	-	-	May be inoperative provided repairs are made prior to completion of next heavy maintenance visit.
3.	Engine Pressure Ratio Limit (EPRL) System				Moved to MMEL Item 34-41.
4. ***	Reference Speed Computer (Total Fuel & VREF Indicator -100/-200)	C	1	0	

U.S. DEPARTMENT OF TRANSPORTATION										
FEDERAL AVIATION ADMINISTRATION					MASTER MINIMUM EQUIPMENT LIST					
AIRCRAFT: B-737				REVISION NO : 53			PAGE:			
				DATE: 08/01/2009			31-2			
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2. NUMBER INSTALLED						
				3. NUMBER REQUIRED FOR DISPATCH						
					4. REMARKS OR EXCEPTIONS					
31 - INDICATING / RECORDING SYSTEMS										
5.	Cockpit Voice Recorder (CVR) System									
6. ***	AIDS Maintenance Recorder	D	1	0						
7. ***	Aircraft Condition Monitoring System (ACMS)	D	1	0						
	1) Avionics miniQAR (Quick Access Recorder) (STC's ST02472AT or ST03151AT)	D	1	0						
8.	Common Display System (CDS) (-600/-700/-800/-900)									
	1) Display Units (DU)									
	a) Lower DU	C	1	0	(O) May be inoperative provided: a) All remaining DUs operate normally, and b) It is checked that engine display can be switched to an alternate DU.					
	b) Inboard DU	A	2	1	(O) For EFIS/MAP configuration, one may be inoperative provided: a) It is checked that engine display can be switched to an alternate DU, b) All navigation must be based on ILS/VOR/DME, and c) Repairs are made within one flight day.					
(Continued)										

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : 53 DATE: 08/01/2009		PAGE: 31-3
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3.	NUMBER REQUIRED FOR DISPATCH
					4.
					REMARKS OR EXCEPTIONS
31 - INDICATING / RECORDING SYSTEMS					
8.	Common Display System (CDS) (-600/-700/-800/-900) (Cont'd)				
	2) CDS MAINT Annunciation				
	a) PFD/ND	B	-	0	May be dispatched with faults indicated by CDS MAINT annunciation provided CDS Operational Program Software (OPS) P/N31111-HNP-01A-05 or later, is installed.
	b) EFIS/MAP	A	-	0	May be dispatched with faults indicated by CDS MAINT annunciation provided: a) Captain's Inboard DU operates normally, b) CDS Operational Program Software (OPS) P/N31111-HNP-01A-05 or later, is installed, and c) Repairs are made within one flight day.
9.	Remote Light Sensor System (-300/-400/-500/-600/-700/-800/-900)	C	1	0	May be inoperative provided all manual display brightness controls operate normally.
10.	Speed Reference Selector (-600/-700/-800/-900)	C	1	0	May be inoperative provided speeds can be set using CDU.
11. ***	Mechanical Timer	C	1	0	(O) May be inoperative provided alternate procedures are established and used.
		D	1	0	May be inoperative provided procedures do not require its use.
12. ***	Takeoff Warn Test Switch	C	1	0	
		D	1	0	May be inoperative provided procedures do not require its use.

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

31-4

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	
					3.	NUMBER REQUIRED FOR DISPATCH
						4. REMARKS OR EXCEPTIONS
31 - INDICATING / RECORDING SYSTEMS						
13. ***	Flat Panel Display System (Universal Avionics, Inc. EFI-890) (STC ST03355AT and ST03362AT)					
	1) Inboard DU (ND)	A	2	1	(O) For PFD/ND configuration, one may be inoperative provided:	
	a) Display Control Panel Switches/Control Knobs	A	-	O	May be inoperative provided:	
	(1) TERR	C	2	1	a) Reversionary Display on PFD is checked prior to departure,	
	(2) TFC	C	2	1	b) PFD Lateral Deviation Scale operates normally, and	
	(3) WX	C	2	1	c) Repairs are made within two flight days.	
	2) Outboard DU (PFD)					
	a) Display Control Panel Switches/Control Knobs					
	(1) RA/DA Set	C	2	0	May be inoperative provided approach minimums do not require its use.	
	(2) RA/DA	C	2	0	May be inoperative provided approach minimums do not require its use.	
	(3) RA/Test	C	2	0		
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION				
FEDERAL AVIATION ADMINISTRATION			MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : 53 DATE: 08/01/2009	PAGE: 31-5
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED
			3.	NUMBER REQUIRED FOR DISPATCH
31 - INDICATING / RECORDING SYSTEMS			4.	
			REMARKS OR EXCEPTIONS	
13. ***	Flat Panel Display System (Universal Avionics, Inc. EFI-890) (STC ST03355AT and ST03362AT) (Cont'd)			
	3) Forward Electronic Panel (ND) (-200)	B	1	0
	(1) TERR	B	1	0
	(2) TFC	D	1	0
		C	1	0
	(3) WX	C	1	0
				May be inoperative provided Terrain Awareness and Warning System (TAWS) is considered inoperative. May be inoperative provided TCAS VSI operates normally. May be inoperative provided TCAS is considered inoperative. May be inoperative provided Weather Radar is considered inoperative.
14. ***	TAKEOFF CONFIG Light			
	1) -100/-200/-300/-400/-500 (upon incorporation of Boeing Service Bulletin 737-31A1325)	C	1	0
		C	1	0
				(O) May be inoperative provided flight crew performs a briefing on cabin altitude warning indications and procedures before engine start for the first flight of the day or following any change of either flight crew member. (Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

31-6

SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3. NUMBER REQUIRED FOR DISPATCH	
31 - INDICATING / RECORDING SYSTEMS				4. REMARKS OR EXCEPTIONS	
14. ***	TAKEOFF CONFIG Light (Cont'd)				
	2) -600/-700/-800/- 900 (upon incorporation of Boeing Service Bulletin 737- 31A1332, or production equivalent)	C	2	0	May be inoperative provided the associated CABIN ALTITUDE warning light operates normally.
		C	2	0	(O) May be inoperative provided flight crew performs a briefing on cabin altitude warning indications and procedures before engine start for the first flight of the day or following any change of either flight crew member.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

32-1

SYSTEM & SEQUENCE NUMBER			1.	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
32 - LANDING GEAR							
1. ***	Gear Seal Warning System (-100/-200)	C	1	0	(M) May be inoperative provided gear seal function is checked once each flight day.		
2.	Antiskid System						
	1) (-100/-200/-300/-400/-500)	C	1	0	(O) May be inoperative provided operations are conducted in compliance with AFM.		
	2) (-600/-700/-800/-900)	C	1	0	(M)(O) May be inoperative provided: a) Associated Antiskid channel(s) is deactivated, and b) Operations are conducted in compliance with AFM.		
3.	Parking Brake Valve (-300/-400/-500/-600/-700/-800/-900)	C	1	0	(M)(O) May be inoperative closed provided operations comply with AFM antiskid inoperative decrements.		
4.	Parking Brake Light						
	1) Solenoid Parking Brake Valve Installed (-100/-200)	C	1	0	(O) May be inoperative provided antiskid system is turned OFF when parking brake is used.		
	2) Motor Operated Parking Brake Valve Installed	C	1	0	(M) May be inoperative provided parking brake shutoff valve is verified to operate normally.		
***	3) External Parking Brake Light	C	1	0	(O) May be inoperative provided alternate procedures are established and used.		
		D	1	0	May be inoperative provided procedures do not require its use.		
5. ***	Main Wheel Well Inflatable Seal System (-100/-200)	C	1	0	(M) May be inoperative provided system is deactivated and secured.		

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

32-2

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
32 - LANDING GEAR							
6.	Landing Gear Warning and Indicating System (-100/-200/-300/-400/-500)	C	-	2			Either of two other indicating systems may be inoperative provided center panel indications operate normally.
	1) Secondary Gear Warning System (Pemco F/QC and COMBI)	B	1	0			(O) May be inoperative provided Main Gear and Nose Gear Viewer are accessible during all phases of flight.
7. ***	Automatic Brake System	C	1	0			(M) May be inoperative provided system is deactivated and secured.
8.	Rudder Pedal Nose Wheel Steering System						
	1) Rotary Actuator (-300/-400/-500/-600/-700/-800/-900)	C	1	0			(M)(O) May be inoperative deactivated in disengage position provided: a) Operation of associated systems are not affected, and b) All takeoffs and landings are made by pilot with access to an operating tiller.
9. ***	Direct Reading Tire Pressure Gauge	D	-	0			
10.	Alternate Antiskid Valves (-300/-400/-500/-600/-700/-800/-900)	C	2	0			(M) May be inoperative provided manual braking capability of alternate brake system is verified on associated wheels.
11. ***	Brake Temperature Monitor System	C	1	0			(O) May be inoperative provided AFM Maximum Quick Turnaround Weight limitations are observed.
		D	1	0			(O) May be inoperative provided: a) AFM Maximum Quick Turnaround Weight limitations are observed, and b) Procedures are not based on its use.
12. ***	Nose Wheel Steering Switch (-300/-400/-500/-600/-700/-800/-900)	C	1	0			(M) May be inoperative provided: a) Nose wheel steering is powered by Hydraulic System A, and b) Landing gear transfer valve is verified to operate normally.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT: B-737			REVISION NO : 53 DATE: 08/01/2009		PAGE: 32-3		
SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED			
				3.	NUMBER REQUIRED FOR DISPATCH		
					4. REMARKS OR EXCEPTIONS		
32 - LANDING GEAR							
13.	Hydraulic Brake Pressure Indication System						
	1) (-100/-200)						
	a) Wheel Well Brake Accumulator Gauges	C	2	0	May be inoperative provided associated Flight deck brake pressure indicator operates normally.		
	b) Flight Deck HYD BRAKE PRESS Indicator Systems	C	2	1	(M) One brake indication (A or B) may be inoperative provided associated brake accumulator charge is verified normal once each flight day.		
	2) (-300/-400/-500/-600/-700/-800/-900)						
	a) Wheel Well Brake Accumulator Gauge	C	1	0	May be inoperative provided Flight deck brake pressure indicator operates normally.		
	b) Flight Deck HYD BRAKE PRESS Indicator System	C	1	0	(M) May be inoperative provided brake accumulator charge is verified normal once each flight day.		
14.	Gear Retraction Braking System (-600/-700/-800/-900)	C	1	0	(O) May be inoperative provided: a) After takeoff, landing gear remains extended for two minutes before retraction, and b) Takeoff performance is based on Landing Gear Extended.		
15.	Landing Gear Selector Valve Bypass Module (-600/-700/-800/-900)	C	1	0	(M)(O) May be inoperative provided it is deactivated in normal position.		

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

32-4

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
32 - LANDING GEAR						
16.	Landing Gear Actuation System (-600/-700/-800/-900)	B	1	0		(M)(O) May be inoperative provided: a) Inoperative components are secured by an accepted procedure, b) Landing gear are secured in down position, and c) Airplane is dispatched in accordance with AFM Gear Extended Appendix.
17.	Proximity Switch Electronics Unit (PSEU) System and Supplemental Proximity Sensor Electronics Unit (SPSEU) (-600/-700/-800/-900)					
	1) PSEU Fault	C	-	0		(M) May be dispatched with faults indicated by PSEU light provided PSEU is checked for faults before each departure.
		C	-	0		May be dispatched with faults indicated by PSEU light provided PSEU light can be extinguished.
	2) PSEU Light	C	1	0		(M) May be inoperative provided PSEU is checked for faults before each departure.
***	3) Supplemental Proximity Sensor Electronics Unit (SPSEU) Light (-900ER)	C	1	0		(M) May be inoperative provided SPSEU is checked for faults before each departure.
18.	Landing Gear Alternate Extension System (-600/-700/-800/-900)	B	1	0		(M)(O) May be inoperative provided: a) Inoperative Components are secured by an accepted procedure, b) Landing gear are secured in down position, and c) Airplane is dispatched in accordance with AFM Gear Extended Appendix.
19.	Main Landing Gear Uplock Springs	B	4	3		(M)(O) One spring on one main gear Uplock mechanism may be missing provided landing gear lever remains in UP position for duration of flight until gear extension is required.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST				
FEDERAL AVIATION ADMINISTRATION								
AIRCRAFT:			REVISION NO :		53		PAGE:	
B-737			DATE:		08/01/2009		32-5	
SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED				
				3.	NUMBER REQUIRED FOR DISPATCH			
					4.	REMARKS OR EXCEPTIONS		
32 - LANDING GEAR								
20.	Landing Gear Frangible Fitting (-600/-700/-800/-900)	C	2	0	(M)	May be broken or missing provided fitting is replaced with a hydraulic cap assembly.		
21.	Flap Landing Warning Switch, S138 (-600/-700/-800/-900)	C	1	0	(M)	Switch contacts normally in use may be inoperative provided: a) S138 switch is rewired using an alternate set of contacts, and b) PSEU BITE is used to verify normal operation of S138 switch.		
22.	Two-position Tail Skid							
***	1) (-800)							
	a) Retraction Mechanism	C	1	0	(M)(O)	May be inoperative provided: a) Tail skid is secured in retracted position, and b) Appropriate performance adjustments are applied.		
		C	1	0	(M)(O)	May be inoperative provided: a) Tail skid is secured in extended position, and b) Appropriate performance adjustments are applied.		
	b) Cartridge Core Assembly	B	1	0	(M)(O)	May be inoperative provided: a) Detailed AMM inspection reveals no internal and external structural damage, b) Tail skid is secured in retracted position, and c) Appropriate performance adjustments are applied.		
(Continued)								

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

32-6

SYSTEM & SEQUENCE NUMBER	1. ITEM	2.	NUMBER INSTALLED		
32 - LANDING GEAR				3. NUMBER REQUIRED FOR DISPATCH	
				4. REMARKS OR EXCEPTIONS	
22.	Two-position Tail Skid (Cont'd)				
	2) (-900ER)				
	a) Retraction Mechanism	C	1	0	(M)(O) May be inoperative provided: a) Tail skid is secured in retracted position, and b) Appropriate performance adjustments are applied.
		C	1	0	(M)(O) May be inoperative provided: a) Tail skid is secured in extended position, and b) Appropriate performance adjustments are applied.
	b) Cartridge Core Assembly	B	1	0	(M)(O) May be inoperative provided: a) Detailed AMM inspection reveals no internal and external structural damage, b) Tail skid is secured in retracted position, and c) Appropriate performance adjustments are applied.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737	REVISION NO : DATE:	54 10/12/2009	PAGE: 33-1
---------------------------	-------------------------------	------------------	----------------------

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	
					3.	NUMBER REQUIRED FOR DISPATCH
					4.	REMARKS OR EXCEPTIONS
33 - LIGHTS						
1.	Cockpit/Flight Deck/Flight Compartment and Instrument Lighting System	C	-	-		Individual lights may be inoperative provided remaining lights are: a) Sufficient to clearly illuminate all required instruments, controls, and other devices for which it is provided, b) Positioned so that direct rays are shielded from flight crew's eyes, and c) Lighting configuration and intensity is acceptable to flight crew.
2.	Cabin Interior Illumination (Includes Pemco -300QC and -400 COMBI)					
	1) Passenger and Combi Configurations Without Photolumi- nescent Emergency Escape Path Marking Systems	C	-	-		Individual lights may be inoperative provided sufficient lighting remains for cabin attendants/cargo couriers to perform their duties.
	2) Passenger and Combi Configurations With Photolumi- nescent Emergency Escape Path Marking Systems	C	-	-		Individual lights may be inoperative provided: a) Sufficient lighting remains for cabin attendants/cargo couriers to perform their duties, and b) Remaining lighting is sufficient to charge Photoluminescent Emergency Escape Path Marking System.

FEDERAL AVIATION ADMINISTRATION

AIRCRAFT:

REVISION NO :

PAGE:

33-2

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED	3.	NUMBER REQUIRED FOR DISPATCH	4.	REMARKS OR EXCEPTIONS
33 - LIGHTS									
3.	Passenger Notice System ("NO SMOKING/FASTEN SEAT BELT/ RETURN TO SEAT") Signs	C	-	-	-	-	(M) May be inoperative provided: a) Associated passenger seat or lavatory from which a passenger lighted information sign is not readily legible is not occupied, and b) Associated seat or lavatory is blocked and placarded - DO NOT OCCUPY.		
		C	-	-	-	-	NOTE: These provisos are not intended to prohibit lavatory use or inspections by crewmembers. (O) May be inoperative and associated passenger seat or lavatory may be occupied provided: a) PA system operates normally, and b) PA system is used to notify passengers and cabin crew when associated sign(s) are placed on or off.		
	1) All Cargo, Supernumerary/ Courier Area Lighted Information Signs	C	-	-	-	-	(O) May be inoperative provided alternate procedures are established and used to notify couriers/ supernumeraries when associated sign(s) are placed on or off.		
	2) Aural Tone System	C	1	0	1	0			
	3) Flight Deck Automatic Function	C	1	0	1	0	(O) May be inoperative provided: a) Manual control function operates normally, and b) Alternate procedures are established and used.		
4.	Lower Cargo Compartment Light Systems (Fwd/Aft)	C	-	0	-	0			
	1) Light Lens (-100/-200/-300/-400/-500/-900)	C	-	0	-	0	May be broken/missing provided associated light bulb is removed.		
(Continued)									

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737	REVISION NO : DATE:	54 10/12/2009	PAGE: 33-3
---------------------------	-------------------------------	------------------	----------------------

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
33 - LIGHTS					
4.		Lower Cargo Compartment Light Systems (Fwd/Aft) (Cont'd)			
	2)	Light Lens (-600/-700/-800 prior to incorporation of Boeing Service Bulletins 737-21-1135, 737-26-1121, and 737-1122, or Production Equivalent	C -	0	May be broken/missing provided associated light bulb is removed.
	3)	Light Lens (-600/-700/-800 upon incorporation of Boeing Service Bulletins 737-21-1135, 737-26-1121, and 737-1122, or Production Equivalent	C -	-	Any number from rear lower cargo compartment and one from forward lower cargo compartment may be broken/missing provided associated light bulb is removed.
5.		High Intensity or Strobe Lights System			
***	1)	All Models (Except Models with STC's ST01821LA, ST01873LA, and ST02015LA)	C 1	0	
	2)	Models with STCs ST01821LA, ST01873LA, and ST012015LA	C 1	0	May be inoperative for day operations.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

33-4

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
33 - LIGHTS							
6.	Anti-Collision Beacons (Without Blended Winglet, -800/-900/-900ER Blended Winglet, and –700 Blended Winglet With Dual Glass Lens) (Except STC’s ST01821LA and ST01873LA)	C	2	0		May be inoperative for night operations provided wing tip and tail strobe lights are installed and operate normally.	
	1) Blended Winglet	C	2	0		May be inoperative for day operations.	
	a) (-700 with Single Plastic Lens)	C	2	0		May be inoperative for day operations.	
	b) (-800 with Light Fence)					NOTE: Both anti-collision beacons must be operative for night operations.	
	c) (-300/-500 with STC ST01219SE and Winglet Strobe Lights)	C	2	0		Deleted Revision 45 a. Incorporated into Item 33-6.	
		C	2	0		May be inoperative for night operations provided winglet strobe lights operate normally.	
		C	2	0		NOTE: Tail strobe light may be inoperative.	
	2) (STC’s ST01821LA and ST01873LA)	C	2	0		May be inoperative for day operations.	
7.	Wing Illumination Lights	C	2	0		(O) May be inoperative provided ground de-icing procedures do not require their use.	
***	1) Overwing Ice Lights (Grimes Aerospace STC ST500CH)	C	2	0			

MASTER MINIMUM EQUIPMENT LIST

PAGE:

33-5

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST	
FEDERAL AVIATION ADMINISTRATION					
AIRCRAFT:		REVISION NO :		54	PAGE:
B-737		DATE:		10/12/2009	33-6
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
33 - LIGHTS					
11.	Wing Tip Position Lights (Cont'd)				
	2) Light Bulbs/ Lamps (-700/ -800 Blended Winglet with Single Plastic Lens)	C	-	5	Any except following minimum may be inoperative for night operations: a) Both stationary red wing tip bulbs, b) One stationary green wing tip bulb, and c) One stationary white tail light bulb at each wing tip position.
		B	-	4	Any except following minimum may be inoperative for night operations: a) One stationary red wing tip bulb, b) One stationary green wing tip bulb, and c) One stationary white tail light bulb at each wing tip position.
	a) Stationary Red Wing Tip Light Bulbs/ Lamps				DELETED REVISION 49a
12. ***	Door Locked Light (Flight Deck to Cabin) (Not 14 CFR 25.795 Compliant)	C	1	0	May be inoperative provided locking function operates normally.
13.	Master Caution Lights				Deleted prior to Rev. 27.
14.	Exterior Emergency Lighting System	B	1	0	May be inoperative for day operations
		B	1	0	May be inoperative for all-cargo night operations provided forward entry door escape slide lights operate normally.

U.S. DEPARTMENT OF TRANSPORTATION
FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737	REVISION NO : 54 DATE: 10/12/2009	PAGE: 33-7
---------------------------	--	----------------------

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
33 - LIGHTS						
15.	Interior Emergency Exit Lighting System					
	1) Mixed or All-Cargo Configuration	C	1	0	Lights may be inoperative in cargo areas provided: a) No persons occupy that area, and b) Forward entrance door light operates normally at all times.	
***	2) Stowage Bin Bullnose Lights (-600/-700/-800/-900)	C	-	-	Light assemblies installed above aisle (curved edge of stowage bins) may be inoperative provided no two adjacent (opposite side) light assemblies are inoperative.	
***	3) Advance Technology Interior (ATI) (Aisle Light Assemblies) (-200/-300/-400/-500)	C	-	-	Light assemblies installed above aisle (curved edge of stowage bins) may be inoperative provided no two adjacent (opposite side) light assemblies are inoperative.	
	4) Flight Deck Exit Light	C	1	0	May be inoperative for day operations.	
16.	System annunciator Lights, Left and Right (Pilot's Light Shield)	C	-	-	(O) One light may be inoperative for an operating system	
		C	-	-	May be inoperative for an associated inoperative system	
17.	Flight deck Master Lights Test and Individual Light's Press-to-Test Features	C	-	-	(O) May be inoperative provided intended function of associated light(s) is verified once each flight day.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

33-8

SYSTEM & SEQUENCE NUMBER		1.	2.		NUMBER INSTALLED		
					3.	NUMBER REQUIRED FOR DISPATCH	
						4.	REMARKS OR EXCEPTIONS
33 - LIGHTS							
18.	Wheel Well Lights						
	1) Dome Lights	C	3	0			
	2) Inspection Flood Lights						
	a) (-100/-200/ -300/-400/ -500)	C	3	1			Main gear lights may be inoperative for day operations only.
		D	3	0			Lights may be inoperative provided a landing gear indicating system other than viewer system and independent of center panel is installed and operates normally.
	b) (-600/-700/ -800/-900)	D	2	0			
19.	Floor Proximity Emergency Escape Path Marking System (All Models and STC's)						
	1) Incandescent Lighting System	C	-	-			Individual lights may be inoperative provided minimum acceptable lighting levels specified in one of following documents are complied with: a) FAA engineering approval letter, b) FAA approved report of Type Design holder, c) Limitations and Conditions section of the applicable Supplemental Type Certificate (STC) or d) An FAA approved report incorporated in the Master Drawing List for the applicable STC.
	2)Photoluminescent Lighting System	C	-	-			Components may be inoperative provided minimum acceptable lighting levels specified in one of following documents are complied with: a) FAA engineering approval letter, b) FAA approved report of Type Design holder, c) Limitations and Conditions section of the applicable Supplemental Type Certificate (STC) or d) An FAA approved report incorporated in the Master Drawing List for the applicable STC.
20. ***	LOGO Light System	D	1	0			

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT: B-737			REVISION NO : DATE:		54 10/12/2009		PAGE: 33-9
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED		
					3. NUMBER REQUIRED FOR DISPATCH		
					4. REMARKS OR EXCEPTIONS		
33 - LIGHTS							
21. ***	Main Deck Cargo Compartment Lighting (737C, STC's ST01566LA, SA2969SO, SA2970SO, ST00287AT, ST00283AT, ST01827LA, and ST01961SE)						
		1) Cargo Door Floodlights	C	2	0	(M) May be inoperative for night operations provided alternate procedures are established and used.	
			C	2	0	May be inoperative for day operations.	
					NOTE: Not required for all passenger operations.		
22. ***	Main Deck Cargo Door-System Annunciator Light (737-300 QC, PEMCO Aeroplex, Inc. -300/-400, and STC's ST01566LA, and ST01961SE)	2) Cargo Compartment Lights (STC's ST00283AT, and ST01827LA)	C	-	0	(M) May be inoperative for night operations provided alternate procedures are established and used.	
			C	-	0	May be inoperative for day operations.	
		1) System Annunciator Lights, Pilot's Overhead Panel (737-300QC, and STC's ST01566LA, and ST01961SE)	A	2	1	(M)(O) One warning light may be illuminated provided: a) Alternate procedures are established and used to verify main cargo door is closed and locked, and b) Repairs are made within two flight days.	
					(Continued)		

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

33-10

SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
33 - LIGHTS				3. NUMBER REQUIRED FOR DISPATCH	
22. *** Main Deck Cargo Door-System Annunciator Light (737-300 QC, PEMCO Aeroplex, Inc. -300/-400, and STC's ST01566LA, and ST01961SE) (Cont'd)				4. REMARKS OR EXCEPTIONS	
2) System Annunciator Lights, Operator Control Panel (737-300 QC, PEMCO Aeroplex, Inc. -300/-400, and STC's ST01566LA, and ST01961SE)	A		-	-	(M)(O) One warning light may be inoperative provided: a) It is not a VENT DOOR OPEN light, b) Vent door handle is locked, c) Outside view port is verified green, d) Individual lock is not loose, e) Main cargo door is verified closed, latched and locked, and f) Repairs are made within two flight days.
3) Hydraulic System Arm Pressure Indicator Lights (PRESS), Operator Control Panel (STC SA2969SO)					DELETED REVISION 49.
4) Hydraulic System Green Indicator Lights, Operator Control Panel (STC SA2969SO)					DELETED REVISION 49
23. Master Dim System	B	1	1	0	Dim function may be inoperative provided: a) TEST and BRT functions operate normally, b) Except during light test, switch is placed in BRT, and c) Light intensity is acceptable to flight crew.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

33-11

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
33 - LIGHTS						
24. ***	Sterile Flight Compartment Light System	C	1	0	(O) May be inoperative provided alternate procedures are established and used.	
		D	1	0	May be inoperative provided procedures do not require its use.	
25.	Service Area Light Systems (Nose, Electrical Equipment, Air Conditioning, Aft Accessory, APU, Tailcone Compartments, and Fueling Panel)	C	-	0		
		D	-	0	May be inoperative for day operations.	
26.	Main Cargo Compartment In-Flight Access Alert System (STC ST01961SE)	C	-	0	May be inoperative provided in-flight access to the main deck cargo compartment is prohibited.	
	1) Main Cargo Compartment Lights	C	-	0	May be inoperative provided in-flight access to the main deck cargo compartment is prohibited.	
	2) Main Cargo Compartment Alert Horns	C	2	0	May be inoperative provided in-flight access to the main deck cargo compartment is prohibited.	

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT: B-737			REVISION NO : DATE:		54a 05/12/2010		PAGE: 34-1
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2. NUMBER INSTALLED			
				3. NUMBER REQUIRED FOR DISPATCH			
					4. REMARKS OR EXCEPTIONS		
34 - NAVIGATION							
1.	Mach/Airspeed Indications						
	1) Mach Indications	C	2	1	One may be inoperative provided one Mach/Airspeed warning and Mach trim system operates normally.		
	a) (-100/-200/-300/-400/-500)	C	2	0	May be inoperative provided: a) Airplane remains at or below FL 230, and b) Airspeed remains at or below 320 KIAS.		
	b) (-600/-700/-800/-900)	C	2	0	May be inoperative provided: a) Airplane remains at or below FL 280, and b) Airspeed remains at or below 320 KIAS.		
***	2) Airspeed Indicators (-300/ -400/-500)	C	2	1	One may be inoperative provided: a) EFIS Speed Tape displays are installed and operate normally, and b) One Mach/Airspeed warning operates normally.		
***	3) EFIS Speed Tape (-300/-400/ -500)	C	2	0	May be inoperative provided airspeed indicators are installed and operate normally at each pilot's station.		
***	4) Airspeed Cursor (-100/-200/-300/ -400/-500)	A	2	1	(O) One may be inoperative provided: a) Alternate procedures are established and used, and b) Repairs are made within three flight days.		
***	5) External Airspeed Markers (Bugs) (-100/-200/-300/ -400/-500)	C	-	0	(O) May be inoperative or missing provided alternate procedures are established and used.		
***	6) Digital Airspeed Readout (-100/-200/-300/ -400/-500)	C	-	0			

MASTER MINIMUM EQUIPMENT LIST

PAGE:

34-2

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-3

SYSTEM & SEQUENCE NUMBER	1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
34 - NAVIGATION				
2. Mach/Airspeed Warning Systems (Cont'd)				
2) Clacker (Cont'd)				
b) (-300/-400/ - 500/-600/ - 700/-800/ - 900) (Cont'd)	B	2 0		Systems may be inoperative provided: a) Both Mach indicators operate normally, b) 330 KIAS/.76 Mach airspeed limitations are observed, and c) If overspeed warning occurs below .76 Mach, system must be deactivated by pulling associated circuit breaker and observe speed limits.
3. Altimeter Vibrators				
1) Servo-Pneumatic	C	2 1		One may be inoperative provided associated air data computer operates normally.
2) Pneumatic	C	2 1		One may be inoperative provided VMC exists at departure and arrival airports.
3) Pneumatic (With Electric/ Electronic Altimeter)	C	1 0		May be inoperative provided VMC exists at departure and arrival airports.
4) One Pneumatic and one Servo-Pneumatic	C	2 1		Servo-Pneumatic may be inoperative provided associated air data computer operates normally.
5) Standby Altimeter Vibrator (With Electric/ Electronic Altimeter)	C	2 1		Pneumatic may be inoperative provided VMC exists at departure and arrival airports.
5) Standby Altimeter Vibrator (With Electric/ Electronic Altimeter)	C	1 0		May be inoperative provided VMC exists at departure and arrival airports.
4. *** Static Air Temperature Indication	D	- 0		
5. Total Air Temperature Indication	C	- 0		May be inoperative provided an alternate air temperature indication (e.g. PDCS, FMCS, RAT, SAT) operates normally.

U.S. DEPARTMENT OF TRANSPORTATION				
FEDERAL AVIATION ADMINISTRATION			MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737		REVISION NO : 54a DATE: 05/12/2010		PAGE: 34-4
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED
34 - NAVIGATION			3.	NUMBER REQUIRED FOR DISPATCH
				4. REMARKS OR EXCEPTIONS
6.	Attitude Director Indicators (ADI)			Delete prior to revision 27.
7.	Standby Horizon Indicator			
	1) Standby Attitude Indicator	B	1	0
***	2) ILS Indication	D	1	0
8.	Angle of Attack Indications	C	-	0

9.	Turn and Bank Indicators			
***	1) Rate of Turn Indicators (-100/ - 200/-300/-400/ - 500)	C	2	1
		C	2	0
10.	Directional Gyro Compass System			May be inoperative provided Standby Horizon Indicator operates normally.
11.	Non-Stabilized Magnetic Compass	B	1	0
		B	1	0
		C	1	0
				Deleted prior to Revision 27.
				(O) May be inoperative provided any combination of three gyro or INS (IRU) stabilized compass systems are operative.
				(O) May be inoperative provided:
				a) Any combination of two gyro or INS (IRU) stabilized compass systems are operative, and
				b) Airplane is operated with dual independent navigation capability, and under positive radar control by ATC on enroute portion of flight.
				(O) May be inoperative for flights that are entirely within areas of magnetic unreliability provided two stabilized directional gyro systems are installed, operative, and used in conjunction with free gyro navigation techniques.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST	
FEDERAL AVIATION ADMINISTRATION					
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010		PAGE: 34-5
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		
			3. NUMBER REQUIRED FOR DISPATCH		
			4. REMARKS OR EXCEPTIONS		
34 - NAVIGATION					
12.	Flight Director Systems	C	2	0	May be inoperative provided approach minimums do not require its use.
13.	Distance Measuring Equipment Systems	D	-	-	Any in excess of those required by 14 CFR may be inoperative.
14.	Marker Beacon Receiver System	C	-	0	May be inoperative provided approach minimums do not require its use.
15.	Weather Radar	C	-	0	May be inoperative provided Radar System is not required by 14 CFR.
		D	-	1	May be inoperative provided one remaining Radar System operates normally.
***	1) Windshear Detection and Avoidance System (Predictive)	B	-	0	(O) May be inoperative provided alternate procedures are established and used. NOTE: Operator's alternate procedures should include reviewing windshear avoidance and windshear recovery procedures.
		C	-	0	(O) May be inoperative provided: a) Alternate procedures are established and used, and b) Windshear Warning and Guidance System (Reactive) operates normally.
***	2) Autotilt/ Multiscan Function (Including STCs ST01843AT, ST01470LA-D)	C	1	0	May be inoperative provided manual tilt function operates normally.
***	3) Stabilization Function	C	1	0	(M) May be inoperative provided: a) Manual tilt control operates normally, and b) Antenna is verified to scan in a horizontal plane with tilt at zero degrees.
16.	Radio Compass Systems (ADF)	D	-	-	Any in excess of those required by 14 CFR may be inoperative.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-6

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3.

NUMBER REQUIRED FOR DISPATCH

4.

REMARKS OR EXCEPTIONS

34 - NAVIGATION

17.

VHF Navigation
Systems (VOR/ILS)1) (-100/-200/-300/ -
400/-500)

D

-

-

Any in excess of those required by 14 CFR, and not
powered by a Standby Bus, may be inoperative
provided approach minimums do not require its use.a) Auto Tune
Function

C

-

0

(O) May be inoperative provided:
a) Enroute or approach procedures do not require
its use, and
b) Manual tuning operates normally.2) (-300/-400/-500
GNLU-920 MMR,
STC ST00998LA-
D)

D

-

-

Any in excess of those required by 14 CFR, and not
powered by a Standby Bus, may be inoperative
provided approach minimums do not require its use.a) Equipment
Cooling Fan

B

2

0

3) (-600/-700/-800/ -
900)

a) VOR Systems

D

2

-

Any in excess of those required by 14 CFR, and not
powered by a Standby Bus, may be inoperative.

b) ILS Systems

D

2

-

Any in excess of those required by 14 CFR, and not
powered by a Standby Bus, may be inoperative
provided approach minimums do not require its use.c) Auto Tune
Function

Deleted MMEL Rev 52.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-7

**SYSTEM &
SEQUENCE
NUMBER**

ITEM

1.

2. NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

34 - NAVIGATION

18. ATC Transponders
and Automatic
Altitude Reporting
System

B

-

0

May be inoperative provided:

- a) Operations do not require its use, and
- b) Prior to flight, approval is obtained from ATC facilities having jurisdiction over planned route of flight.

D

-

1

Any in excess of those required by 14 CFR may be inoperative.

1) Elementary and
Enhanced
downlink Aircraft
Reportable
Parameters not
Required by 14
CFR

A

-

0

May be inoperative provided:

- a) Operations do not require its use, and
- b) Repairs are made prior to completion of next heavy maintenance visit.

*** 2) ADS-B Extended
Squitter
Transmissions

A

-

0

May be inoperative provided:

- a) Operations do not require its use, and
- b) Repairs are made prior to completion of next heavy maintenance visit.

19. Instrument
Comparator or
Warning System
(-200/-300/-400/
-500, includes STC
ST03355AT)

C

-

0

May be inoperative provided approach minimums do not require its use.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-8

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
34 - NAVIGATION						
20.	Radio Altimeter Systems					
	1) Receiver/Transmitters					
	a) (-100/-200)	A	-	0	(M)(O) May be inoperative deactivated provided:	a) Approach minimums or operating procedures do not require its use, b) Associated autopilot is not used for approach and landing, c) Autothrottle is not used for approach and landing, and d) Repairs are made within two flight days.
		C	-	0	(M)(O) May be inoperative deactivated provided:	a) Approach minimums or operating procedures do not require its use, b) Associated autopilot is not used for approach and landing c) Autothrottle is not used for approach and landing, and d) GPWS is not required by 14 CFR.
	b) (-300/-400/-500)	C	2	1	(M)(O) May be inoperative deactivated provided:	a) Approach minimums or operating procedures do not require its use, b) Associated autopilot is not used for approach and landing, c) Autothrottle is not used for approach and landing, and d) GPWS operates normally.
		A	2	1	(M)(O) May be inoperative deactivated provided:	a) Approach minimums or operating procedures do not require its use, b) Associated autopilot is not used for approach and landing, c) Autothrottle is not used for approach and landing, and d) Repairs are made within two flight days.
	c) (-600/-700/800/-900)	- C	2	1		
	2) Indications	C	-	2		
		C	-	0		

MASTER MINIMUM EQUIPMENT LIST

PAGE:

34-9

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
34 - NAVIGATION								
21. ***	Air Data System (non Electric Airspeed Indicators (-200)	A	-	0				
22.	Alternate Static System (-100/-200)	C	1	0				
23. ***	True Airspeed Indication	C	-	0				
24.	Airspeed Indicators (-300/-400/-500)							
25.	Altitude Alerting System	A	1	0				(O) May be inoperative provided: a) Autopilot with altitude hold, and altitude capture operates normally, b) Enroute operation, i.e. RVSM, do not require its use, c) Airplane does not depart from a designated airport (as listed in the operator's MEL) where repair or replacement can be made, and d) Repairs are made within 3 flight days.
	1) Aural Alert	C	-	0				May be inoperative provided: a) Visual alert operates normally, and b) Auto-pilot with altitude hold and altitude capture operates normally.
	2) Visual Alert	C	-	0				May be inoperative provided: a) Aural alert operates normally, and b) Auto-pilot with altitude hold and altitude capture operates normally.
26.	Terrain Awareness and Warning System (TAWS) (Includes STC ST03355AT &ST03362AT)							
	1) Ground Proximity Warning System (GPWS)	A	1	0				
	a) Modes 1 thru 4	A	4	0				
	b) Test Mode	A	1	0				
(Continued)								

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-10

SYSTEM & SEQUENCE NUMBER	ITEM	1.
--------------------------------	------	----

2.	NUMBER INSTALLED
----	------------------

3.	NUMBER REQUIRED FOR DISPATCH
----	------------------------------

4.	REMARKS OR EXCEPTIONS
----	-----------------------

34 - NAVIGATION

26. Terrain Awareness
and Warning System
(TAWS) (Includes
STC ST03355AT
&ST03362AT)
(Cont'd)

1) Ground Proximity
Warning System
(GPWS) (Cont'd)

c) Glideslope
Deviation(s)
(Mode 5)

C

2

1

B

2

0

d) Advisory
Callouts

B

-

0

e) Windshear
Warning and
Flight
Guidance
Mode
(Reactive)

C

-

0

B

1

0

C

1

0

2) Terrain System –
Forward Looking
Terrain Avoidance
(FLTA) and
Premature
Descent Alert
(PDA) Functions

B

1

0

3) Terrain Displays

C

-

1

B

-

0

a) Vision One (STC
ST03355AT)

D

-

0

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION				
FEDERAL AVIATION ADMINISTRATION			MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010	
PAGE: 34-11				
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	
			3. NUMBER REQUIRED FOR DISPATCH	
34 - NAVIGATION			4. REMARKS OR EXCEPTIONS	
26.	Terrain Awareness and Warning System (TAWS) (Includes STC ST03355AT &ST03362AT) (Cont'd)			
***	4) Runway Awareness and Advisory System (RAAS)	C	1	0
27.	Long Range Navigation Systems (INS, Loran, Omega)	C	-	0

28.	Performance Data Computer System (PDCS)	C	1	0

29.	Speed Command (Fast-Slow) Indicators (-100/ - 200/-300/-400/ - 500)	C	2	0

30.	ADI Test (-100/ - 200/-300/-400/ - 500)	C	2	0

31.	Speed Cursor Remote Drive	C	1	0

32.	Instrument Transfer Switching System	C	1	0
33.	Vertical Gyro System (-100/-200)			
	1) Number 1 and 2	C	2	1
***	2) Auxiliary Gyro	C	1	0
34.	Standby Altimeter Vibrator			

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-12

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

34 - NAVIGATION

35.	Inertial Reference System(s) (IRS) (-300/-400/-500/-600/-700/-800/-900)	B	2	1	
	1) IRS Data Display (Aft Overhead Panel)	C	1	0	
	2) IRS Ground Crew Call Horn	C	1	0	
36.	Flight Management Computer System (FMCS)				
***	1) (-200 CMA-900 FMS/GPS)	D	1	0	
	a) Annunciator Lights/ Switches (STC ST6895-AT)	C	9	0	
	(1) NAV/FMS	D	2	0	
		A	2	1	
	(2) WPT	C	2	0	
		A	2	1	
	(3) GPS APPR CAP	C	1	0	
		C	1	0	
	(4) GPS INT	C	2	0	
		A	2	1	
	(5) OFFSET	C	2	0	
	b) FMU	C	-	1	
	c) MCDU	C	1	0	

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-13

**SYSTEM &
SEQUENCE
NUMBER**

ITEM

1.

2. NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

34 - NAVIGATION

36. Flight Management
Computer System
(FMCS) (Cont'd)

1) (-200 CMA-900
FMS/GPS)
(Cont'd)

d) Navigation
Databases

e) DME

f) GPS

g) HSI Switching
Unit (STC
ST01676AT)

2) (-300/-400/-500/
-600/-700/-800/
-900)

a) FMC Alert
Lights

b) Computer

(1) -300/-400/ -
500

(2) -600/-700/ -
800/-900

c) CDU/MCDU

C

-

-

C

1

0

C

1

0

C

1

0

C

1

0

C

2

0

C

2

1

C

2

1

C

2

0

C

-

1

C

-

0

C

-

0

C

-

1

C

-

0

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-14

SYSTEM & SEQUENCE NUMBER	ITEM	1.
--------------------------------	------	----

2.	NUMBER INSTALLED
----	------------------

3.	NUMBER REQUIRED FOR DISPATCH
----	------------------------------

4.	REMARKS OR EXCEPTIONS
----	-----------------------

34 - NAVIGATION

36.	Flight Management Computer System (FMCS) (Cont'd)			
-----	---	--	--	--

2)	(-300/-400/-500/ -600/-700/-800/ -900) (Cont'd)			
----	---	--	--	--

***	d) Alternate Navigation Control Display Unit (ANCDU)			
-----	---	--	--	--

(1) CRT ANCDU (-300/-400/ -500)	C	-	0	
--	---	---	---	--

(2) LCD ANCDU (-700IGW)	C	-	0	
----------------------------	---	---	---	--

e) Navigation Databases	C	-	-	
----------------------------	---	---	---	--

***	3) Universal Avionics UNS-1F (STC ST03356AT and ST03362AT)	C	2	0
-----	---	---	---	---

a) Navigation Computer Unit (NCU)	C	2	0	
---	---	---	---	--

b) Control Display Unit (CDU) (-300)	C	2	0	
--	---	---	---	--

c) Global Navigation Satellite System (GNSS)	C	2	0	
--	---	---	---	--

(Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-15

**SYSTEM &
SEQUENCE
NUMBER**

ITEM

1.

2. NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

34 - NAVIGATION

36. Flight Management
Computer System
(FMCS) (Cont'd)

*** 3) Universal Avionics
UNS-1F (STC
ST03356AT and
ST03362AT)
(Cont'd)

d) Navigation
Display (ND)
Caution
Annunciator
Data Block
(FMS Alerts)
(-300)

e) ND Flight Plan
Status Block
(-300)

f) Navigation
Databases

37. Windshear Warning
and Flight Guidance
System
(Reactive)

38. Pitch Limit Indication
(PLI)

39. EFIS Speed Tape

C

2

0

C

2

1

C

2

1

C

-

-

B

1

0

C

-

0

C

2

0

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-16

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
34 - NAVIGATION							
40.	Traffic Collision and Avoidance System (TCAS) (Includes STC ST03355AT and ST03362AT)	B	-	0			
		C	-	0			
***	1) Combined Traffic Alert (TA) and Resolution Advisory (RA) Dual Display	C	2	1			
40.	Traffic Collision and Avoidance System (TCAS) (Cont'd)						
	2) Resolution Advisory (RA) Display System(s)	C	2	1			
		C	-	0			
	3) Traffic Alert (TA) Display System(s)	C	-	0			
	4) Audio Functions	B	1	0			
***	5) Airspace Selection Function	C	-	0			
41. ***	Engine Pressure Ratio Limit (EPRL) System (-100/-200)	C	1	0			
42.	Radio Magnetic Indicators (RMI)						
	1) (-100/-200)	C	-	1			
	2) (-300/-400/-500)	C	-	1			
	3) (-600/-700/-800/ -900)						
	a) EFIS/Map	C	3	1			
***	b) PFD/ND	C	1	0			

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST		
FEDERAL AVIATION ADMINISTRATION						
AIRCRAFT:			REVISION NO :		54a	PAGE:
B-737			DATE:		05/12/2010	34-17
SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
34 - NAVIGATION						
43. ***	Radio Height Alert	D	2	0		
44. ***	Head-Up Display System (HUD)	D	-	0	May be inoperative provided procedures do not require its use.	
					NOTE: Any mode which operates normally may be used.	
45. ***	Global Positioning System (GPS)	C	-	0	May be inoperative provided alternate procedures are established and used.	
		D	-	0	May be inoperative provided procedures do not require its use.	
46. ***	Microwave Landing System (MLS)	D	-	0	May be inoperative provided approach procedures do not require its use.	
47. ***	ILS Beam Deviation Lights	C	2	0	May be inoperative provided approach minimums do not require their use.	
48. ***	EFIS Control Panel					
	1) Map Switches (-300/-400/-500)					
	a) VOR/ADF	C	2	1		
	b) NAV AID	C	2	1		
	c) ARPT	C	2	1		
	d) RTE DATA	C	2	1		
	e) WPT	C	2	1		
***	2) Decision Height Reference (DH REF) Indication (-300/-400/-500)	C	2	0	May be inoperative provided: a. Approach procedures do not require its use, and b. Decision height is displayed on both EADI's	
(Continued)						

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-18

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS	
34 - NAVIGATION						
48.	EFIS Control Panel (Cont'd)					
	3) Map Switches (-600/-700/-800/ -900)					
	a) POS	C	2	1		
	b) STA	C	2	1		
	c) ARPT	C	2	1		
	d) DATA	C	2	1		
	e) WPT	C	2	1		
49.	Right IRS DC Power Supply System (-300/-400/-500/-600/-700/-800/-900)	B	1	0	(O) May be inoperative provided: a) Remaining IRS Mode Selector Unit lights are not illuminated, and b) Autopilot dual channel mode is not used during approach.	
50.	ILS System (-600/-700/-800/ -900)				Deleted in Revision 37, relief incorporated into Item 34-17.	
51. ***	Metric Altimeter	D	-	0	May be inoperative provided operations do not require its use.	
52. ***	Performance Management System (PMS) with Windshear Detection/Alerting System (STC SA2018NM)	C	-	0	(O) May be inoperative provided: a) TAT Indicator operates normally, b) PMS remains uncoupled from autopilot, c) Autothrottle system is considered inoperative, and d) Windshear Detection and Guidance is considered inoperative.	

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT:			REVISION NO :		54a		PAGE:
B-737			DATE:		05/12/2010		34-19
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2. NUMBER INSTALLED			
				3. NUMBER REQUIRED FOR DISPATCH			
					4. REMARKS OR EXCEPTIONS		
34 - NAVIGATION							
53. ***	Automatic Dependent Surveillance-Broadcast (ADS-B) System	D	-	0	May be inoperative provided it is not required by 14 CFR. NOTE: If ADS-B is installed in lieu of or as a replacement for 14 CFR required equipment, repair category in operator's MEL will be same as that of 14 CFR required equipment.		
	1) Cockpit Display and Traffic Information (CDTI)	D	-	0	NOTE: Cockpit Display Traffic Information (CDTI) display of data from other aircraft systems may be used.		
	2) CDTI Control Panel	D	-	0	May be inoperative provided: a) Flight ID can be set, and b) Screen display is acceptable to flight crew.		
	3) Data Link Transmitter(s)	D	-	0	NOTE: In some aircraft the Data Link Transmission is an integral part of the transponder and relief is provided in that section.		
	4) Data Link Receivers	D	-	0			
	5) ADS-B Applications	D	-	0			
54. ***	Integrated Standby Flight Display (ISFD) System						
	1) Attitude Display	B	1	0	May be inoperative provided: a) Operations are conducted in Day VMC only, and b) Operations are not conducted into known or forecast Over-The-Top conditions.		
	2) ILS Indication	D	1	0			
	3) Heading Display	C	1	0			
	4) Metric Altimeter Display	D	1	0	May be inoperative provided operations do not require its use.		
	5) Dedicated Battery	C	1	0			

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

34-20

SYSTEM & SEQUENCE NUMBER	ITEM	1.
--------------------------------	------	----

2.	NUMBER INSTALLED
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3.	NUMBER REQUIRED FOR DISPATCH
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4.	REMARKS OR EXCEPTIONS
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34 - NAVIGATION

55. ***	Vertical Situation Display (VSD) System (-600/-700/ -800/-900)	C	1	0	(O) May be inoperative provided alternate procedures are established and used.
		D	1	0	May be inoperative provided procedures do not require its use.
56. ***	Global Navigation Satellite Landing System (GLS) (-600/-700/-800/ -900)	D	2	-	May be inoperative provided approach minimums do not require its use.

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST			
AIRCRAFT: B-737			REVISION NO : DATE:			52 04/29/2008	PAGE: 35-1
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2. NUMBER INSTALLED			
				3. NUMBER REQUIRED FOR DISPATCH			
				4. REMARKS OR EXCEPTIONS			
35 - OXYGEN							
1.	Crew Oxygen System					Deleted prior to Revision 27.	
2.	Passenger Service Units (PSUs)	B	-	-		(M) May be inoperative provided: a) Associated seats are blocked and placarded to prevent occupancy, and b) Units operate normally for all usable lavatory and flight attendant locations.	
	1) Automatic Presentation	C	1	0		(M)(O) May be inoperative provided: a) Alternate deployment system is verified to operate normally, and b) Airplane remains at or below FL 300.	
	2) Door Latches	B	-	-		(M) Automatic opening feature of door latch(es) may be inoperative unlatched, and taped closed provided: a) PSU oxygen system operates normally, b) Flight remains at or below FL 250, and c) Passenger(s) occupying seat(s) with inoperative door latch(es) are briefed on oxygen mask procedure.	
3.	Oxygen Pressure Indicators						
	1) Flight Deck Crew Oxygen Indicator	C	1	0		(M) May be inoperative provided an alternate procedure is used to verify that oxygen supply is above minimum requirements for dispatch.	
***	2) External Service Panel Crew Oxygen Indicator	C	1	0		(M) May be inoperative provided an alternate procedure is used to verify that oxygen supply is above minimum requirements for dispatch.	
	3) Flight Deck Passenger Oxygen Indicator (-100/-200)	C	1	0		(M) May be inoperative provided an alternate procedure is used to verify that oxygen supply is above minimum requirements for dispatch.	
	4) Flight Deck Crew/Passenger Oxygen Indicator (-700C)	C	1	0		(M) May be inoperative provided an alternate procedure is used to verify that oxygen supply is above minimum requirements for dispatch.	
(Continued)							

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

35-2

SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
				3. NUMBER REQUIRED FOR DISPATCH	
					4. REMARKS OR EXCEPTIONS
35 - OXYGEN					
3. Oxygen Pressure Indicators (Cont'd)					
5) Overpressure Discharge Indication Disk	C	1	0		(O) May be damaged or missing.
4. Portable Oxygen Dispensing Units (Bottle and Mask)	D	-	-		(M) Any in excess of those required by 14 CFR may be unserviceable or missing provided:
					a) Required distribution of serviceable bottles is maintained throughout aircraft, and
					b) Bottles not properly serviced are replaced, serviced, or removed at next available maintenance facility.
5. Passenger Oxygen System	B	1	0		(M)(O) May be inoperative provided:
					a) Flight is not conducted where minimum enroute altitude is above 14,000 feet MSL,
					b) Both air conditioning packs operate normally,
					c) Remaining components of pressurization system operate normally,
					d) Airplane remains at or below FL 250,
					e) Portable oxygen units are provided for 10% of passengers, and
					f) Passengers are appropriately briefed.
	C	1	0		May be inoperative for all-cargo configuration.
	B	1	0		May be inoperative provided flight is conducted at or below 10,000 feet MSL.
6. PBE Smoke Hoods	D	-	-		Any in excess of those required by 14 CFR may be inoperative.
7. External Service Panel, Oxygen Fill Station	C	1	0		(M) May be inoperative provided leak-tight integrity of oxygen supply system is not affected.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

36-1

SYSTEM &
SEQUENCE
NUMBER

ITEM

1.

2.

NUMBER INSTALLED

3. NUMBER REQUIRED FOR DISPATCH

4. REMARKS OR EXCEPTIONS

36 - PNEUMATICS

1. Manifold Isolation
Shutoff Valve

1) (-100/-200)

C

1

0

(M) May be inoperative provided:

- a) Valve remains closed except for engine start, and
- b) Airplane is not operated in known or forecast icing conditions.

2) (-300/-400/-500)

C

1

0

(M) May be inoperative provided:

- a) Modified Main Engine controls or production equivalent have been installed,
- b) Valve remains closed except for engine start, and
- c) Airplane is not operated in known or forecast icing conditions.

3) (-600/-700/-800/
-900)

C

1

0

(M) Except for ER operations beyond 120 minutes, may be inoperative provided:

- a) Valve remains closed except for engine start, and
- b) Airplane is not operated in known or forecast icing conditions.

2. Ground Pneumatic
Connector Check
Valve

C

1

0

(M)(O) Except for ER operations beyond 120 minutes, may be inoperative open provided:

- a) Right pneumatic manifold remains depressurized after starting right engine,
- b) Airplane is not operated in known or forecast icing conditions, and
- c) Altitude remains at or below FL 250.

C

1

0

May be inoperative closed.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

36-2

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		
				3.	NUMBER REQUIRED FOR DISPATCH	
					4.	REMARKS OR EXCEPTIONS
36 - PNEUMATICS						
3.	Precooler Control Valves					
	1) (-100/-200)	C	2	0	(M)(O)	May be inoperative provided: a) Associated engine bleed shutoff valve remains closed after engine start, and b) Airplane is not operated in known or forecast icing conditions.
	2) (-300/-400/-500/-600/-700/-800/-900)	C	2	0	(O)	Except for ER operations beyond 120 minutes, may be inoperative in any position provided: a) Associated engine bleed shutoff valve remains closed, and b) Airplane is not operated in known or forecast icing conditions.
	3) (-300/-400/-500)	C	2	0	(M)	Except for ER operations beyond 120 minutes, may be inoperative full open provided airplane is not operated in known or forecast icing conditions
4.	Pneumatic Pressure Indication Systems	C	2	0	(O)	May be inoperative provided alternate procedures are established and used.
5.	Engine Bleed Air Shutoff Valves (PRSOV)					
	1) (-100/-200)	C	2	0	(M)(O)	May be inoperative provided: a) Valve is secured closed after engine start, and b) Airplane is not operated in known or forecast icing conditions.
	2) (-300/-400/-500/-600/-700/-800/-900)	C	2	0	(M)(O)	Except for ER operations beyond 120 minutes, may be inoperative provided: a) Valve is secured closed before engine start, and b) Airplane is not operated in known or forecast icing conditions.
6.	Dual Bleed Light System	C	1	0	(O)	May be inoperative provided: a) APU bleed air is not used in flight, and b) APU bleed valve is closed before each departure.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

36-3

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

36 - PNEUMATICS

7. 13th Stage Bleed Air
Modulating and
Shutoff Valves
(-100/-200) C

2

0

(M) May be inoperative provided airplane is not
operated in known or forecast icing conditions.8. Engine Bleed Trip
Off Lights C

2

0

(O) Except for ER operations beyond 120 minutes,
may be inoperative provided:
a) Associated engine bleed is not used except
for engine start, and
b) Airplane is not operated in known or forecast
icing conditions.9. High Stage Valves
(-300/-400/-500/
-600/-700/-800/
-900) C

2

1

(M) One may be inoperative locked closed provided a
minimum of 60% N1 is maintained on associated
engine during flight in icing conditions.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

38-1

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**

38 - WATER / WASTE

1. Potable Water
Systems

C

-

-

(M) Individual components may be inoperative provided:

- a) Associated components are deactivated or isolated, and
- b) Associated system components are verified not to have leaks.

NOTE: Any portion of system which operates normally may be used.

C

-

-

(M) May be inoperative provided:

- a) System is drained, and
- b) Procedures are established to ensure that system is not serviced.

2. Lavatory Waste
Systems (Including
Wheelchair
Accessible
Lavatories)

C

-

-

(M) Individual components may be inoperative provided:

- a) Associated components are deactivated or isolated, and
- b) Associated system components are verified not to have leaks.

NOTE: Any portion of system which operates normally may be used.

C

-

-

(M) Associated lavatory system(s) may be inoperative provided:

- a) Associated components are deactivated or isolated to prevent leaks,
- b) Pilot-in-Command will determine if flight duration is acceptable with a forward lavatory unusable, and
- c) Associated lavatory door(s) is secured closed and placarded "INOPERATIVE – DO NOT ENTER".

NOTE: These provisions are not intended to prohibit inspections by crewmembers.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

46-1

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
46 – INFORMATION SYSTEMS					
1.	***	Electronic Flight Bag (EFB) System			
		1) Class 3 EFB (Boeing)	C 2	1	
			C 2	0	(O) May be inoperative provided alternate procedures are established and used.
			D 2	0	NOTE: Any function, program or document which operates normally may be used.
			D 2	0	May be inoperative provided procedures do not require its use.
		2) (STC ST03165AT Only)	D 2	0	(M) May be inoperative provided procedures do not require its use.
			C 2	0	(M)(O) May be inoperative provided alternate procedures are established and used.
		a) Mounting Cradle	C 2	1	(M)(O) May be inoperative provided alternate procedures are established and used.
		b) Liquid Crystal Display	C 2	1	One may be inoperative provided alternate source for required information is available and used.
		c) Control Panel Module/ Peripheral Connectivity Unit	C 2	1	One may be inoperative provided alternate source for required information is available and used.
		(1) ON/OFF Switch	C 2	1	One may be inoperative in ON position provided: a) EFB Battery charging system operates normally, and b) Normal power to unit is available and operates normally.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

53

PAGE:

DATE:

08/01/2009

46-2

SYSTEM & SEQUENCE NUMBER		1.	2.		NUMBER INSTALLED	
		ITEM			3.	
					NUMBER REQUIRED FOR DISPATCH	
					4.	
					REMARKS OR EXCEPTIONS	
46 – INFORMATION SYSTEMS						
1.	Electronic Flight Bag (EFB) System (Cont'd)					

	2) (STC ST03165AT Only) (Cont'd)					
	d) Computer Processing Unit (CPU)	C	2	1	One may be inoperative provided alternate source for required information is available and used.	
	(1) Back-Up Battery	C	2	1	One may be inoperative provided normal power is available and operates normally.	
	e) Standby Button	C	2	0	May be inoperative in operational mode.	
		C	2	0	May be inoperative in Standby mode provided display is considered inoperative.	
	3) Stowage/Charger Assembly (STC ST01118CH Only)					
	a) Class 1 EFB w/ All Battery Types	D	1	0	May be inoperative provided procedures do not require its use.	
	b) Class 1 EFB w/ Lithium Ion Battery	C	1	0	(M)(O) May be inoperative provided alternate procedures are established and used.	
					NOTE: If a Class 1 EFB is to be used, alternate procedures must insure the battery is charged to a "sufficiently charged" state at appropriate time intervals.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

47-1

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**

47 – INERT GAS SYSTEM

1. Nitrogen Generation
*** System (NGS) (All
Models)1) Serial Number D 1 0
(S/N) 34333 or
34450 (prior to
incorporation of
Boeing Service
Bulletin 737-47-
1006)a) Nitrogen D 1 0
Generation
Degraded2) All Models (upon A 1 0
incorporation of
Boeing Service
Bulletin 737-47-
1002, 737-47-
1003, 737-47-
1004, 737-47-
1005, 737-47-
1006 or
production
equivalent)a) Nitrogen C 1 0
Generation
Degraded(M) May be inoperative provided NGS shutoff valve is
deactivated closed.(M) May be inoperative provided:
(a) NGS shutoff valve is deactivated closed, and
(b) Repairs are made within ten flight days.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT: B-737	REVISION NO : DATE:	54a 05/12/2010	PAGE: 49-1
---------------------------	-------------------------------	-------------------	----------------------

SYSTEM & SEQUENCE NUMBER			ITEM	1.	2.	NUMBER INSTALLED	
						3.	NUMBER REQUIRED FOR DISPATCH
							4. REMARKS OR EXCEPTIONS
49 - AIRBORNE AUXILIARY POWER							
1.	Auxiliary Power Unit (APU)	C	1	0			(O) Except for ER operations, may be inoperative provided: a) Procedures do not require its use, and b) Visual confirmation is made that no damage has occurred to APU exhaust area.
2.	APU Annunciator LOW OIL PRESSURE and OVER SPEED Lights	C	2	0			May be inoperative provided APU Auto Shutdown System operates normally.
3.	APU Auto Shutdown System (-100/-200/-300/-400/-500)	C	1	0			(M) Except for ER operations, may be inoperative provided: a) APU is not used in flight, b) APU annunciator lights operate normally, and c) APU is monitored continuously.
4.	APU Annunciator LOW OIL QUANTITY/MAINT Light	C	1	0			(M) May be inoperative and APU used provided oil quantity is checked once each flight day.
5.	APU EGT Indicator						
	1) Model GTCP85-129	C	1	0			(O) Except for ER operations, may be inoperative provided: a) All warning and caution lights operate normally, b) APU is used to supply electrical power, and for starting one engine only, and c) Passengers are not permitted on board until APU has been shut down.
	2) Model GTCP36-280, APS-2000 and AS 131-9B	C	1	0			
6.	APU Inlet Door	C	1	0			(O) May be inoperative open.
		C	1	0			(O) Except for ER operations, may be inoperative in any other position if APU is not used.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

49-2

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
49 - AIRBORNE AUXILIARY POWER						
7.	APU Bleed Air System	C	1	0		(M) May be inoperative closed. NOTE: APU may be used to provide electrical power.
		C	1	0		(O) Except for ER operations, may be inoperative provided: a) APU bleed air check valve operates normally, and b) APU is not operated.
8. ***	APU DC Fuel Boost Pump	D	1	0		
9.	APU Surge Control System					
***	1) Surge Bleed Valve (Models GTCP85-129 and APS-2000) (-100/-200/-300/-400/-500)	C	1	0		May be inoperative in open position provided APU bleed air is not used for engine start on ground. NOTE: Relief also applies to airplanes modified by STC SA5730NM or ST00131SE provided APU is not operating during approach.
		C	1	0		May be inoperative in closed position provided APU operation is limited to FL 250 or below. NOTE: Relief also applies to airplanes modified by STC SA5730NM or ST00131SE.
	2) Surge Control Valve (Model AS 131-9B) (-600/-700/-800/-900)	C	1	0		May be inoperative in open position provided APU bleed air is not used. NOTE: APU may be used to provide electrical power.
		C	1	0		(O) Except for ER operations, may be inoperative in closed position provided APU is not used.
10. ***	APU Cockpit Hourmeter (-100/-200/-300/-400/-500)	D	1	0		
11. ***	APU Start Counter Meter (-100/-200/-300/-400/-500)	D	1	0		

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST	
FEDERAL AVIATION ADMINISTRATION					
AIRCRAFT: B-737			REVISION NO : 54a DATE: 05/12/2010		PAGE: 49-3
SYSTEM & SEQUENCE NUMBER			1. ITEM		
			2. NUMBER INSTALLED		
			3. NUMBER REQUIRED FOR DISPATCH		
			4. REMARKS OR EXCEPTIONS		
49 - AIRBORNE AUXILIARY POWER					
12.	APU Annunciator HIGH OIL TEMP/ FAULT Light	C	1	0	
13. ***	APU Fuel Heater (-100/-200/-300/-400/-500)	C	1	0	(M) May be inoperative provided APU operates normally.
14. ***	APU Flap Indicator Interlock System (-100/-200 Modified by STC SA5730NM or ST00131SE)	C	1	0	(O) May be inoperative provided: a) Remaining APU surge bleed valve is operating, and b) APU bleed air is used during approach.
		C	1	0	(O) May be inoperative provided APU is not operating during approach.
15.	Start Power Unit (-600/-700/-800/-900)	C	1	0	(M) Except for ER operations, may be inoperative provided procedures do not require use of APU.
	1) AC/DC Start Systems	C	2	1	
16.	Start Converter Unit (-600/-700/-800/-900)	C	1	0	(M) Except for ER operations, may be inoperative provided procedures do not require use of APU.
	1) Voltage Regulator Function	C	1	0	Except for ER operations, may be inoperative provided APU generator is not used for electrical power.
NOTE: APU may be used as a pneumatic source.					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

52-1

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
52 - DOORS						
1. ***	Forward Air Stair	D	1	0		NOTE: Any mode that operates normally may be used.
2. ***	Aft Air Stair (-100/-200)	C	1	1		Electrical mode may be inoperative provided door operates normally as an emergency exit in passenger configuration.
		D	1	0		May be inoperative in all-cargo configuration only.
3.	Door Warning Light System					
	1) Entry/Service/Cargo/Equipment/Airstair	C	-	0		(M) May be inoperative provided associated door is verified closed and locked. NOTE: On -600/-700/-800/-900, if two or more entry/service door warning lights are inoperative due to failed door sensors, overwing exit flight lock system and mid exit flight lock system (-900ER) will not function properly. Refer to MMEL item 52-15
	2) Overwing (-600/-700/-800/-900)	C	-	0		(M) May be inoperative provided: a) Associated door is verified closed and latched, and b) Associated flight lock is verified to operate normally.
	3) Cabin Door Indication System (-800EF STC ST02000NY Only)	C	1	0		(O) May be inoperative provided associated doors are verified in accordance with following prior to taxi, takeoff, and landing; - Entry Area/Main Lounge is Open - Private Bedroom is Closed - Guest Lavatory is Closed - Aft Lounge/Galley is Open
***	4) Mid-Exit (-900ER)	C	1	0		(M) May be inoperative provided associated door is verified closed and latched.

MASTER MINIMUM EQUIPMENT LIST

PAGE:

52-2

SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED		3.	NUMBER REQUIRED FOR DISPATCH		4.	REMARKS OR EXCEPTIONS
52 - DOORS											
4. ***	Tire Burst Screen Warning Light System (-100/-200/-300)	C	1	0						(M) May be inoperative provided: a) Main wheel well screens are inspected for security and damage before each departure, and b) For combined Equipment/Tire Burst Screen Warning Light, visually verify that electronics compartment and lower nose compartment is secured and locked, and main wheel well screen is secured and undamaged before each departure.	
5.	Left Main Cabin Door Pressure Stop Fittings										
	1) Aft Airstair Door and Forward Entry Door	B	-	-						(M)(O) One per door may be broken or missing provided: a) There are no visible defects on other fittings for associated door, b) Pressure differential does not exceed 6.0 psi, and c) Analog cabin pressure control system standby control mode operates normally and STBY is used.	
		B	-	-						(M)(O) One per door may be broken or missing provided: a) There are no visible defects on other fittings for associated door, b) Pressure differential does not exceed 6.0 psi, c) Digital cabin pressure control system AUTO or ALTN control mode operates normally, and d) Alternate procedures are established and used.	
	2) Aft Door Without Airstairs	B	-	-						(M)(O) One per door may be broken or missing provided: a) There are no visible defects on other fittings for associated door, b) Pressure differential does not exceed 3.4 psi, and c) Analog cabin pressure control system standby control mode operates normally and STBY is used.	
(Continued)											

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737		REVISION NO : 54 DATE: 10/12/2009		PAGE: 52-3	
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
52 - DOORS				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
5.	Left Main Cabin Door Pressure Stop Fittings (Cont'd)				
	2) Aft Door Without Airstairs (Cont'd)	B	-	-	(M)(O) One per door may be broken or missing provided: a) There are no visible defects on other fittings for associated door, b) Pressure differential does not exceed 3.4 psi, c) Digital cabin pressure control system AUTO or ALTN control mode operates normally, and d) Alternate procedures are established and used.
6.	Lower Cargo Doors Pressure Stop Fittings				
	1) (All Models)	A	24	22	(M) Any one may be broken or missing on each door or frame provided: a) No defects are visible on other fittings for associated door, b) Cabin pressure controller AUTO mode operates normally, c) Adjacent stop fittings are inspected within 25 flights, and d) Not more than 50 flights are made before completion of repairs or replacements.
	2) (-100/-200/-300/-400/-500/-900)	C	24	20	(M)(O) Two may be broken or missing on each door or frame provided airplane is operated in an unpressurized configuration only.
	3) (-600/-700/-800 prior to incorporation of Boeing Service Bulletin 737-21-1135, 737-26-1121, and 737-26-1122, or production equivalent)	C	24	20	(M)(O) Two may be broken or missing on each door or frame provided airplane is operated in an unpressurized configuration only.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

52-4

SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
52 - DOORS				3. NUMBER REQUIRED FOR DISPATCH	
				4. REMARKS OR EXCEPTIONS	
6. Lower Cargo Doors Pressure Stop Fittings (Cont'd)					
4) (-600/-700/-800	C	24	20		
upon incorporation of Boeing Service Bulletin 737-21-1135, 737-26-1121, and 737-26-1122, or production equivalent)					

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST	
AIRCRAFT: B-737		REVISION NO : 54 DATE: 10/12/2009		PAGE: 52-5	
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED	
52 - DOORS				3.	NUMBER REQUIRED FOR DISPATCH
					4. REMARKS OR EXCEPTIONS
10.	Main Cabin Cargo Door (PEMCO Aeroplex, Inc. STC SA2969SO)				
	1) Latch Pin, Latch Base and Lower Jamb Latch Fitting	A	8	7	(M)(O) One may be broken or missing from main cargo door provided: a) A visual check is made before departure to ensure no defects are visible on other latch bases, pins or lower jamb latch fittings, b) Latch pin and latch base of damaged latch does not interfere with continuous safety operation of remaining latches and pins, c) Flight is conducted in unpressurized configuration, d) Procedures are established and used to ensure main and lower lobe cargo compartments remain empty, or are verified to contain only empty cargo handling equipment, ballast (ballast may be loaded in ULDs), and/or Fly Away Kits. e) Repairs are made within two flight days.
	2) Hydraulic Cylinder Latching Mechanism	B	2	1	(M) One may be inoperative provided remaining latch cylinder is operative through gear box.
		C	2	0	(M) May be inoperative provided door may be latched and unlatched manually.
	3) Hydraulic System Control Valve	B	1	0	(M) May be inoperative provided door may be locked and unlocked manually.
	4) Lifting Actuator Assembly	B	2	0	(M) May be inoperative provided door is verified latched and locked.
	5) Double Piloted Check Valve	B	1	0	(M) May be inoperative provided door may be locked and unlocked manually.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

52-6

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
52 - DOORS						
10.		Main Cabin Cargo Door (PEMCO Aeroplex, Inc. STC SA2969SO)(Cont'd)				
	6)	Lock, Lock Mount and Locking Fittings	A	2	1	(M)(O) One may be inoperative provided: a) No defects are visible on remaining lock or lock mount of associated door, and b) Repairs are made within two flight days.
	7)	Sequence Valves	B	2	0	(M) May be inoperative provided door is verified latched and locked.
	8)	Priority Valve	B	1	0	(M) May be inoperative provided door is verified latched and locked.
	9)	Hydraulic Lock Actuators	C	2	0	(M) May be inoperative provided door can be unlocked and unlatched manually.
11.		Main Cargo Door Electrically Powered Hydraulic Pump (Standalone Hydraulic System Only) (PEMCO Aeroplex, Inc. STC SA2969SO)	C	1	0	(M) May be inoperative provided door is closed, latched, and locked before each departure.
12.		Main Cargo Door Hydraulic Hand Pump (PEMCO F, QC and COMBI models only)	C	1	0	(M) May be inoperative.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST					
FEDERAL AVIATION ADMINISTRATION									
AIRCRAFT:				REVISION NO :		54		PAGE:	
B-737				DATE:		10/12/2009		52-7	
SYSTEM & SEQUENCE NUMBER		ITEM		1.		2.		NUMBER INSTALLED	
						3.		NUMBER REQUIRED FOR DISPATCH	
								4. REMARKS OR EXCEPTIONS	
52 - DOORS									
13.	Main Cargo Door Lift/Operating Systems								
	1) Electric and/or Manual Mode (-200C and STC SA2969SO)	C	-		1			One may be inoperative provided remaining mode operates normally.	
		C	-		0			(M) May be inoperative provided door is verified closed and locked before each departure.	
	2) Electric Mode (-700C)	C	1		0			(M) May be inoperative provided Manual Mode is verified to operate normally.	
	3) Hydroelectric and/or Manual Mode (STC's ST01566LA, ST00287AT, ST01827LA, and ST01961SE)	C	2		1			One may be inoperative provided remaining mode operates normally.	
	a) (STC ST01566LA and ST01961SE)	C	2		0			(M) May be inoperative provided door is verified closed, latched and locked before each departure.	
14. ***	Lower Cargo Doors Hold Open Mechanism/Device	C	2		0			(M) May be inoperative provided Door Balance Mechanism operates normally.	
		C	2		0			May be inoperative provided cargo compartment remains empty.	
15.	Flight Lock System								
	1) Overwing Exit (-600/ -700/-800/ -900)	C	-		0			(M)(O) May be inoperative provided: a) Each affected exit is verified to be capable of being unlatched and opened before each departure, and b) A person employed by operator is designated to remain seated in passenger seat nearest affected exit when cabin differential pressure is less than 4.0 psi.	
(Continued)									

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

52-8

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED			
			3. NUMBER REQUIRED FOR DISPATCH			
			4. REMARKS OR EXCEPTIONS			
52 - DOORS						
15.	Flight Lock System (Cont'd)					
***	2) Mid Exit (-900ER)	C	-	0	(M)(O) May be inoperative provided: a) Each affected exit is verified to be capable of being unlatched and opened before each departure, and b) A person employed by operator is designated to remain seated in passenger seat nearest affected exit when cabin differential pressure is less than 4.0 psi.	
16.	Main Cabin Exit/Slide (All Cargo Configuration)	C	-	0	All doors/slides in cargo area except L1/R1 may be inoperative or slide missing without restriction.	
		B	-	1	L1 may be inoperative or slide missing provided R1 operates normally.	
		B	-	1	R1 may be inoperative or slide missing provided L1 operates normally.	
		B	-	0	May be inoperative or slide missing provided: a) Only essential crew members including official observer(s) in observer seat(s) are allowed on the flight, and b) An alternate means of egress is available.	

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST		
FEDERAL AVIATION ADMINISTRATION						
AIRCRAFT:			REVISION NO :		54	PAGE:
B-737			DATE:		10/12/2009	52-9
SYSTEM & SEQUENCE NUMBER	ITEM	1.	2.	NUMBER INSTALLED		
				3.	NUMBER REQUIRED FOR DISPATCH	
					4.	REMARKS OR EXCEPTIONS
52 - DOORS						
17. ***	Boeing/C&D Aerospace Enhanced Flight Deck Security Door Automatic Locking System (14 CFR 25.795 Compliant)	C	1	0	(M)(O)	May be inoperative provided: a) Automatic locking system is deactivated, b) Door dead bolt operates normally and is used to lock door, c) Alternate procedures are established and used for locking and unlocking door using dead bolt.
	1) Flight Deck Access Panel System (Keypad, Door Chime)	C	1	0	(M)(O)	May be inoperative provided: a) Keypad is deactivated, and b) Alternate procedures are established and used
	a) LEDs	C	3	0	(O)	May be inoperative provided alternate procedures are established and used.
***	b) Door Bell Mode	C	1	0	(O)	May be inoperative provided alternate procedures are established and used.
	c) Switch Guard	C	1	0		May be inoperative or missing provided flight deck door LOCK FAIL light operates normally.
	2) Flight Deck Door LOCK FAIL Light	C	1	0	(M)	May be inoperative provided automatic lock controls are verified to operate normally.
	3) Flight Deck Door AUTO UNLK Light	C	1	0	(M)	May be inoperative provided: a) Automatic lock controls are verified to operate normally, and b) Door chime operates normally.
	4) Fight Deck Door Lock Control Selector	C	1	0	(M)(O)	May be inoperative provided: a) Keypad is deactivated, b) Automatic lock is verified to operate normally, and c) Alternate procedures are established and used.
	5) Flight Deck Door Pressure Relief Panels					Item moved to 52-20, Revision 46.

U.S. DEPARTMENT OF TRANSPORTATION					
FEDERAL AVIATION ADMINISTRATION			MASTER MINIMUM EQUIPMENT LIST		
AIRCRAFT: B-737			REVISION NO : 54 DATE: 10/12/2009		PAGE: 52-10
SYSTEM & SEQUENCE NUMBER			1.	2. NUMBER INSTALLED	
ITEM				3. NUMBER REQUIRED FOR DISPATCH	
				4. REMARKS OR EXCEPTIONS	
52 - DOORS					
18. ***	Boeing/C&D Aerospace Enhanced Flight Deck Security Door Dead Bolt (14 CFR 25.795 Compliant)	C	1	0	May be inoperative provided automatic lock controls operate normally.
19. ***	JAMCO Flight Deck Security Door Automatic Locking System (14 CFR 25.795 Compliant)	C	1	0	(M)(O) May be inoperative provided: a) Automatic locking system is deactivated, b) Mechanical Catch (Latch) Pin operates normally and is used to lock door, and c) Alternate procedures are established and used for locking and unlocking flight deck door using Mechanical Catch (Latch) Pin.
	1) Door Automatic Locking Solenoid	C	2	1	(M) One may be inoperative provided remaining locking solenoid operates normally.
	2) Door Warning System				
***	a) Speakers	C	2	1	(M)(O) One may be inoperative provided remaining speaker is verified to operate normally once each flight day.
***	b) LED (Green Indicator Light)	C	2	0	
***	c) Aural Warning System	C	1	0	(M)(O) May be inoperative provided: a) AUTO UNLK Light is verified to operate normally, and b) Alternate procedures are established and used.
	3) Door Control Panel				
***	a) Door LOCK FAIL Light	C	1	0	(M) May be inoperative OFF provided automatic lock controls are verified to operate normally.
(Continued)					

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST				
FEDERAL AVIATION ADMINISTRATION								
AIRCRAFT:			REVISION NO :		54		PAGE:	
B-737			DATE:		10/12/2009		52-11	
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2. NUMBER INSTALLED				
				3. NUMBER REQUIRED FOR DISPATCH				
					4. REMARKS OR EXCEPTIONS			
52 - DOORS								
19.	JAMCO Flight Deck Security Door Automatic Locking System (14 CFR 25.795 Compliant) (Cont'd)							
	3) Door Control Panel (Cont'd)							
***	b) Door AUTO UNLK Light	C	1	0	(M)(O) May be inoperative OFF provided: a) Automatic lock controls are verified operate normally, b) Aural Warning system operates normally, and c) Alternate procedures are established and used.			
***	c) Door HARD LOCK Light	C	1	0	(M)(O) May be inoperative provided: a) Automatic lock controls are verified to operate normally, and b) Alternate procedures are established and used.			
***	d) Door UNLKD Switch/UNLK Switch Position	C	1	0	(M)(O) May be inoperative provided: a) Door can be opened manually from flight deck, b) Remaining automatic lock controls are verified to operate normally, and c) Alternate procedures are established and used.			
***	e) Door UNLKD Light	C	1	0	(M)(O) May be inoperative provided: a) Automatic lock controls are verified to operate normally, and b) Aural warning system operates normally.			
	f) Door EMRG ENTRY ACTIVE Light	C	1	0	(M) May be inoperative provided door aural warning system is verified to operate normally.			
	g) Door OPEN Light	C	1	0	(M)(O) May be inoperative provided Automatic Lock controls are verified to operate normally.			
	4) FLIGHT DECK DOOR Warning/ Caution Light	C	1	0				

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

52-12

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH		4. REMARKS OR EXCEPTIONS
52 - DOORS						
19.		JAMCO Flight Deck Security Door Automatic Locking System (14 CFR 25.795 Compliant) (Cont'd)				
***	5)	Cabin Pushbutton Entry Pad/Keypad	C	1	0	(O) May be inoperative provided alternate procedures are established and used.
	a)	Keypad Indicator Lights	C	3	0	(M)(O) May be inoperative provided: a) Keypad is verified to operate normally, and b) Alternate procedures are established and used.
20.		Flight Deck Door Pressure Relief Panels				
***	1)	JAMCO Flight Deck Security Door Pressure Relief Latches (14 CFR 25.795 Compliant)	A	3	0	May be inoperative in latched position provided repairs are made within two flight days.
***	2)	Boeing/C&D Aerospace Enhanced Flight Deck Security Door (14 CFR 25.795 Compliant)	A	2	0	May be inoperative provided: a) Panels are in latched position, and b) Repairs are made within two flight days.
21.		JAMCO Flight Deck Security Door Mechanical Catch Pin Lock (14 CFR 25.795 Compliant)	C	1	0	(M) May be inoperative provided automatic lock system is verified to operate normally.

22.		Flight Deck Door Hold Open Device (e.g. Door Stop, Foot Plunger, etc.)	D	1	0	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54

PAGE:

DATE:

10/12/2009

52-13

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**

52 - DOORS

23.
***Flight Deck Door
Viewing Port

A

1

0

(O) May be inoperative provided:

a) Alternate procedures are established and
used.

b) Repairs are made within three flight days.

C

1

0

(O) May be inoperative provided:

a) An electronic flight deck door visual surveillance
system is installed and operates normally, and

b) Alternate procedures are established and used.

1) All Cargo
Configuration

C

1

0

May be inoperative provided courier/supernumerary
compartment remains empty.

D

1

0

May be inoperative provided procedures do not require
its use.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT:			REVISION NO :		54a		PAGE:
B-737			DATE:		05/12/2010		73-1
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2. NUMBER INSTALLED			
				3. NUMBER REQUIRED FOR DISPATCH			
				4. REMARKS OR EXCEPTIONS			
73 - ENGINE FUEL & CONTROL							
1.	Fuel Heater Timers (-100/-200)	C	2	1	(O) One may be inoperative provided associated fuel heater VALVE OPEN light operates normally.		
2.	Fuel Heater Valves (-100/-200)	C	2	0	(M)(O) May be inoperative closed provided fuel temperature is maintained at or above 32 degrees F (0 degrees C).		
3.	Fuel Heater VALVE OPEN Lights (-100/-200)	C	2	0	(M) May be inoperative provided valve is verified to operate normally before each departure.		
		C	2	0	(O) May be inoperative provided fuel temperature is maintained at or above 32 degrees F (0 degrees C).		
4.	Fuel Filter Differential Pressure Warning Systems						
	1) (-100/-200)	C	2	1	(O) May be inoperative provided fuel heater system is checked to operate normally.		
	2) (-300/-400/-500/-600/-700/-800/-900)	C	2	1	(M) May be inoperative provided malfunction is verified to be in warning system.		
5.	Fuel Flow Indication Systems	C	2	1	One may be inoperative provided: a) N1, N2 for associated engine operate normally, and b) Both main tank fuel quantity indicators operate normally.		
6. ***	Fuel Used Indicators	C	2	0			
7.	Power Management Control (PMC) Systems (-300/-400/-500)	C	2	0	(O) May be inoperative provided: a) Both PMC's remain OFF, and b) AFM Appendix performance adjustments are applied.		
8.	Power Management Control (PMC) INOP Lights (-300/-400/-500)	C	2	0	(O) May be inoperative provided: a) Both PMC's remain OFF, and b) AFM Appendix performance adjustments are applied.		
9.	Low Idle Altitude Switch (-400)				Deleted in Revision 30.		

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

54a

PAGE:

DATE:

05/12/2010

73-2

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

73 - ENGINE FUEL & CONTROL

10. Fuel Control ENG
VALVE CLOSED
Indicating System
(-600/-700/-800/
-900)

C

2

0

(M) May be inoperative provided associated valve is
verified to operate normally.

11. Electronic Engine
Control (EEC)
(-600/-700/-800/
-900)

1) Normal (ON)
Mode

C

2

0

(O) May be inoperative provided:

- a) Both engines are operated in ALTERNATE mode,
- b) Strut/Wing leading edge over-braided wire bundles are installed per Boeing Service Bulletin or production equivalent, and
- c) Applicable AFM performance adjustments are applied.

12. Electronic Engine
Control (EEC)
Alternate Power
Supply System
(-600/-700/-800/
-900)

A

4

3

(M) May be inoperative deactivated provided repairs are made in accordance with the times established in Boeing Maintenance Planning Data document, D626A001, Section 1, Items 73-020-01 and 73-020-02.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST			
FEDERAL AVIATION ADMINISTRATION							
AIRCRAFT: B-737			REVISION NO : 52 DATE: 04/29/2008		PAGE: 74-1		
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED		
					3. NUMBER REQUIRED FOR DISPATCH		
					4. REMARKS OR EXCEPTIONS		
74 - ENGINE IGNITION							
1.	Ignition Systems						
	1) (-100/-200)						
	a) High Energy System (Twin 20 Joule)	C	4	2	Except for ER operations, left igniter may be inoperative on each engine.		
	b) Low Energy System (4 Joule)	C	2	0	(O) May be inoperative provided switching is available to permit selection of operative high energy system for continuous ignition.		
	2) (-300/-400/-500/-600/-700/-800/-900)						
	a) Left Ignition Systems	B	2	1	One may be inoperative provided: a) Ignition Select Switch remains in BOTH position, and b) Right ignition systems operate normally.		
		C	2	0	(O) Except for ER operations, may be inoperative provided: a) Ignition Select Switch remains in BOTH position, and b) Associated engine right ignition system operates normally.		
	b) Right Ignition Systems	B	2	1	(M)(O) One may be inoperative provided: a) Ignition Select Switch remains in BOTH position, b) Left ignition systems operate normally, and c) Associated engine left igniter is connected to AC Standby Bus by an acceptable configuration.		
		C	2	0	(M)(O) Except for ER operations, may be inoperative provided: a) Ignition Select Switch remains in BOTH position, b) Associated engine left ignition systems operate normally, and c) Associated engine left igniter is connected to AC Standby Bus by an acceptable configuration.		

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

75-1

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3. NUMBER REQUIRED FOR DISPATCH****4. REMARKS OR EXCEPTIONS**

75 – BLEED AIR

1. Gravel Protection
*** System (-100/-200)

D

1

0

(M) Valves may be inoperative closed provided
operations do not require its use.2. High Pressure
*** Turbine Clearance
Control (HPTCC)
Timer(s) (-300/
-400/-500)

C

2

0

(M) May be inoperative provided system(s) are
deactivated.

U.S. DEPARTMENT OF TRANSPORTATION				MASTER MINIMUM EQUIPMENT LIST		
FEDERAL AVIATION ADMINISTRATION						
AIRCRAFT:			REVISION NO :		52	PAGE:
B-737			DATE:		04/29/2008	77-1
SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		
				3.	NUMBER REQUIRED FOR DISPATCH	
					4. REMARKS OR EXCEPTIONS	
77 - ENGINE INDICATING						
1.	Engine Pressure Ratio Systems (-100/-200)					
	1) Digital Counters	C	2	0		
	2) EPR Reference Selectors	C	2	1		
2.	N1 Tachometers					
	1) (-100/-200)	B	2	1	(O) One may be inoperative provided N2 and fuel flow indicator for associated engine operates normally.	
***	a) Digital Counters	B	2	0	NOTE: An indicator with an operating pointer is considered to operate normally.	
	2) (-300/-400/-500/-600/-700/-800/-900)					
	a) Digital Counters	B	2	0	(O) Except for EIS/CDS equipped airplanes, may be inoperative provided autothrottle is used for takeoff thrust setting.	
					NOTE: An indicator with an operating pointer is considered to operate normally.	
	b) Reference N1 Bugs	C	2	1		
	d) Manual Set Indication	C	2	0		
***	3) N1 Warning Lights (-100/-200/-300/-400/-500)	B	2	0	May be inoperative provided associated N1 pointer operates normally.	

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

77-2

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED	
			3.	NUMBER REQUIRED FOR DISPATCH	
				4.	REMARKS OR EXCEPTIONS
77 - ENGINE INDICATING					
3.	N2 Tachometers				
	1) (-100/-200)	B	2	1	(O) One may be inoperative provided: a) N1 and fuel flow indicators for associated engine operate normally, and b) An alternate starting procedure is established and used.
	2) (-300/-400/-500)	B	2	1	(O) One may be inoperative provided: a) N1 and fuel flow indicators for associated engine operate normally, b) An alternate starting procedure is established and used, and c) Engine #1 N2 tach generator operates normally.
***	3) Digital Counters	C	2	0	May be inoperative except for EIS/CDS equipped airplanes. NOTE: An indicator with an operating pointer is considered to operate normally.
***	4) N2 Warning Lights (-100/-200/-300/-400/-500)	B	2	0	May be inoperative provided associated N2 pointer operates normally.
4.	Fuel Flow Meters				Moved to Item 73-5 prior to Revision 30.
5.	Vibration Indicating Systems				
***	1) (-100/-200)	C	2	0	
	2) (-300/-400/-500/-600/-700/-800/-900)	C	2	1	
6.	EGT Indications				
***	1) Digital Counters	C	2	0	May be inoperative except for EIS/CDS equipped airplanes. (Continued)

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

77-3

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

77 - ENGINE INDICATING

6. EGT Indications
(Cont'd)*** 2) EGT Warning
Lights (-100/
-200/-300/-400/
-500)

C

2

0

May be inoperative provided associated EGT pointer
operates normally.

7. EPR Computer

Moved to Item 34-41 in Revision 30.

8. Fuel Used
Indicators

Moved to Item 73-6 prior to Revision 30.

9. Abnormal Start
*** Indication Systems
(-300/-400/-500/
-600/-700/-800/
-900)

C

2

0

10. LOW IDLE Light
(-300/-400/-500)

B

1

0

(M) May be inoperative provided:
a) Engine idle control system is verified to
operate normally, and
b) Both engines installed are "modified" engines
(Boeing SB 737-77-1031 or production
equivalent).

U.S. DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION				MASTER MINIMUM EQUIPMENT LIST			
AIRCRAFT: B-737			REVISION NO : 52 DATE: 04/29/2008		PAGE: 78-1		
SYSTEM & SEQUENCE NUMBER		ITEM	1.	2.	NUMBER INSTALLED		
					3.	NUMBER REQUIRED FOR DISPATCH	
						4. REMARKS OR EXCEPTIONS	
78 - ENGINE EXHAUST							
1.	Thrust Reverser Systems						
	1) (-100/-200)	C	2	1	(M)(O) One may be inoperative provided thrust reverser is deactivated and secured closed.		
		C	2	1	(M)(O) One may be inoperative provided: a) Thrust reverser guide carriage is verified to be in over-center (forward thrust) position, and b) Override System is armed only after landing.		
					NOTE: Relief also applies to airplanes modified by STC SA5730NM or ST00131SE.		
	2) (-300/-400/-500)	C	2	1	(M)(O) One may be inoperative provided thrust reverser is locked in forward thrust position.		
	3) (-600/-700/-800/-900)	C	2	1	(M)(O) One may be inoperative provided: a) Thrust reverser is locked in forward thrust position, and b) Appropriate performance adjustments are applied.		
2.	REVERSER UNLOCKED Lights (-100/-200/-300/-400/-500/)	C	2	1	(M) One may be inoperative provided reverser is locked in closed (forward thrust) position.		
					NOTE: Relief also applies to airplanes modified by STC SA5730NM or ST00131SE.		
3.	Thrust Reverser In Transit Lights				Deleted in Revision 30.		
4. ***	Thrust REVERSER ARMED Light(s)	C	-	0	(M) May be inoperative provided lights are deactivated.		
					NOTE: Relief also applies to airplanes modified by STC SA5730NM or ST00131SE.		
5.	Thrust Reverser Override Switches (-100/-200)	C	2	1	One may be inoperative for an associated inoperative thrust reverser.		
					NOTE: Relief also applies to airplanes modified by STC SA5730NM or ST00131SE.		

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

78-2

SYSTEM & SEQUENCE NUMBER		1. ITEM	2. NUMBER INSTALLED	3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
78 - ENGINE EXHAUST					
6. ***	Thrust Reverser LOW PRESSURE Light (-100/-200)	C	1	0	(M) May be inoperative provided accumulators are charged before each departure. NOTE 1: Reverse thrust may not be available when System A pressure is lost. NOTE 2: Relief also applies to airplanes modified by STC SA5730NM or ST00131SE.
7.	REVERSER Lights (Aft Overhead Panel) (-300/-400/ - 500/-600/-700/ -800/-900)	C	2	1	(M) One may be inoperative provided associated reverser is locked in closed (forward thrust) position.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

79-1

**SYSTEM &
SEQUENCE
NUMBER****ITEM****1.****2.****NUMBER INSTALLED****3.****NUMBER REQUIRED FOR DISPATCH****4.****REMARKS OR EXCEPTIONS**

79 - ENGINE OIL

1.

Oil Quantity
Indication System

B

2

1

(M) Except for ER operations, one may be inoperative provided:

- a) Oil tank is filled to maximum recommended capacity at each refueling,
- b) There is no evidence of above normal oil consumption or leakage, and
- c) Associated low oil pressure warning system operates normally.

1) Oil Quantity
Indicator Test
Switch (-100/
-200/-300/-400/
- 500)

C

1

0

(M) May be inoperative provided:

- a) Oil tanks are filled to maximum recommended capacity at each refueling,
- b) There is no evidence of above normal oil consumption or leakage, and
- c) Engine low oil pressure warning systems operate normally.

2.

Oil Filter Bypass
Warning Systems

C

2

1

(M) One may be inoperative provided:

- a) Malfunction is in warning system, and
- b) Oil filter is inspected for presence of contaminants one each flight day.

3.

Oil Temperature
Indicators

Deleted prior to Revision 27.

4.

Oil Low Pressure
Warning Systems

B

2

0

May be inoperative provided associated oil quantity indication operates normally.

5.

Oil Pressure
Indicators

Deleted prior to Revision 27.

U.S. DEPARTMENT OF TRANSPORTATION

FEDERAL AVIATION ADMINISTRATION

MASTER MINIMUM EQUIPMENT LIST

AIRCRAFT:

B-737

REVISION NO :

52

PAGE:

DATE:

04/29/2008

80-1

SYSTEM & SEQUENCE NUMBER		1. ITEM	2.	NUMBER INSTALLED		3. NUMBER REQUIRED FOR DISPATCH	4. REMARKS OR EXCEPTIONS
80 - STARTING							
1.		Starter Valve Open Indications					
***		1) (-100/-200)	C	2	0		May be inoperative provided Start Valve Arming System is installed and operating normally.
		2) (-300/-400/- 500/-600/-700/-800/-900)	C	2	1		(O) One may be inoperative provided it is checked after engine start that associated valve is closed.
2.		Engine Starter Auto Cutout					
***		1) (-100/-200)	C	2	0		May be inoperative provided: a) Flight crew manually selects Start Switch OFF at 40% N2, and b) Takeoff in icing conditions is not permitted with No. 1 Engine Starter Auto Cutout inoperative.
		2) (-300/-400/- 500)	C	2	0		May be inoperative provided flight crew manually selects Start Switch OFF at 46% N2.
		3) (-600/-700/-800/-900)	C	2	0		May be inoperative provided flight crew manually selects Start Switch OFF at 55% N2.
3.		Starter Valves					
		1) (-100/-200)	C	2	0		(M)(O) May be inoperative provided alternate starting procedures are established and used.
		2) (-300/-400/- 500)	C	2	1		(M)(O) One may be inoperative provided: a) Modified Main Engine Controls or production equivalent have been incorporated, b) Associated start valve light operates normally, and c) Manual override start procedures are used.
		3) (-600/-700/-800/-900)	C	2	1		(M)(O) Except for ER operations, one may be inoperative provided: a) Associated start valve indication operates normally, and b) Manual override start procedures are used .
4.		Starter Valve Arming System (-100/-200)	C	1	0		May be inoperative provided Starter Valve Open Lights are installed and operating normally.
